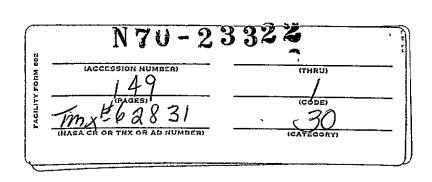
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#### OFERATIONAL TRAJECTORY DATA REPORT

September 10, 1968



AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY, REVISION I

Вy

Gordon W. Solmon

SPECIAL PROJECTS SECTION
FLIGHT MECHANICS BRANCH
FLIGHT TEST ANALYSIS DIVISION
AERO-ASTRODYNAMICS LABORATORY

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



DEG 1968

#### GEORGE C. MARSHALL SPACE FLIGHT CENTER

OPERATIONAL TRAJECTORY DATA REPORT

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Gordon W. Solmon

George C. Marshall Space Flight Center untsville, Alabama

#### ABSTRACT

This report presents the launch vehicle operational trajectory for the AS-205/CSM-101 mission. Included is a discussion of the predicted trajectory, mission objectives and constraints, and a tracking summary.

The S-IVB/CSM will be placed into a  $120 \times 150$  nautical mile elliptical orbit and will aid in further verification of the launch vehicle and spacecraft subsystems.

#### TABLE OF CONTENTS

		Page
ABSTRACT		i
TABLE OF CONTENTS		ii
LIST OF TABLES		iv
LIST OF ILLUSTRATIONS		v
DEFINITIONS AND SYMBOLS		vi
SUMMARY		1
SECTION I. Introduction		2
SECTION II. Mission Description		2
A. Mission Objectives		2
B. Mission Groundrules	and Constraints	2
SECTION III. Vehicle Characteristics	and Environment	3
^. Configuration	·	3
. Mass		3
· Propulsion		3
. Aerodynamics		4
. Environment Descri	ption	4
SECTION IV. Guidance and Control		4
A. Guidance		4
B. Control		5
SECTION V. Trajectory Description		5
A. Powered Flight		5
B. Orbital Phase		5
C. Spent Stage		6.
SECTION VI. Launch Vehicle Performa	ance Summary	6
SECTION VII. Tracking Summary		6
SECTION VIII. Dispersion Summary		6
SECTION TX Reference		7

#### TABLE OF CONTENTS (Cont'd)

		Page
APPENDIX A.	"Orbital Timeline and Venting Sequence"	87
APPENDIX B.	"Guidance Presettings"	100
APPENDIX C.	"Trajectory Listing, English Units"	. * 96
APPROVAL		134
DISTRIBUTION		135

#### LIST OF TABLES

Table	Description	Page
1	Flight Sequence of Events	. 8
2	Launch Vehicle Trajectory Summary	10
3	S-IB End Conditions of Flight	11
4	S-IVB End Conditions of Flight	12
5	S-IB Stage Trajectory Listing	14
6	Ullage Stage Trajectory Listing	- 21
7	S-IVB Stage Trajectory Listing	28
8	Orbital Flight Listing	35
9	S-IB Spent Stage Re-entry Data	45
10	Tracking and Telemetry Sta. Coordinates	51
11	Launch Vehicle Performance Characteristics	52
12.	Launch Vehicle Weight Breakdown	53
13	Dispersion Envelope at S-TB/S-IVB Separation	54
14	Dispersion Envelope at J-2 Engine Cutoff Signal	59
APPENDIX A:	"Orbital Timeline and Venting Sequence"	87
. 1A	Orbital Vent Sequence and Impulse History	88
2A	Orbital Attitude Maneuvers	89
APPENDIX B:	"Guidance Presettings"	90
1B .	S-IB Stage Steering Polynomial	91
ŹB	S-IB Pitch Attitude Commands	92
3в	IGM Presettings.	93
APPENDIX C:	"Trajectory Listing, English Units"	96
1C	S-IB Stage Trajectory Listing	97
2C	Ullage Stage Trajectory Listing	104
3C	S-IVB Stage Trajectory Listing	111
4C	Orbital Flight Listing	118
5c	S-IB Spent Stage Re-entry Data	128

#### LIST OF ILLUSTRATIONS

Figure	Description	Page
1	S-IB Stage Altitude vs Ground Range	64
2	S-IB Stage Space-Fixed Velocity vs Time	65
3	S-IB Stage Space-Fixed Path Angle vs Time	66
4	S-IB Stage Longitudinal Acceleration vs Ti	67
5	S-IB Stage Dynamic Pressure vs Time	68
6	S-IB Stage Pitch Attitude Commands vs Time	69
7	S-IB Stage Pitch Angle of Attack vs Time	70
8	S-IVB Stage Altitude vs Ground Range	71
9	S-IVB Stage Space-Fixed Velocity vs Time	72
10	S-IVB Stage Space-Fixed Path Angle vs Time	73
11	S-IVB Stage Longitudinal Acceleration vs Time	74
12	S-IVB Stage Pitch Attitude Commands vs Time	75
13	S-IVB Stage Yaw Attitude Commands vs Time	76
14	S-IVB Stage Space-Fixed Velocity vs Flight Path Angle	77
15	S-IVB Stage Thrust vs Time	78
16	Wind Profile	79
17	SA-205 Orbital Ground Projection	80
18	Launch Phase Ground Projection With Communication Coverage Above 5 Degree Elevation Angle	81
19	Launch and Orbital Phase Mission Timeline With Tracking and Telemetry Coverage	82
20	Vehicle Profile	86

#### DEFINITIONS AND SYMBOLS

Aerodynami	ĹC	Heatin <sub>6</sub>
Indicator	(2	AHI)

Altitude (ALT)

Vehicle altitude above the referenced ellipsoid measured along the geocentric position vector.

Angle of Attack, Pitch (ALP)

Angle between the pitch plane relative velocity component and the longitudinal axis of the vehicle measured positive nose up.

Central Range Angle (RANGLE)

Angle between the instantaneous space fixed position vector and the space fixed position vector at Guidance Release.

Descending Node Argument

Angle measured in the equatorial plane between the orbit plane descending node and the space fixed meridian plane defined at Guidance Reference Release.

Drag

Component of the resultant aerodynamic force along the relative velocity vector, measured positive opposite to the velocity vector.

Dynamic Pressure (QQQ)

 $\frac{1}{2}$  x (Density) x (Relative Velocity)<sup>2</sup>

Earth Fixed Position (XE, YE, ZE)

Position vector components in an earthfixed pad-centered plumbline coordinate system. The XE axis is coincident with the reference ellipsoid normal, positive upward. The ZE axis is parallel to the earth-fixed aiming azimuth and is positive downrange. The YE axis completes a righthanded system. (PASCS 10)

Earth Fixed Flight Path Angle (VTHE)

Angle between the earth fixed velocity vector and the earth fixed geocentric position vector (PASCS 11), measured positive downrange from the position vector.

Earth Fixed Crossrange

YE component of the earth fixed position. (PACS 10)

#### DEFINITIONS AND SYMBOLS (Cont'd)

Earth Fixed Velocity Components (DXE, DYE, DZE) Velocity vector components in PASCS 10.

Earth Fixed Velocity Magnitude (VVVE)

 $\sqrt{\text{DXE}^2 + \text{DYE}^2 + \text{DZE}^2}$ 

Space Fixed Flight Azimuth (AZ\*)

Angle between the instantaneous space fixed velocity vector direction and true north, measured positive east of north.

Earth Fixed Flight Azimuth (AZI)

Angle between the instantaneous earth fixed velocity vector direction and true north, measured positive east of north.

Geocentric Declination (DECL)

Angle between the geocentric radius vector and the true equatorial plane measured positive north of the equator.

Geodetic Latitude (LATT)

Angle between the reference ellipsoid normal through the point of interest and the true equatorial plane, measured positive north of the equator.

Ground Range (RANGE)

Surface range measured from the launch site to the sub vehicle point.

inclination

Angle between the instantaneous flight plane and the equatorial plane.

Longitude (LONG)

Angle between the Greenwich meridian plane and the projection of the geocentric position vector in the equatorial plane, measured positive east of Greenwich.

Longitudinal Acceleration (DDYM)

That part of the total measurable acceleration directed along the longitudinal axis of the vehicle.

Mach Number (MACH)

(Relative Velocity)  $\frac{b}{4}$  (Local speed of sound)

#### DEFINITIONS AND SYMBOLS (cont'd)

Thrust

Vehicle thrust .

Mass

Mass of the vehicle.

Pitch, Yaw, Roll (PHIP, PHIY, PHIR)

Eulerian angle of vehicle attitude measured with respect to the space fixed coordinate system. Vehicle attitude is defined by the ordered rotation of pitch, yaw, and roll respectively. PHIP = negative nose downrange; PHIY = positive nose right; PHIR = positive clockwise looking aft.

Pitch, Yaw, Roll Rates (DPHIP, DPHIR, DPHIR)

Vehicle attitude rates.

Radius (RRR)

 $\sqrt{x^2 + y^2 + z^2}$ 

Relative Velocity (VR)

Velocity relative to the atmosphe: (includes wind velocity).

Space Fixed Position (X, Y, Z)

Position vector components in a space fixed, earth centered, plumbline coordinate system defined at Guidance Reference Release. The X axis is parallel to the reference ellipsoid normal which passes through the launch site. The Z axis is parallel to and positive in the same direction as the earth fixed firing azimuth. The Y axis completes the right handed system. This is Project Apollo Standard Coordinate System 13. (PASCS 13)

Space Fixed Flight Path Angle (VTH\*)

Angle between the space-fixed velocity vector and the radius vector (PASCS 13), measured positive downrange from the radius vector.

Space Fixed Velocity Components (DX, DY, DZ)

Velocity vector

in PASCS 13.

Space Fixed Velocity Magnitude (VVV\*)

$$\sqrt{DX^2 + DY^2 + DZ^2}$$

Tim

Instantaneous flight time referenced to first motion.

Command Attitudes (CHIP, CHIY, CHIR)

Programmed vehicle attitude commands.

#### DEFINITIONS AND SYMBOLS (Cont'd)

IU - Instrument Unit

SLA - Spacecraft Lunar Module Adapter

SM - Service Module

CM - Command Module

CSM - Command Service Module

LES - Launch Escape System

PU - Propellant Utilization System

APS - Auxiliary Propulsion System

IGM - Iterative Guidance Mode

GRR - Guidance Reference Release

IECO - - Inboard Engine Cutoff

OECO - Outboard Engine Cutoff

PMR - Programmed Mixture Ratio

GCS - Guidance Cutoff Signal

#### SUMMARY

This report documents the AS-205/CSM-101 Launch Vehicle Operational Trajectory, Revision 1, data. These data are presented from Guidance Reference Release (GRR) through S-IVB/IU lifetime. All detailed trajectory data are generated using an unbiased S-IB stage tilt program. The revision is necessary because of the decision to have an open loop propellant utilization system in the S-IVB stage.

The predicted S-TB and S-TVB stage end conditions are summarized below:

		S-IB/S-IVB Separation	Guidance Cutoff Signal
Flight Time	(sec)	144.49	614.63
Altitude	(km)	62.00	227.94
Space Fixed Velocity	(m/sec)	2325.8	7780.7
Space Fixed Path Angle	(deg)	63.420	90.007
Range	(km)	61.95	1819.97

. The nominal weight at Guidance Cutoff Signal is predicted to be 67757 pounds. This is 1452 pounds in excess of the nominal propellant depletion weight.

The supporting tracking analysis will be published by R-AERO-FT. Summaries of the tracking coverage are presented in Figures 17 - 19.

Summaries of the  $\pm$  3  $\sigma$  trajectory flight envelopes at S-IB/S-IVB separation and S-IVB cutoff are presented in Tables 13 and 14. Detailed dispersion data will be presented in the "AS-205/CSM-101 Launch Vehicle Dispersion Analysis, Revision 1."

#### SECTION I. INTRODUCTION

The AS-205/CSM-101 Launch Vehicle Operational Trajectory, Revision 1, is presented in this document. Detailed trajectory data are presented from Guidance Reference Release (GRR) through S-IVB/IU lifetime.

Other pertinent data such as tracking and dispersion summaries are also included.

Acknowledgments are made to personnel of R-AERO-FT for providing the tracking data and to Mr. William R. Bailey for the dispersion summaries.

#### SECTION II. MISSION DESCRIPTION

#### · A. Mission Objectives:

The basic purpose of the AS-205/CSM-101 mission is to launch and insert the manned Block II Apollo spacecraft into a near earth orbit to verify the spacecraft/crew operations and subsystems performance for an orbital mission.

The planned mission profile and support plans reflect the requirements for the following primary objectives: (1) Demonstrate CSM/crew performance in an earth orbital environment; (2) Demonstrate crew/space vehicle/mission support facilities performance during an earth orbital mission; (3) Demonstrate the adequacy of the launch vehicle attitude control system for orbital operation; (4) Demonstrate CSM active rendezvous with the S-IVB/IU/SLA; (5) Demonstrate S-IVB orbital safing capability; (6) Evaluate S-IVB J-2 engine ASI line modification.

The secondary test objectives are: (1) Evaluate the S-IVB/IU orbital coast lifetime capability; (2) Demonstrate CSM manual launch vehicle orbital attitude control.

#### B. Mission Groundrules and Constraints:

The following mission criteria and vehicle constraints have been included in shaping the S-IB stage profile and in establishing the S-IVB terminal conditions: (1) Launch from AFETR Pad 34; (2) Pad oriented launch azimuth of 100 degrees east of north; (3) Flight azimuth of 72 degrees east of north; (4) S-IB stage tilt program shaped for load relief during high dynamic pressure region; (5) Guidance command angle rate limitat of 1 deg/sec in pitch and yaw; (6) S-IVB cutoff conditions for a 120/150 nautical mile elliptical orbit insertion.

#### SECTION III. VEHICLE CHARACTERISTICS AND ENVIRONMENT

#### A. Configuration:

The AS-205/CSM-101 Saturn IB Launch Vehicle consists of a S-IB stage, a S-IVB stage, an instrument unit, and a payload. The payload consists of the Launch Escape System (LES), Command Module (CM), Service Module (SM), and a Spacecraft Lunar Module Adapter (SLA). An outboard profile of the complete configuration is shown in Figure 20.

#### B. Mass:

The AS-205 vehicle mass characteristics used to generate the operational trajectory are defined in Reference 1. Table 12 presents a vehicle weight breakdown of the launch vehicle and payload.

#### C. Propulsion:

The S-IB stage is powered by eight H-1 engines which have a nominally rated sea level thrust of 200,000 pounds each. The predicted thrust history for each H-1 engine including thrust decay and the associated turbine engine thrust were obtained from Reference 2.

The S-IVB stage is powered by a single J-2 engine which has a rated vacuum thrust of 200,000 pounds at a nominal mixture ratio of 5:1. The S-IVB stage has an open loop propellant utilization system which is set for a mixture ratio of 5.5 to 1 and a corresponding thrust for approximately 300 seconds. The mixture ratio then shifts to 4.5 to 1 for the remainder of the S-IVB stage. The predicted J-2 engine thrust history from 90% thrust to Guidance Cutoff Signal (GCS) was obtained from Reference 2. The thrust buildup and decay histories were obtained from References 4 and 5.

The S-IB stage is decelerated at separation by four solid propellant retro-motors (TE-M-29) mounted on the S-IB/S-IVB interstage. These motors nominally provide 36,720 pounds of thrust each (Reference 6).

There are three solid propellant ullage motors mounted on the S-IVB aft portion to provide the necessary positive acceleration to settle the S-IVB stage propellants for J-2 engine start. These motors are rated at 3460 pounds of thrust each (Reference 7).

The S-IVB Stage Auxiliary Propulsion System (APS) provides roll control during J-2 burn and pitch, yaw, and roll control during the orbital portion. The APS consists of two modules, each containing three 150 pound thrust (vacuum) hypergolic rocket engines.

#### p. Aerodynamics:

Aerodynamic data for the S-IB and S-IVB stages of powered flight were obtained from Reference 8. Reference 9 presents the orbital drag data.

#### E. Environment Description:

The 1963 Patrick Reference Atmosphere Model defines the atmospheric properties incorporated in the trajectory simulation. The earth model and potential function are those of the Fischer Earth Model. Since the planned launch date is for a "low wind period," the S-IB stage tilt program is not wind biased. However, the median wind for the month of October was included in the trajectory. The wind profile was derived from Reference 10 and presented graphically in Figure 16.

#### SECTION IV. GUIDANCE AND CONTROL

The Saturn IB inertial guidance system performs navigation evaluations, issues discrete commands, initiates certain guidance and control functions, and issues steering commands to guide the launch vehicle to the pre-specified terminal conditions.

#### A. Guidance:

The AS-205/CSM-101 Launch Vehicle guidance is divided into three phases: (1) Pre-IGM; (2) IGM; (3) Orbital.

The Pre-IGM phase provides guidance commands in pitch, yaw, and roll from lift-off to the time of IGM initiation. The S-IB pitch program polynomials and the yaw and roll command histories are presented in Appendix B.

The S-IVB stage is steered to the desired terminal conditions by the Iterative Guidance Mode (IGM). The IGM equations and logic which provide guidance in pitch and yaw are initiated at TB3 + 25 seconds. The IGM constants are presented in Table 3B.

The orbital guidance mode provides pitch, yaw, and roll attitude commands during the orbital phase. The orbital maneuvers are presented in Table 2A.

#### B. Control:

The S-TB stage is controlled in pitch, yaw, and roll by four swivelable H-1 engines. The S-TB stage control law, gains, and network characteristics are given in Reference 11.

The J-2 engine provides pitch and yaw attitude control throughout S-IVB powered flight. The Auxiliary Propulsion System provides the roll control. The S-IVB control data are presented in Reference 11.

#### SECTION V. TRAJECTORY DESCRIPTION

#### A. Powered Flight:

All of the detailed trajectory information is based on an unbiased S-IB tilt program. Table 1 presents the nominal sequence of events for the AS-205/CSM-101 Launch Vehicle Operational Trajectory. Table 2 presents a summary of trajectory parameters at pertinent events from GRR to loss of S-IVB/IU attitude control. Tables 3 and 4 present convenient summaries of the S-IB and S-IVB stage end conditions. Detailed tabular listings are contained in Tables 5 - 9 and Appendix C. Graphical displays of various trajectory parameters are presented in Figures 1 - 15.

The S-IB launch phase begins at Guidance Reference Release, which is assumed to be 5 seconds prior to first motion, and ends at S-IB/S-IVB physical separation. Separation occurs at 144.49 seconds after first motion.

During the S-IB powered flight phase, a constant thrust bias of .4 per cent of the vehicle sea level longitudinal thrust is included. This is an attempt to minimize the effects of an apparent systematic shift in the ground to flight test performance level.

The S-TVB stage J-2 engine start command is issued 2.7 seconds after the S-TB stage outboard engines cut off. Approximately 25 seconds after outboard engine cutoff, the Iterative Guidance Mode is initiated and steers the S-TVB stage to the desired terminal conditions. These terminal conditions are nominally reached approximately 614.6 seconds after first motion.

#### B. Orbital Phase:

The orbital phase begins ten seconds after Guidance Cutoff Signal (GCS) and continues through the guaranteed S-IVB/IU lifetime. During the orbital

phase the S-IVB stage is required to perform certain attitude maneuvers. This is accomplished by the APS. Detailed information concerning the orbital maneuvers is presented in Appendix A. The orbital trajectory is listed in Table 8.

#### C. Spent Stage:

This portion of flight is initiated when the four retro-rockets are ignited at S-IB/S-IVB separation. The S-IB spent stage re-entry trajectory is presented in Table 9.

#### SECTION VI. LAUNCH VEHICLE PERFORMANCE SUMMARY

The predicted S-IB and S-IVB stage performance characteristics presented in Table 11. These data are time averages of the detailed vehicle performance data presented in Reference 2. These data are to be used in post-flight trajectory analysis.

#### SECTION VII. TRACKING SUMMARY

The tracking and telemetry station coordinates are presented in Table 10. A summary of the launch and orbital timeline is presented in Figure 19. The associated orbital ground projection is given in Figure 17.

#### SECTION VIII. DISPERSION SUMMARY

Tables 13 and 14 present the  $\pm$  3  $\sigma$  dispersions for certain trajectory parameters at S-IB/S-IVB separation and S-IVB cutoff, respectively. These data are extracts from the data to be presented in the "AS-205/CSM-101 Launch Vehicle Dispersion Analysis, Revision 1."

#### SECTION IX

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- 4. "Saturn S-IVB 205 Stage Acceptance Firing Report," SM-47471, DAC, July 1966.
- 5. "Saturn IB J-2 Engine Characteristics," R-P&VE-66-M-16, May 11, 1966.
- 6. "Model Specification Rocket Motor Solid Propellant, Thiokol Chemical Corp., Model No. TE-M-29-5."
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- 8. "Axial Force Characteristics of the Saturn IB, AS-205/CSM Vehicle," R-AERO-AD-68-25, April 16, 1968; and "Static Aerodynamic Characteristics of the Apollo-Saturn IB Vehicle," NASA TMX-53657.
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- 10. "Cape Kennedy Wind Component Statistics 0 to 60 km Altitude for 72 Degree Flight Azimuth for Monthly and Annual Reference Periods," R-AERO-Y-90-66, March 23, 1966.
- 11. "Control Gains and Shaping Networks for AS-205, AS-207, and AS-209, S-IB and S-IVB Stages," R-ASTR-F-68-15, February 7, 1968.

TABLE 1
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY
FLIGHT SEQUENCE OF EVENTS

Nominal Filg (HR:MIN:SEC)		Program Time (SEC)	Event
- 0:0:05.0	<b>5,00</b>	Hilly week have high made	Guidance Reference Release (GRR)
- 0:0:03.1	- 3.10	shade Aller ange best state	Initiate S-IB mainstage ignition sequence.
0:0:00.0	0.00	Shift was said help had	First motion.
0:0:00.2	0.20	(0.0)1	Lift-off signal; initiate Time Base 1.
0:0:10.2	10.20	(10.0);	Initiate pitch and roll maneuvers.
0:1:15.0	75.0	tific lade talls, non-quie	Maximum dynamic pressure.
0:1:40.2	100,20	(100.0) <sub>I</sub>	Control gain switch point.
0:2:00.2	120,20	(120.0)1	Control gain switch point.
0:2:13.9	133.91	(133.71) <sub>1</sub>	Enable S-IB propellant level sensors.
0:2:14.5	134.50	(134.3)1	Tilt arrest.
0:2:16.9	136.91	(0.0) <sub>2</sub>	Level sensor activation; Initiate Time Base 2.
0:2:20.1	140.11	(3.2) <sub>2</sub>	Inboard engine cutoff (IECO).
0:2:23.1	143.11	(0.0)3	Outboard engine cutoff (OECO); Initi
0:2:24.4	144.41	(1.3)3	Separation signal.
0:2:24.5	144.49	describes describes	S-IB/S-IVB physical separation.
0:2:25.8	145.81	(2.7)3	J-2 engine start command.
0:2:28.2	148.16	Act was day took had	Ullage burn out.
0:2:29.4	149.41	ryado dama rjado, spado abalif	90% J-2 thrust level.
0:2:31.8	151.81	(8.7)3	Command P. U. system activation.
0:2:36.4	156.41	(13.3)	Jettison ullage rocket motors.
0:2:43.1	163.11	*** *** *** ***	Jettison launch escape tower.

TABLE 1 (Cont'd)
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY
FLIGHT SEQUENCE OF EVENTS

Nominal Fligh (HR:MIN:SEC)		Program	Event
0:2:48.1	(SEC)	Time (SEC) (25.0) <sub>3</sub>	Event Command IGM initiation.
-		<b>.</b>	
0:5:43.1	-343.11	(200.0)3	Control gain switch point.
0:7:34.4	454.41	$(311.3)_3$	Command mixture ratio shift.
0:10:14.6	614.63	<b>**</b>	Guidance cutoff signal (GCS).
0:10:14.8.	614 <b>.</b> 83	(0.0)4	Initiate Time Base 4 (Reflects an approximate 0.2 second systems delay).
0:10:24.6	624.63	200 dad 407 day 400	Orbital insertion.
1:34:26.8	5666.83	(5052.0)4	Initiate LOX dump.
1:42:26.8	6146.83	(5532 <b>.</b> 0) <sub>4</sub>	Initiate cold helium dump
1:46:27.8	6387.83	(5773.0)4	Completion of LOX dump.
2:30:14.8	9014.83	(8400.0)4	Termination of cold helium dump.
2:29:55.0	8995.00	<b></b>	Begin manual control of S-IVB attitude from spacecraft.
2:32:55.0	9175.00	****	End manual control of S-IVB attitude from spacecraft; return control to I. U.
2:54:55.0	10495.00		Nominal CSM physical separation.
3:17:31.8	11851.83	(11237.0)4	Start stage control sphere helium dump.
4:30:14.8	16214.83	(15,600.0)4	Restart cold helium dump.
4:41:22.1	16882.13	(16,267.3)4	Completion of stage control sphere helium dump.
4:50:14.8	17414.83	(16,800.0)4	Completion of cold helium dump.

TABLE 2
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY SUMMARY

Event	Flight 'ime (sec)	Altitude (km)	S.F. Velocity (m/sec)	S.F. Flight Path Angle (deg)	S.F. Azimutn (deg)	Geodetic Lat Pos.North (deg)	Longitud (Pos East (deg)
uidance Ref. Release	- 5.0	0.03	408,97	90.000	90.00	28.52	- 80.56
irst Motion	0.0	0.03	408.97	90.000	90.00	28.52	- 80:56
lax. Dyn. Pressure	75.0	12.32	742.13	58.219	83.41	28.53	- 80.52
ilt Arrest	134.5	51.77	2066.79	61.964	76.10	28.65	- 80.11
Inboard Engine Cutoff	140.1	57.41	<b>2</b> 267 <b>.</b> 77	6 .722	75.77	28.67	- 80.02
Outboard Engine Cutof:	143.1	60.55	2325.01	63.168	75.70	28.69	- 79.98
S-IB/S-IVB Physical Sc	144.5	62,00	2325.84	63.420	75.70	28.69	- 79.96
J-2 Eng. Start Command	145.8	63.36	2320.54	63.675	<b>75</b> . <b>71</b> .	28.70	- 79.94
Ullage Case Jettison	156.4	73.91	2334.08	65.569	. 75.73	28.75	- 79.96
Initiate IGM	168.25	85.02	2377.35	67.534	75.72	28.80	- 79.56
EMR Sensed by IGM	454.8	219.16	4973.77	88.104	80.73	30.52	- 71.30
S-IB Stage Impact	556.2	0.	413.00	102.590	90.08	29.79	75.64
Guidance C/O Signal	614.6	227.94	7780.67	90.007	85.90	31.53	- 61,99
Orbit Insertion	624.6	227.95	7787.41	90.002	86.31	.31.58	- 61.24
CSM Separation	10495.0	240.22	7779.86	90.282	60.87	12.99	- 164.41

AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY
S-IB STAGE END CONDITIONS OF FLIGHT

Space Fixed Path Angle: 63.420 (deg Space Fixed Flight Azimuth: 75.70 (deg Earth Fixed Flight Azimuth: 72.23 (deg Geocentric Declination: (Pos. North) 28.53 (deg Geodetic Latitude: (Pos. North) 28.69 (deg	Flight Time: OECO + 1.379 Seconds	144.49	(sec)
Space Fixed Velocity:  Space Fixed Path Angle:  Space Fixed Flight Azimuth:  Earth Fixed Flight Azimuth:  Geocentric Declination: (Pos. North)  Geodetic Latitude: (Pos. North)  2325.84 (m/sc  (deg. 3.420 (deg.	Radius:	6435263.	(m)
Space Fixed Path Angle: 63.420 (deg Space Fixed Flight Azimuth: 75.70 (deg Earth Fixed Flight Azimuth: 72.23 (deg Geocentric Declination: (Pos. North) 28.53 (deg Geodetic Latitude: (Pos. North) 28.69 (deg	Altitude:	61995.	(m)
Space Fixed Flight Azimuth: 75.70 (deg Earth Fixed Flight Azimuth: 72.23 (deg Geocentric Declination: (Pos. North) 28.53 (deg Geodetic Latitude: (Pos. North) 28.69 (deg	Space Fixed Velocity:	2325.84	(m/sec)
Earth Fixed Flight Azimuth: 72.23 (deg Geocentric Declination: (Pos. North) 28.53 (deg Geodetic Latitude: (Pos. North) 28.69 (deg	Space Fixed Path Angle:	63.420	(deg)
Geocentric Declination: (Pos. North) 28.53 (deg Geodetic Latitude: (Pos. North) 28.69 (deg	Space Fixed Flight Azimuth:	75.70	(deg)
Geodetic Latitude: (Pos. North) 28.69 (deg	Earth Fixed Flight Azimuth:	72.23	(deg)
George Databases, (2000 to the total to the total to the total to the total to	Geocentric Declination: (Pos. North)	28.53	(deg)
Longitude: (Pos. East) - 79.96 (deg	Geodetic Latitude: (Pos. North)	28.69	(deg)
	Longitude: (Pos. East)	- 79.96	(deg)

### Space Fixed Position and Velocity Components

X	=	6434124.	(m)
Y	<u>==</u>	35609.	(m)
Z	=	115707.	(m)
DX.	=	1002.54	(m/sec)
DY	=	118.63	(m/sec)
DΖ	=	2095.32	(m/sec)

#### Vehicle Attitude and Attitude Rate

Pitch Attitude Angle	<b>≕</b> ⊷	59.31	(deg)
Yaw Attitude Angle	= -	0.07	(deg)
Roll Attitude Angle	=	0.00	(deg)
Pitch Rate		0.00	(deg/sec)
Yaw Rate	==		(deg/sec)
Roll Rate	=	റ.00	(deg/sec)

AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY
S-IVB STAGE END CONDITIONS OF FLIGHT

Flight Time: GCS	614.63	(sec)
Radius:	6600292.	(m)
Altitude:	227941.	(m)
Space Fixed Velocity:	7780.67	(m/sec)
Space Fixed Path Angle:	90.007	(deg)
Space Fixed Flight Azimuth:	85.90	(deg)
Earth Fixed Flight Azimuth:	85.67	(deg)
Geocentric Declination: (Pos. North)	31.37	(deg)
Geodetic Latitude: (Pos. North)	31.53	(deg)
Longitude: (Pos. East)	61.99	(deg)
Inclination:	31.605	(deg)
Descending Node Argument	119.0	(deg)

#### Space Fixed Position and Velocity Components

X	===	6258278.	(m)
Y	==	144698.	(m)
Z	===	2092098.	(m)
DX	***	- 2472.80	(m/sec)
DY	<b>::::</b>	412.48	(m/sec)
DZ	=	7365.73	(m/sec)

#### Vehicle Attitude Angles

Pitch Attitude Angle	=	- 105.87	(deg)
Yaw Attitude Angle	=	3.37	(deg)
Roll Attitude Angle	=	36	(deg)

#### Osculating Conic Parameters

*Perigee Altitude	=	222.11	(km)
*Apogee Altitude	==	254.42	(km)
Eccentricity	=	.0024	•
Semi-major Axis	=	6616.43	(km)
True Anomaly	=	357.26	(deg)
Period	=	89.27	(min)

\*Referenced to Equatorial Radius (6378.165 km)

#### TABLE 4 (Cont'd)

## AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY - S-IVB STAGE END CONDITIONS OF FLIGHT

Flight Time: Orbit Insertion	624.63	(sec)
Radius:	6600287.	(m)
Altitude:	227951.	(m)
Space Fixed Velocity:	7787.41	(m/sec)
Space Fixed Path Angle:	90.002	(đeg)
Space Fixed Flight Azimuth:	86.31	(deg)
Earth Fixed Flight Azimuth:	86.11	(deg)
Geocentric Declination: (Pos. North)	31.41,	(deg)
Geodetic Latitude: (Pos. North)	31,58	(deg)
Longitude: (Pos. East)	61.24	(deg)
Inclination:	31.605	(deg)
Descending Node Argument	119.0	(deg)

#### Space Fixed Position and Velocity Components

X	=	6233099.	(m)
***		****	

Y = 148816. (m)

Z = 2165667. (m)

DX = -.2561.23 (m/sec)

 $PY = 410.93 \quad (m/sec)$ 

 $DZ = 7342.68 \quad (m/sec)$ 

#### Vehicle Attitude Angles

Pitch Attitude Angle = - 105.86 (deg)

Yaw Attitude Angle = 3.36 (deg)

Roll Attitude Angle = - .69 (deg)

#### Osculating Conic Parameters.

*Perigee Altitude	333	222.12	(km)
*Apogee Altitude	ta	277.47	(km)
Eccentricity	***	.0042	
Semi-major Axis	=	6627.96	(km)
True Anomaly	==	359.61	(deg)
Period	=	89.50	(min)

\*Referenced to Equatorial Radius (6378.165 km)

TABLE 5

S-IB POWERED FLIGHT TABLES							
·	TIME	ALT	RRR ·	VTH+	V V V •	A Z •	· 184
	(SEC)	(M)	(M)	(DEG)	(M/S)	(DEG)	G=M2/M=RAD}
	-5-00			90.00	<del></del>	90+00	0.
	0.00	34.	. 6373356.	90.00	4u8•97	90•0u '	.0.
	5,00	65	6373386	——— કે લે + 2 2	409 • 12		
	10.00	165.	6373486.	80.15	409.73	89.98	11043
	I 5 + 00	343+			411.20	89.97	40562
•	20.00	611.	6373932+	81.23	414.56	89.43 89.83	113960
	25.00	<del></del>	<del>6374300.</del>	70-44	4-20-40	89.65	268643
	30.00	1457•	· 6374779•	75.51	429.60		558323
	35.00	2057•	6375 <u>\$</u> 79	7 2 • 5 5 <del></del>	443.00	88.92	1055354
	40+D0	2789.	6376111•	69.65	461.27	88.36	1850850
	45 • 00	3645	6376986 <del></del>	66+93	484.92	87.68	3051491
	50.00	4673.	6378014.	64.48	514.39	86+9-1	47.72908-
	<del>55+00</del>	5AA4+	<del></del>	<del></del>	549+74	86.07	7115698
	60+00 ° .	7244•	6380565.	60.71	590+22	85.21	10104790
			6382089.	59+00	634.40	84.32	13729510
	7.0 + 0 0	10459•	6383778.	58.80	684.72	83.42	17988484
	75-00-	12322	<del></del>	58 · 2 2	7 4 2 + 1 3	82.53	22823860
	80.00	14374.	6387691.	57.82	806•91 879•1-3		28054564
	85+00		<del></del>			80.88	33282835
. 4	90.00	19092+	0392406 •	57.56	959+43	80 • 1 1	35282655
·	95 • 00	21/79+	<u> </u>	5 /7 2	1048 • 13 • • • • • • • • • • • • • • • • •	79.41	42425169
	100+00	24697 •	. 6398386•	58.01	1145.07	78.7 <i>}</i>	46143030
	1-05 - 00	27853	6401160+	58×4J	1250+36	78,20	49242733
	110.00	31257.	6404561.	50.06	1364 • 24	70 + 20 7-7-68	61-7-249-4-7-
	115-00	34916	<u> </u>	<u> </u>		77 • 21	53663070
	120.00	38843.	6412139 •	59.97	1019.92	7	5513956B
	25.00	43034+		oJ.62	1-763 + 1 <del>2</del>	76.41	56247171
	130.00	47504 •	6420790•	61.32	1917 • 51	7.6 + 1 D	56985417-
		5 1-7-66 <del></del>		6-1-+-9 a	2066 • 7-9	76.07	57076956
	135.00	52253 •	6425534 •	62.04	2084:02	78.U7 75.78	57697244
		<u>573</u> 94v	<del></del>		<del></del>	75,77	57703813
	1) [40.11	5/409•	6430662,	62.72	2267.77	75 • 7 U · · · ·	57901080
	<del>2}-143+11</del>		<del></del>		23-25-0-1	, <del>.</del>	58071429
	144.41	61913.	6435180.	63+40	2326.09	75.70 75.70	58078837
	3) 144.49	61996	6435 <del>763+</del>	63+#2	2325.484		

1) Inboard Cutof.	1)	Inboard	Cutoff
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<sup>2)</sup> Cutboard Cutoff
3) Separation

TABLE 5 S-IB PÓWERED FLIGHT TABLES

	5-15 POWEKED FLIGHT TABLES						
AAAE	RANGE	AZI	VTHE	THRUST	MASS	TIME	
( M/S	(M)	· (DEG)	(DEG)	· (N)	(KG)	(SEC)	
0.0		. N/A··	N/A	0	585942.88		
0.0	0•	N/A	N/A	6947027.	595942.80	0.00	
12a7	0 •	N/A	U.4U	715039-7 <del></del>	57.2028 - 98	5 • QQ	
27 • 4	~ ] •	N/A	0 • 45	7193015,	558115.08	10.00	
	×2•	N/A	0 • 28	7261442	544056+98	-15.09	
63.1	. 3.	54•85	0.79	7316200+	. 529998 . 88.	20.00	
	1-0+	6.7.5	2-09	<del>7,</del> 3 <i>657464</i>	5.15.363.435	<del>26+00</del>	
107•7	35.	69.42	3.90	7417116+	501727+82	30.00	
	88	70•29	6 • 15	7474618+	487566+99	35+00	
162+2	163.	70.61	8.74	7539065.	473406+14	40.00	
193.9		70•B2	l 1 • 5 8 <sup>-</sup>	7607769	459230+52	15 • 00	
229.0	580∗	71.64	14.59	7677844.	445054.89	50.00	
	925+	7-1-4-1-8	1-77 O	7746457v	<del>4-3-08-78-66</del>	<del></del>	
309.9	1401.	71.30	20.84	7805254.	416702-42	60.00	
351+5	203!+	71.40	24 • O4	7865551 <b>,</b>	402478.96	65 • 0 0	
398 + 9	2841.	71.50	27 • 25	7930847•	388255.51	70+00	
452 . 8	3865	71.57	30 • 33	7989454	37401-1→ <del>9-3</del>	75+00	
513.8	5136.	71 • 63	33.25	8039161.	359774•40	80.00	Å
582.0	<u> </u>	71-69	35.95		<del>345577+93</del>	85+0D	<u> </u>
658+D	8558.	71.75	38.54	8107523.	331381+46	90.00	
<del>7</del> 42+0	10790.	71.80	———41 • O3	8124984	317244+66	95+00	
834+2	13426.	71.05	43.36	8131797.	303107+67	100.00	
934+9	i65U9+	71 • 89	<del>45 • 5 2</del>	8 1-3 1-1 04 v	28399 <del>6-</del> !-1	1-05+00	
104404	20082,	71.93	47.51	8124551.	274884.36	110.00	
	241-89	/1+9.7	49.36	8-1-1-1-3-2-7	2 <del>,</del> 0 <del>820+53</del>	<del></del>	
1291.7	28877.	72.02	51 • 12	8091718.	246756.72	120.00	
1430 • 8	34202	<u> </u>	52 + 8 1	<del>8</del> 069348 <del></del>	2-3 2 7 6-6-1-2 3	<del></del>	
1581.2	40218.	72.09	54 + 41	8040994•	218775.74	130+00	
1-727 • 0	46273	72.13	55.477	-8008U8-2	206285+79		
1743.9	46985.	72.13	55.92	· 8003879•	. 204898.02	135.00	
1850*1	54565	72418	5/-20	7949385	191020-28		
1924+1 	54741.	72.18	5/*31 5/-94	7947549, 3500u63.	190716.96	) 140 • 11	1) _2
1979 • 6	618179	72.23	58+26	208407	184858-35	144.41	
1979.2	61948	72.23	58+28	197848	<u> </u>	<u> </u>	_3

1) Inboard Cutoff
2) Outboard Cutoff
3) Separation

			TABLE- 5		***************************************	
		\$~I	B POWERED FLIGHT TABLE	es		·
TIME	X	Y	2	ΧO	DY	DZ
(SEC)	. (M)	(M) ,	(M)	(m/s)	(M/S)	(M/S
-5-00	(37333).	17047	-5539	0.00	126.38	388.9
0.00	6373330•	17679	-3594.	0.00	126.31	388.9
	637336D+	18310	-1649	1 2 • 4 4	126+20	388 • 9
10.00	63/3450.	18941.	295.	27+07	126+08	388 • 9
15.00	63736343	195713	2240	43 •·7 6·	126•44	389∘(
20.00	6373899 •	20201•	4187.	62 • 52	125.99	389 • 9
25.00·	<del>63/4263</del>	<del>20031</del>	<del></del>	83-48	1-25+78-	3.9.2 + 1
30.00	6374737•	21461+	8113.	106.54	125.95	396 • 6
35+00-	6375332+ <del></del>	22091		1-31 • 78	1-25 + 89	403 • 3
40.00	6376059•	22720.	12157.	159 • 1 4	125.75	414+
45+00-	<u>6376927+</u>	23348	14262+	1 38 • 5.9	125+59	428
50.00	6377948.	23976.	16452.	220.01	125.44	447 • 7
<del>55+00-</del>	<del>6379130</del>	240031 -	-18-/-46	253÷07	125+29	47-1-0-0
60.00	4.08 F N 8 F 9	25229.	21170.	286 • 61	125 - 13	500 • S
	6381993.	75×54•		318.56	124 - 97	534 • 3
70+00	6383668•	26478 €	20529 •	351.77	124.61	574•1 620•!
75 v 00 -	6-3 0 5 5 1 5 •		29513.	387 • 48	124 • 62	674 •
. 80•00 `	6387547.	2//25.	32/46.	425 • 70	124.44	734
	<del>6389716+</del>	<del>283+6</del>				803•
90.00	6392214.	28967.	40108+	509+09 553+10	124.05	881+
95+00-	6394569 ·····	29567.		<del>-</del>	123.58	968•
100.00	6397746 •	30205•	48939.	590+62 645•61	123,58	1063
<del>1</del> 05+00 <del>-</del>	640085a	<del>3</del> 0623 <del>√</del>		. 694 • 12	122.92	1168+
110.00	6404206+	31438.	59590.	743.93	122,52	1201
<del></del>	<del>6467301</del>	32057.		794.33	122.09	1406.
120.00	6411647.	32663.	72427• 79795•	845 • 22	121.44	1542+
	6415745*	33272.	87872.	890.09	120.68	1690
130.00	6420100+	33677 •	87872. 95801.	943+55	119.96	1834
<del></del>	6424240	34417.	96723.	948.79	119.88	1851
135.00	6424713.	34479.	104411	10,13+89	119.02	2025
140+00	6429592	35UĴ&	106633.	1005•18	119.00	2029
1) 140 - 11	6429702.	35089. 36446.	108033.	1012.33	11.400	2389
_2)143+11	<del></del>		115542.	1003.26	118.63	2095 • 2
144.41	6434045 • 6434124 •	35600 <b>.</b> 35609.	1155920 115707	1003.20	118.63	2095

1) Inboard Cutoff
2) Outboard Cutoff
3) Separation

TABLE 5

		S-1	B POWERED FLIGHT TABLE	ES		
TIME	Xε	YE	ZE	DXE	DYE	DZE
(SEC)	(8)	. (M)	(M)	(M/S)	(M/S)	(M/S)
-5.00	35,				0.00	-0.400
0.00	35.	<b>~</b> 0 *	~O.	-0.00	Q0 • 0 ~	0.00
	45+	-0+	-0.	1 2 • 7.1		
10.00	165.	-0.	~O•	27.48	-0.10	<b>#0.15</b>
15+00	343.	······································		44.28		
20.00	611.	~2 ·	~O.	63.18	-0.09	0 • 7 7
25.00	979			84-27	~0.06	3+00
30.00 35.00	14574	~2 *	.33.	107.50 132.87	-0.04	7 • 23 14 • 20
40.00	2790•	~2 <b>.</b>	181.	160.40		24.52
45.00	3665	-4 -	337.	190.04		38•76
50.00	4693•		576.	221.68	-0.4D	57.51
55.00	5885	-6.	92-1	255-0U	-0.40	91-17
60.00	7244	-11.	1396.	288.83	~0.58	109.74
	8770		2025	321•13		143.04
70.00	10458	-17.	2837.	354.75	-0.70	182.51
75.00	1 2 3 2 1	-21.	3862	394493		228+61
80.00	14372	<b>-25</b> •	5134.	429 • 7 1	-D.77	201.68
95.00	16623	29.	6690+	471-10		341-82
90.00	19086.	-32.	8566.	514.49	· ~0.72	410+21
	21-7-70 ····	36	1 0 d 0 7 +	559+37		487.56
100.00	24683.	39.	13456. '	გე5•88	-0.54	573+52
105+00	27.832+		1 655 <del>Z_,</del>	654 <b>a</b> DO	-0 ° 43	668 + 20
110.00	31225.	-43.	20153.	703•79	<b>~0</b> • 3 Q	771+76
115.00	34872		24290	755+0A		
120.00	38777.	<b>→44</b>	29019.	807 - 69	០∙ប្ត	1008.57
25+00	4 2 9 4 4	-43.	34395+	859 • 83	0 • ¡ 8	1143 + 67
130.00	47377 •	~42.	46476+	913.36	0.29	1290+78
1.3.9 + 5.0	51597	——————————————————————————————————————	46603.	962 + 29	0.45	1434+15
135.00	52079•	-40.	47324• 55005•	967•77 1025•45	0 • 47 0 • 72	1450 • 76
140.00	57059.	-38+				· · · · · · · · · · · · · · · · · · ·
1) 140.11	57172 · 60270 ·	-37·	55184. 	1026.80	0.73 0.97	1627+32 
144.41	61611.	-34.	62362	1026.45	1.08	1692.69
3) 144.49	<u> </u>	34,	62496,	1025 • 7.6	1.09	1692.75

<sup>1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Separation

TABLE 5

#### S-IB POWERED FLIGHT TABLES

TIME	РНІР	PHIY	PHIR ,,	DPHIP	DPHIY	OPHIR
(SEC)	. (DEG)	· (UEG)	(DEG)	(SEG/S) ·	(DEG/S)	, (DEG/S)
5-00			28.00		<del>0+00-</del>	<u>~0+00</u>
0.00	-0.02	0.01	27.99	-0.00	0.00	-0.00
5+00			26.01		0+04-	.0.01
10.00	0.02	~0.02	26.00	U • 00	0.00	-0.01
15+00		-0.05	24-66	0 + 1 2		*1.53
20.00	-1-40	<b>9∙</b> 07	19.17	-0.23	0.00	-D + 8 6
<del>25+00</del>	2-74	<del>- U + O 7</del>		<del></del>	<del></del>	<del>~Ω</del> -•-9 8
30.00	. 4.46	0.05	9 • 13	-0 - 3 9	-0.01	-1.00
35.00-		- 0 - 0 2		0+46		n1 • 00
40.00	-9·ú3	-0.03	-0.19	-v·51	-0.01	*0 * 0 4
45.00	1 1 68-	-u.03	-0.01	0+55	0+00-	+0+00
50.00	-14.51	-0.03	-0.01	-0.58	-0.00	*0.00
55.00		<del>-6.03</del>			0U	~0+00
60.00	-20.39	-G•04	-v•01	-0.60	-0.00	-0.00
65.00	-20457			0.62	-0.00-	-0.00
=			-U.02	-0+59	<b>~0.00</b>	-0.00
70.00	-26.56	~0.05		0 • 6 2	0 - 40-	-0.00
75+00	<del></del>	-0-05	-0.01	~0.62	0.00	0.00
80,00	~32,59 <del>35,7?</del> -	+∪.05 <del>-∪.05</del>			000	-0-00
<del></del>	-38.84	-0.05	0+00	-0.59	0.00	•0•00
, 90.00 95.00	-30+01 	-0.05 -0.04		0.50	0 • 0 0	•0.00
<del></del>	-43.93		0.00	-0.46	0.00	.0.00
100.00		<b>~0.0</b> 4		0.42		-0.00
		-0-06	-U•UU	-0.39	-0.00	•0 • 00
110.00	~48,15	<b>~</b> U•06	-U-U3			-000
115.00-	50-44	<del>-U-06</del>	~0.00	+O•45	+0.00	•0•00
120.00	-52.72	~0 ± 0.7	~0+00	0-42	0.00	-0.00
1-25+00-		~u • <del>1 6</del>		~0.41	-0.00	0+00
130.00	-57.08	-0.15	-0.00			-0+00
		<del>-(), ], 5</del>		-0.37	~0.00	+0+00
135.00	-50.08	+0.15	-0.00			-0+-00
	<del>5</del> 4+31-	-0.16		<del></del>	<del>00</del>	0.00
1) 140+11	-59.31	-0.16	U - O O	-0.01		
2)-143-11-	<del>54.32</del> _	<del>-0.08</del>	-0.00	0 <del>- 0 1</del>	0.01	Q+Q0
144.41	~ <u>5</u> 9.31	-0.07	-u - 00	0.00	0.00	-0.00
-3)1-44-49	#59+31-	~0+0 <del>7</del>	U_A U O	0.00	0-00-	*0*00
		<u> </u>				
1) Inboard Cutoff					<del></del>	
2) Outboard Cutoff						
<ol><li>Separation</li></ol>						

TABLE 5

		S-IP				
TIME	41H2	CHIY	CHIR	LONG	DECL	LATT
(SEC)	(DEG)	. (DEG)	(DEG)	(DEG)	(0EG)	· ( DEG
	G-00	<u> </u>	<u> </u>	~80.56	28.36	29.45
. 0.00 . 0.00	0.00	0.00	-28.00	80.56	28.36	28+5
5.00	0.00		28.00		28 <b>+</b> 36	Z8 • S
10.00	0.00	0.00	-20.00	-80.56	28.36	28 • \$
16+00	C			F&U+ <u>b</u> 6	28.34	28+5 28+5
120+00	1.66	0.00	-18.20	-80.54	20+36	
25.50		<del></del>			20-34	28.5
30.00	-4.93	0.00	-6 + 20	-80·56	28.36	
<u>                 35                   </u>	7-1-		-3 + 20		28+36	28+5
40.00	-9.58	0.400	0.00	-80.50	20+36 20+36	
45 - 00	1-2-30	U • 00	O + () ()	-40+5¢	28.36	28.5
50.00	-15.21	0.00	<b>u •</b> 00	-80.55		28+5
<del>55+00</del>		<del>0.00</del>	0+00	-80.55	. 28,36	28.5
\$0.00	-21 • 43	0.00	U • U 0		28.37	28,5
<u>\$</u> 5+00			Ü.00	-80+53	28.37	28.5
70.00	-27+79	0.00 	O • O O:	-80.52	28.37	28•5
<del>75+00</del>	-30+9 <del>0</del> -33-89	υ.Ου υ.Ου	0.00	-80.51	28.38	28+5
80.00			U+O			28v5
	- 39.47	0.00	0.00	-80.48	26.39	28+5
90.00	- 34.47 - 42.03				28 + 37	28•5
75+00 100+00	44.41	0.00	, 0.00	-80.43	28+40	28.5
100.00 105.00		0.00	0.00		28 • 41	28•5
110.00	- 48.57	0.00	00.0	-80.37	28.42	28+5
110.00					28.43	28.5
120.00	-53.21	0.00	0.00	-80.28	28 - 44	28 - 6
125.00	-55.40	0.00	0.00	=80 ^ 23	28*46	28 + 6
130.00	-57.50	• ច • ០ព	ა∙0ე	-80+17	28 • 47	28 • 6
	~-5°.32		0.00		28+49	28.4
135.00 .	- SP.32	0.00	0.00	-80+10	28 . 49	28 4 6
<u> </u>	<del>532</del>	<del></del>		80+03		<del></del>
1) 140.11	- 59.32	J.00	0.00	-80.03	. 28.51 28.63	28°6
2) (43-11-		<del>0,0</del> 0	<u>0+0</u> \	-79.98 -79.96	28,53	28+6
144.41	-99.32	0.00	0.00	-/9.96 	28.53	20.6
3) 144.49	59.32	0,00				

<sup>1)</sup> Inhoard Cutoff
2) Outboard Cutoff
3) Separation

S-IB POWERED FLIGHT TABLES							
TIME	MACH	DRAG .	, ALP	ପ୍ରଧ୍ୟ	٧R	TACEL	
(SEC)	(U)	(N)	(DEG)	' (N/M2)	(M/S)	(M/S	
			N/A ·	1	0.00	0.	
-5.00		23.		1	1.22	11.	
0.00	0.00	2727	4 -7.0		12,76		
5+00	0.08	11961.	2.07	441+	27.50	12•	
10.00	0.13	30239-	1.00			1 3 a	
20+00	0.13	45362.	84.0	2238 •	63.22	13.	
20.00 -25.00	0.25	61067		3853+	84+40	1-4+	
30.00	0.32	85783.	0.41	٠٥٥٥٠	107.82	14•	
_35.00 35.00	0.32	114206			1 33 • 6'4	15•	
40.00	. 0•48	146090.	-0.41	11871.	162-16	15	
45.00	0.58	181289	+0 • 5 1	15500 •	193.59		
50.00	٠ ٧٤٠٥ .	220856.	-0.58		228.20	161	
55-00	<u> </u>	297903		23.284+	266+09	1.7-	
60.00.	0.96	528640.	<b>~</b> 0.67	26777•	306 • 39	17	
65,00	-• 1-	7-374-23+		29237.	347+38	17 ·	
70.00	1 • 2 9	668145.	<b>~</b> U•94	30886•	392.56	 18.	
<u>_75→00</u>		580164·*	×u +-7.4	31585+	444.57	-	
80.00	1.76 .	469622	-0.37	30869•	504.29	21	
_85+00-	2103	403269*		28275+	<u>579+01</u>	23	
90.00	2 • 2 8	312173.	0 + 1 4	23696 •	658 • 81 743 • 94	24	
95 +00	2 • 5 3	<b>233835</b>		19014	836.61	- 26	
100.00 .	2 + 8 !	171529.	0.03	14834。	936.67		
_1-0500	3 + 1-l	1 2 1-2-1 4 +	<del>-</del> <del>-</del> <del>-</del>	0211-	1043.93	29	
110.00	3 • 41	81842.	-0.14 -0.53	58444	1167.29		
	3472 -	49768.	*U+33	#û52•	1290.81	32	
120.00	4.05	21763.		2740	1429.87	34	
125+00	4+·40	2487	-1.91	1865	1580 • 27	36	
130.00	4.79	-7341.	-1 · 7 l 2 · 28	1325+	17-26 • 08	39	
<del>-134</del> √50	5 • · <u>2 · · </u>	-1-3 8 8 3 <del></del>	-2.32	1275	1742.93	39	
135.00	5+32	-14537, 	-2 - 3 2	847	1919014	41	
-140.00	6.00	-22964	+1.U8	837•	1923.17	41	
140+11	6+D1 6+30			6Ū4.	1978043		
	6.35	÷13982.	-0.04	510+	1978.58	1 *	
144.41				504.	1978.27		

<sup>-1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Soparation

			TABLE 6		
		ULLAGI	E AND J-2 BUILD-UP TAE	BLES	
TIME	ALT .	RHR	VTH.	V V V +	AZ.
(SEC)	(4)	(M)	(UEG)	(M/S)	(DEG)
140.40	61995.		43,42	2325.84	75.70
145.81	63364. 65682.	6436630. 6438944.	63.68	2320•54 2312•92	75 • 71 75 • 7.3.
149+41	67020.	6440281.	64.35	2314.09	75.74

ULLAGE AND J-2 BUILD-UP TABLES								
TIME	MASS	THRUST	YTHE	AZI	RANGE	VVVE		
(SEC)	(KG)	(N)	(DEG)	(526)	(H)	(H/S)		
144.49 145.81 148.08	138707.72 138683.39 138487.22 138403.47	41321. 41102. 437107. 838448.	58-28 58-56 59-05 59-34	72.23 72.25 72.27 72.29	64151 • 64151 • 67935 • 70153 •	1979.29 1973.01 1963.72- 1963.97		

		+	TABLE, 6							
	ULLAGE AND J-2 BUILD-UP TABLES									
TIME	Х	Y	Z	υx	DY	ΩZ				
(SEC)	(M)	(%)	(8)	(M/S)	. (4/5)	(M/S)				
144.49	434134	* 35409-	11570/	1002.54	118+63	2095.32				
145.81	6435440. 6437664.	35/66. 36035.	118475.	990+02 969+21	118.57 18.48	2095.40 2095.71				
149.41	6438946.	36173.	126024.	959+94	118 - 41	2102.26				

TABLE 6 ULLAGE AND J-2 BUILD-UP TABLES TIME х Y Z DY DΧ DΖ (SEC) (M) (M) (M) (M/S) (M/S) (M/S) 6434124. 35.609 11570/-1002.54 2095.32 6435440. 6437.664. 145.81 35/66. 118475 990.02 118.57 2095.40 2094.71 \_969.421.. 959.94 148.08 36035 123235 118 48. 149.41 6438946. 126024. 36193. 118.41 2102+26

TABLE 6

ULLAGE AND J-2 BUILD-UP TABLES									
TIME	PHIP	PHIY	PHIR	орнір	орніч	DPKIR			
(SEC)	(DEG)	(0EG)	(DEG)	(0£6/5)	(DEG/S)	(DEG/S)			
144849				<u> </u>		-0.00			
145.81 <del>148.</del> 08	-59.30	-0.06	-0.00	0.02	0+01	0.00			
149.41	-59.03	-0.10	0.06	. 0.08	0.06	0.0S			

			TABLE 6							
ULLAGE AND J-2 BUILD-UF TABLES										
TIME	CHIS	, CHIĂ	CHIK	LONG	DECL .	LATT				
(SEC)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)				
144.49	-52.32		U-1)n	<del>-79.</del> 96	28.53	<u></u>				
145.81	-59.32 -59.32	.0.00	0.00	-79 • 9 4 -79 • 9 0	28.54· 28.55	28•70 28•71				
148+08 149+41	- 59.32	0.00	0.00	-79.88	28.56	28.72				

			TABLE 6			
		ULLAGE	AND J-2 BUILD-UP TABL	ES		-
TIME	CHIS	, сній	CHIK	LONG	DECL '	LATT
(SEC)	(DEG)	(086)	(DEG)	(DEG)	(DEG)	(DEG)
	-59.32	·	0.00	-79.96	28.53	
145.81	-59.32	. o • o o	0.00	-79 + 94	28 + 54 +	28.70 -
148+08 '	-59.32	0.00	0.00	-79.4.9.0 -79.4.88	28.55 28.56	28 • 7 <u>2</u>
		•		-		
•			•			
•	•					
		•				
			,			
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	•		,		•	
26 .			•			
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	•				•	

・パラムヤナ シブ	7	444	of the second se	The formal designation of the first of the formal between the contract of the first

- S-IVB POWERED FLIGHT TABLES										
	TIME	ALT	RKR	vin+	¥ <b>∀</b> ¥*	AZO	RANGL			
	(SEC)	(8)	. (W)	(0EG) '	(8/5)	(DEG)	(DEG			
1}		· · · · · · · · · · · · · · · · · · ·		·		75.74				
•	150+00	. 67611.	\$440×71*	ልጓ*ቸል	2315+27	75 + 74	0+4			
		73711-			2334+08	75,73 <u></u> _				
	[63+]]	80274.	6453519.	. 66.70	2357 • 27	75.73	0+8			
	70 + 00 <del></del>		6459833 <sub>*</sub>			-75.72				
	120.00	.163823.	6477622+	70-60	2469.93	75-93	1 - 2			
	- 10±00	11/213.	61120121	<del>~~~~*********************************</del>	<u>z508101</u>	<del></del>				
	230.00	133780.	6506913.	75.36	2679+04	76.44	. 2.0			
<del>(40-11-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-</del>	250.00		4519799.	77.39		76.73	2 1 4			
	270.00	158349.	4531412.	79.22	2940+61	77.04	2 • 8			
	-270.00		6541624		<del></del>	77.36	3 • <del>2</del>			
	310+00	178129.	4551118*	82+28	3257+19	77.70	3+7			
<del></del>	-330-00		655332			78-05				
	150.00	193697 •	45666G6.	84.64	3633.17	78.42	4 * 8			
	370+00	200049	6572957	65 > 56	3845.96	78.91				
	390.00	205557+	6578483.	\$£+38	4076.58	79 * 22	0 + 6 6 + 6			
**************************************	410+00	2-1-0-4 \$ 4 <del></del>		87.05 07.59	4327 • 82 4600 • 81	7.9 . 6 6	7 • 3			
`	430.00 <u>,</u> - <del>450.</del> 00	214441.	6587929.		4000.83 4096.83	80 + 1 2 60 + 41				
0.3		219185.	6591838.	01.38	. 4973.77	· 88.73	8 + 3			
2)	434*/5 470r00	221473			·		0 r J			
	490.00	223903.	6596505.	80.94	5445.34	81.08	9 . 7			
v-100	-5-10-00	275481*****	4578239*		5736+U\$					
	530.00	220907.	6599422.	87.56	6054+30	62.85	11.6			
(15/10/1 <del>-10/10/10/10/10/10/10/10/10/10/10/10/10/1</del>	550,00		6000104.	49-79	<del></del>					
	570.00	227960.	6600392.	89.94	6781.33	89.20	13.7			
		228040.			7200+08	84,93				
	610.00	2279496	6602020	90.09	- 7665.37	85,71	10.0			
3}-		727941	6400272		7-					
4)		22/951.	6600287.	90.00	7787 • 41	86.31	17.0			

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

TABLE 7

### S-IVB POWERED FLIGHT TABLES

		SPIVE FUNDAGE FAIGHT TABLES							
	, TIME	MASS	THRUST	Atue	AZI	RANGE	YVVE		
	(SEC)	(KG)	(N)	(DEG)	(066)	(M) ·	(8/5)		
	1)   " ? : 4 ]	138403.47	833448,	69.34	72.29	70153	1963.97		
	150.00 156.41	1382/1.21	877263. 	59•46 60•81	72·29 72·36	71141+	1964.74 1979.37_		
	163.11	135083484	1010782	62.18	/2.43	93546.	1798.47		
	170.00	1-2-9 3 3 7 + 1-4	I-D10849	63.55	72.49	105714	2021 • 72 -		
	190.00	124482.83	1007592.	66.93	72.94	142568 .	2097.74		
	<u></u>	1-76-1-7-4/		- <del> </del>	7-3-+-4-1	18174-2	2   88 1-3-		
	230.00	114755.66	1010718.	72.82	73.89	223372*	2292 0 49		
	250.00	107891.32	1007098	75.31	74.39	267616+	2412:05-		
	270.00	105036.88	1011138.	77.52	74 * 8 9	314649*	2545.36		
	5 & 0 * -0 t)	100185.92	<del></del> 1006414 <del></del>	79 <b>•</b> 48	75 • 3 <del>'</del>	364670*	2693+23-		
	310+00	95331.57	1010490*	81 - 19	75.90	417692.	2856 • 19		
	330-00				7 6 + 41	474556	3034+17		
	350+00	85635.84	1009047.	83.96	76.93	534916.	3228.62		
	370+00	80789+01	1006358 <del></del>	65 • 66	77:46	5992¢4*	3440 + 1 \$		
	390.00	75949.55	1004056.	85.98	78.01	667909*	3669+86		
	410+00	71100+9-1	1006154	86.74	78.57	791200	3920•3 <i>5</i>		
	430.00	66265.19	1005360.	87.36	79.15	8195510	4142.76		
	450.00	61426+49		67.84	74.75	903382 <del></del>			
29	2) 454.75	60286.96	1008089.	8/.93	79.89	924153.	4565.15		
	470-00	57371·•40	779446	88 * 36	80.36	992866	4763052-		
	470.00	53684.02	779303.	88.85.	#1.00	182440.	.5034-13		
	510.00	49997-24	7 7.9   4 4	89.22		1107430	5326+59		
	530.00	46311+28	778924.	69-53	82+34	1293294.	5644+69		
	<u>\$50.00</u>	42626+33	778652.	89.78	83-04		5991.60		
	570+00	38942.76	778266*	89.94	83+82	1524829+	6371.55		
	540+00	35261+50	7.7.7.57-2	9u+01	84.63	-16517.92	6790+25_		
	610+00	31534.83	776228.	90.04	85 • 47	1787271.	7255,54		
	4) 624.63	30734-06 30670•74	775758. 0.	90.00	85 • 67 86 • 11	161-997-2	7370+8 <del>1</del> 7377+54		

1) 90% Thrust

2) Mixture Ratio Shift

3) Guidance Cutoff
4) Orbit Insertion

TABLE 7

S-IVB POWERED FLIGHT TABLES										
TIME	· x	Y .	Z	ox .	DΥ	DZ				
(SEC)	( M )	(M) ·	(m)	(M/S)	. (M/S)	(M/S)				
1) 149.41	4438946	<del>-3</del> -6- <del>1-9-3</del>	126024+	<del>- 959+94</del> -		<del>2102+2</del> 6				
150.00	6437511.	36262	127266.	956 • 17	118.37	2105+28				
156+41	644551-6+	-3.7 0 1.9-	140877	918 • D9-	117.70	2142.70				
163.11	6451538 •	37405.	155371•	879.55	116.99	2183.90				
170+00	6457445+	-38609	1-70570+	841 → 13-	<u>116+14</u>	2228+11				
190.00	6473277 •	40981.	216379•	741 • 37	121.69	2352+90				
210.00		<del>-43480</del> ~	<del>264720-</del> -	<del></del>	<del>-1-28 + 38</del>	2482+35				
230.00	. 6499046.	46124.	31571/*	549•84	136+26	2618 • 46				
250.00	A509136	-484367-	369512	454•96-	-145.02	2762028				
270 • 00	6517281 •	51930.	426263.	359-57	154.49	2914 • 45				
2.90 -00	6523507 •-	-55120	486153	262 • 6b-	-1-64+57	3076-1				
310.00	6527779 •	58517.	549379.	164•19	175.29	3248 • 32				
330.00	<del>6536</del> 053▼	- <del>62135-</del>	<del>6:6:55*-</del>	<del>62-66-</del>	186+63	3431-05				
350.00	6530266	65987.	686725.	<del>-</del> 42.06	198+66	3627 • 41				
	6528338	—700a7	48 د-1 6-7	15 i • 55-	-211+46	3837 • 11				
390+00	6524166 •	74451 •	840309•	<b>-266•37</b>	225.05	4061+6				
410.00	6517639	<b>-79</b> 095~-	923932+	387.•85	239 • 57	4303 • 7				
430.00	6508548.	84040•	1012580.	-517·81	255.08	4564.4				
450.00	496862	<del>8-7-3-6 √</del> -	<del>-1106053+</del>	657-57-	<del>2.7-17-7</del>	484648				
2) 454.75	6493656 •	90607.	1129643*	-692 • 44	275.93	4917+69				
4/0+00	6482-1-79	94398· <sub>•</sub> -	1-206251-+-	8  -4 + 4 B-	286+80	5099+8				
490.00	6464199 .	100779•	1310672.	<b>-</b> 985•22	301 • 21	5344+91				
<u> </u>	6442706 ·-	1-06948+-	1.420163	-1-166+83	315+79 <del></del>	5607+2				
530.00	6417399 •	113416.	1535087.	-1367 • 18	331.21	5888+6				
550-00		1-20 <del>205 -</del>	<del>1-655050+-</del>	~1-587+56-	<del>348+0</del> 0	6191-05				
570.00	6353753.	127346.	1782915.	-1629.69	366.35	6519+5				
590-00		1-346-7-0	1-9-1 6 d 2 7-+-	-20971-4-	-386-29	6877-+0				
610.00	6269568 .	142799+	2058200.	~2401.23	407+19	7268 01				
3) 614.63		1-44698	20-9-20-9-8	-2472 · 8U	412+48	7365.7				
4) 624.63	6233099 •	148816.	2165667 •	-2561 • 23	410+93	7342066				

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

	,	,	TABLE 7	•	-		
	,	ES	VB POWERED FLIGHT TABL	·s-1	•	•	
OZE	DYE	DXE	ZΕ	* YE	ΧE	TIME	,
(11)	(M/S)	(M/S)	(H)	. (H)	(M)	(SEC)	·
1699.	1 - 49	984.37	70331.			143.41	_1.\_
1702	1.53	980.76	71835.	-26.	67211.	150.00	
	I • 8 <u>0</u>	994+54	82062		23379+		<del>,</del>
1780+	2+13	904.0\$	94653.	#2* *	, 79585.	163.11	
1947	Z	871 • 79	107070		857.17,	170.00	
2076•	11.36	778 • 59 	144788.	144.	142206.	190.00	
2212	33.99	601+85	227897.	474v	129792.	210.00 230.00	
2355+	47.46	S15+31	273565	1030+	140964.	250.00 250.00	
2508+	62.13	426.94	322189	2936.	150405.	270.00	
Z670+	77.94	341+83	373958	4335	150145	2.90 • 00 2.90 • 00	
2843=	94.98	253+95	429074.	6062.	164075.	310.00	
30274		1-6392	487764	B142	165266	330-00	
3225.	132.92	71.68	550268.	10601+	170619.	350.00	
3436+	154+07	~24 + 25	616861	1-3469	1-7-1-1-00 • • • • • • • • • • • • • • • • • •	370+00	
3663.	176.83	-129+35	687434.	16775+	169621*	390.00	
3908.	201 - 44		763522	20554+	144091+	410.00	
4172*	228.05	-342+26	844297.	24846*	160382.	430.00	
4458+	256.97	-462-90	930570.	29-72-		<del>450+00^*</del>	
4530+	264+21	-492+97	951920.	30929•	. 150076.	454.75,	2)
47160	284.65		1022458	35 I-1-5+	4-1-751	470+00	
4967.	312,43	-751 - 20	1119278.	41084.	128249.	490.00	
5236• 55 <i>2</i> 6•	341+51	-91-1-27 -1088+41		47621	1-1-1-65-0	<u>'\$10+00</u>	
5320° 5838-	3/2•/5 406-64	-1080+11 	1328889.	54759+	91603.	530.00	
6176	444.14	-1498-68	* * * * * * * * * * * * * * * * * * * *	<del></del>	<del></del>	-550-00	
65460	484.93	1736+56	1542596.	71054.	40204.	570.00 	
6951•	528.93	-2008 • 36	1824670.	90468•	~29540·	610.00	
7052•	539.090	-2071 • 97		70464• 	-2751U·	610.00	3)
7035.	544.89	-2152+18	1927586.	98370	-60119·		3 <i>)</i> 4)

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

TABLE 7

S-TVR	POWERED	FT.TCHT	TABLES

TIME	91189	PHIY	РНІR	41440	PHIY	OPHIR
(SEC)	· (DEG)	(DEG)	(DEG)	(DEG/S)	(DEG/S)	(DEG/S)
1)-149-41	59-03	<del>-0.10</del>	0.08	U+08	<del>-0+06</del>	O+05
150.00 156.41	-58.99 -59.08	-0.14 -0.35	ܕ09 ————————————————————————————————————	0.06	-0.08 0.02	0.07
163+11	~59.14	-0.31	0+45	-0.01	-0.00	0.00
<del>170 -</del> 00- 190 • 00		—-ບ⊹ົດ5- 2.73	-0.33	-0.44 -0.05	0.00	
210+00		<del>-3.03</del>	<del>∪+0∪</del>	<del></del>	0+02	0+06
230+00 250+00-	-57,95 59,-92	3 • 2 8 — 3 • 4 2	U+26 -+0+47	-0.08 0.09	0 • 0 1 0 • 0 1	-0.04 -0.04
270.00	-61.86	3.50	-0.07	-0 - 1 2	0.00	0.08
<del>290+00-</del> 310•00	<del></del>	3.55- 3.59	-0+1·1 -0•67	-0 - 1 - 0 -0 - 1 - 2		-0.04
330+00-		<del>-3.63</del>	<del>-0+33</del>		<del>0.0</del>	-0+08 -0+05
350.00	-70.97 7-3-39	3 • 65 3 • 67-	-0 • 2 2 0 •-2 2	-0 · 1 2		-0+0= 
390.00	-75.85	3.68	0.30	-0 · 1 3 y ·-1 3	0 • 0 0 0 • 0 0-	≠0•04 ~0•04
440+00- 430+00	<del>70+53</del> -81•13	3-69- '3,69	0.53 -0.05	-u • 13	0.00	0.06
<del></del>	<del>83.+83</del> ⇔84.48	<del>3.69-</del>	<del>-0.20</del>			<del>~D~05</del> ~0•05
470 v 00-	<del>86*47</del>	3 • 6 9 3- <sub>•</sub> 8-1	÷0 +-7·4	0 - 1-4-	000-	-0.06
490+00 510+00-	-89.23 92 <del>-24</del>	3 • 6 3 3 • 5 2-	U • 29 0 • 0 2	-0.08 	-0.01 -0.01	0•05 ~0•05
530.00	-95.57	3.52	~0 • 45	-0.16	0.00	0.06
<del>550.00</del> 570.00	-101.32	<del>3-54-</del> 3-52		<del>0-1-4</del> -0 • 1 4	-0.00 -0.00	<del>~0+0</del> 6 <b>~0</b> •06
<del>5</del> 90+00-		<del>3-,-3-9-</del>	-0 ·-2 6		0-08-	-0+11
610.00 3) 614.63-	-105.88 105.87	3.37 3-37-	~u∘20 - <del>~</del> U•36	0 • 0 2 0 • 0 0	0 • 0 0 	#0+04 #0∗03
4) 624.63	-105.36	3.36	-0.69	0.00	-0.00	-0+03
		- And Annual Confession of the Annual Confessi				
1) 90% Thrust						<del> </del>
2) Mixture Ratio S		•			<del>"</del>	
Guidance Cutoff     Orbit Insertion		-		****	*******	

*******		7	i i	TABLE 7	William Communication and Communication Comm	a militaria systema (spirate and spirate a	3
		•	· s-IV	POWERED FLIGHT TABLE	SS		
,	зніт	СНІВ	CHIY	CHIR	LONG	DECL	LATT
	(SEC)	(DEG)	· (OEG)	(DEG)	(056)	(DEG)	(DEG)
	73147-41	54-32	·····	() () ()	-79-88	28.54	28.72
	150.00	-59.32	0.00	3.00	-79.87	28.56	28.72
	1-5-6 • 4-1	59.32-	0-00-		<del></del> 79 + 76	28.59-	28.75
	163.11	~59.32	0.00	J.U0	-79.65	28.62	28.78
		57.57	1,75-	J•00—	<del></del>	28.65-	-28+81
	190.60	-55.40	<b>-3.11</b>	J • UO	-79+17	28.75	28.91
	<del>210+00</del>		<del></del>	J & O	78-7-2	-28.85-	-38.01-
	230+00	- 58.18	-3.69	)•Ou	-78.38	28.96	29 • 12
<del>~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~</del>	250+00	60 - 15		1+00	77 + 9 <sup>4</sup>	29.07-	29.23
	270.00	- 62 - 14	-3,93	3.00	-77 • 47	29.18	29.34
	290.00-	64.25	<del></del>	) • GD	76+9/	-29.29-	-29.45-
	310.00	- 66.55	-4.05	3+00	<b>-76+44</b>	29.41	29:57
~ <del></del>	3-3-0 + 0 0		<del></del>	)+ <del>0</del> 0	<del>7</del> 5+8 <del>7</del>	2-95-3	-29-70-
	350+00	- 71 + 22	-4.13	3.00	-75+27	29.66	29 · 82 -29 · 95
<del></del>	370.00-	73+65		) + 00	74+62	-290-79-	
	390+00	- 76+12	-4.19	1.00	-73.92	29.92	30.08
	410+00	78-81		).00	73.18	30.05- 30.19	-30 • 21 30 • 35
	430.00	- 81 + 42	-4.23	)•00	-72+38	-	-30-49-
ω ω	<del></del>	<del>0     3</del>		1-09	71-52	30+-3-	30.52
ເລ	2) 454.75	- 84 + 75	-4.24	1.00	-71 • 30 70 • 60	30.36 30.47	-30.63
····	470 v O O	37 - 21		) + O O	-69+62	30.60	30.77
	490.00	- 89.46	~4.19	)•00 )•0 <del>0</del>	68•59	-30.74-	-30+9D-
-		92-59		1.00	-67 • 49	30.87	31.04
	530.00	- 95.90	-4+12	ነ • ጋር			-36
	<del></del>	<del></del>		1 • GO	-65.08	31.12	31.29
	570.00	- 101.64	•4.13 	1 + O O	63.75	-31-24-	-3-1-+40
	590 <del></del>	<del>-1+∪5~⊹5−</del>	• -	)•00	-62.33	31.34	31.51
	610.00	- 106.03	-4:01	) <del>, 0</del> 0		-3·1·- 3·7·-	-31-+53—
	<del>-3)6-1-4-6-3</del>	<del>-106:03-</del> -106:03		1.00 ×-00	+61+24	31.41	31.58
	4) 624.63	-100103	-4.01	7400		~ · · · · · ·	-,

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Tnaertion

TABLE 7

		, s-IVB	POWERED FLIGHT TABLES	3		, ,
TIHE	HACH	DAAG ,	, ALP	999	VR .	TACEL
(SEC)	(U)	(N) .	(DEG)	(N/M2)	(M/S)	(M/S2)
44444444444444444444444444444444444444		*		14	3 4 4 4 - D	
	<del>-6+52-</del>		-1-4-	58		6+3
150.00	6 + 54	1062.	1.59	3**	1963+71	7•3
156.41	-6 + 8 B	-466	-2-97-	.99	1978+32-	7•4
163.11	7 . 26	193.	4 • 4 1	38•	1997+41	7.4
170.00	-7-+60	7.5	-6-18-	13	20Z0+64-	8 • 0
- 190·00	6 • 90	7.	13.67	1 *	2094.63	
210+00-	-5-78-		1-60-0	-0+	21-8-70-1-	8 • 8
230.00	4.53	0.	17-64	<b>0</b> •	2291.74	
250+00-	-4.02-	———Q +——	18.67-	0*	2410.89-	7+1
270.00	3 • 85	Ü+	19.42	0.	2544.19	9.4
290.00	-389	<del></del> 0	49.25-	O +	2692+05-	10.0
310.00	4 - 00	0.	14.77	0 *	2855.01	10+61
330,00	-41-6		4-7-a-5-7		3032+98-	
350.00	4 - 35	0.	19.02	Q •.	3227.43	11.7
370.00	-4-58-	0 <b></b>	-1 d + 35	0.+	3439.00-	12.4
390.00	4 • 84	0.	17+98	<b>U</b> •	3668.67	13 - 2
410.00	-5-1-3		44431	Q- <del></del>	391.9.13-	1-9 = 1 :
430.00	5 - 45	0.	15.09	ប•	4191.57	15+1
450.00	-5-80-	<del></del> 0	43469-	, <b>a</b>	4488-13	
- 2) 454.75	5.09	٥.	13:34	ů∗	4503.97	16+7
430+00-	-61-2-	<del>0</del>	12.03-	-0-+	47.62.033-	-13.5
470.00	6+44	0.	[1+1]	0.	5032.94	14.5
610.00	-6-4-2-			<u>a.a</u>	5325·4 <sub>1</sub>	15.5
530.00	7.18	. 0.	7 • 46 '	D.	5643.51	16.8
550.00	-7-62-			Q	5990.HZ	_18.2
. \$70.0Û	8 • 10	Ö.	4.35	0 •	6370+37	19.9
S90 <b>-</b> 00	-8+63-		-2-39-	_0	67.89.0B	22.0
610.00	9.22	0.	2 + 40	0 *	7254+37	24+5
3) 614.63	9.37	ŏ	2 - 59_	0.		25.2
4) 624.63	9.38	0,	3.36	ů•	7376.37	0 • 0
4/ 027883	7 = 20	**************************************		-		<del></del>
	Antonyou whang name and ship and	gyespensakehohii/1999/dd	<del></del>	Proprietary property contracts	-	pinalaripinaharipina a
1). 90% Thrust		•				
2) Mixture Ratio Shi	K E			,,-		
3) Guidance Cutoff						-
4) Orbit Insertion	· **:	- Company	and the second s	the state of the s		

### TABLE 8

TIPE ALT RRR /TH*	VVV*
(SEC) (M) (M) (DEG	(M/S) (DEG)
624,63 227951. 66U0287. 98.0	7787.41 86.31
.830.00 228645. 6601328. 89.5	7787.89 94.84
1,30.00 230210. 6603267. 89.6	7785.63 172.68
1239.00 232661. 6606664. 89.8	
1430.00° 256J21. 6611591. 89.8	
1630.00 240291. 6617150. 89.7	
1836,00 245413. 6623203. 89.7	7764.82 120.81
2030,00 251231, 6629393, 89,7	
223c, 30 257475. 6635575. 85.70	7750.63 121.06
2430.00 263768. 6640829. 89.3	
2430.00 269654. 6645483. 89.89	
2830.00 274647. 6649117. 89.89	
393%00 278298* 6651568* 89.53	7729.20 104.24
3230-00 283251: 6652736. 89.98	
3430.00 283297. 6652578. 90	
3630.00 278406. 6651106. 9C.C	
3830.00 274735. 6648390. 90.12	
4930+00 259605+ 6644548+ 90.18	
4£30.00 263461. 6639754. 90.19	
4430.00 256818. 6634232. 90.21	
4630-00 250189- 6628254- 90-23	
4830.00 244023. 6622124. 90.22	
5333,00 238666. 6616178. 90,21	
5230.00 234335. 6610757. 90.19	
54305 231115. 66 6185. 90.15	
5636.00 229003. 6672751. 90.17	
5656.80 228714. 6602246. 90.09	7786.96 74.09
5830.00 228065. 6600812. 90.13	
643.400 228843. 6661139. 89.95	
6230.00 231334. 6603804. 89.87	7792.29 96.46
6387.80 234453. 6637499. 89.01	7788.79 102.59
6430.00 235455. 6608709. 85.79	7787.54 104.13
6637.00 241124. 6615590. 89.73	7780.45 110.61
6830.00 247841. 6623680. 89.68	
7333.00 255642. 6632717. 89.60	7761.84 119.09
7230.00 264276. 6642183. 89.65	7751.24 121.08
743).00 273382. 6651544. 89.66	7743.47 121.64
7630.00 262499. 6660287. 89.69	7733.69 126.86
7830.00 291085- 86.7054. 86.76	ከደብቁ የተመቀው ነው። የደብቁቀውን ገጽሞትሮሥ

TABLE 8

			0,40	ANNUAL AMMONIA DECARAGO		
	TIME	ALT	RRA	V T h *	<b>V V V *</b>	
	(SEC)	(M)	(8)	(DEG)	(M/S)	(DEG)
	8130.00	298559•	4171347	,	****	
	8430.00	3:7993.	6674163.	89.83	7712.79	114-87
	8630.00	3:9132.	6681113. 6681550.	89.95	7703.39	103.16
	8950.00	37618.	8679920.	90.02	7702.51	95.55
	9330.00	3.3515.	6676512.	90.10	7704.33	87.45
	9230.00	297101.		90.17	7708.78	79.57
	9430.00	298951.	6670902 <b>.</b> 6603956 <b>.</b>	90.23	7715.60	72.57
	9630.00	279368.	6655835.	90.28	7724.37	66.86
	9830.00	209342.	6646855 <b>.</b>	90.32	7734.57	62.61
	10030.00	259453.	6637353.	96.34 90.34	7745.58	59.84
	10230.00	250323.	6628390.	90.33	7756.75	58.50
1)	10495.00	247224.	6617317.	90.28	7767.44	58.57
-/	10495.00	240224.	6617317.	90.28	7779.86	60.87
	10630.00	236189.	6612467.	90.25	7779.86 7785.10	60-87
	10830.00	231705	6606628.	90.13		63.04
	11030.00.	229,86.	66U2707.	90.11	7791.19 7795.06	67.54
	11230.00	228289.	6600958.	90.02		73.57
	11430.00	229225.	6671503.	90• 92 99• 94	7796.58	80.91
	11635.30	231784.	6604311.	39•36 39•36	7795•71 7792•53	89.07
	11845.00	236101.	6609523.	89.78	7780.89	97.32
	12032.00	241398.	6616022.	89.12		135.24
	12240.00	248452.	6624501.	89• v8	7779 <b>.</b> 92 777J <b>.</b> 76	111.28
	12432.50	255999.	6633197.	39.65	7761-21	116.26
	12643.00	265515.	6643334.	89.65	7753.18	119.40
	12832.01	.273761.	6051911.	89.67	7739.88	121-26
	13040.00	283194.	6669369.	89.77	7729.18	121-61
	13232.00	291343.	0668376.	89.75	7720.29	120.54
	13440.50	298949.	6674338.	89.81	7712.29	118.20
	13032.00	3)4313.	5678422.	89.88	7705.86	114.15
	13840.00	317632.	6680876.	89.95	7703.48	[09.00
	14332.00	308662.	6681v34.	90. 73	7702.89	102.31 94.61
	14240.60	376876.	6679147.	90.11	7705.06	86.17
	14432.00	302632.	6675525.	90.17	7709.56	78.68
	14640.00	295765.	6669759.	90.24	7716.84	71.57
	14832.00	257719.	6662987.	90.28	7725.39	66.28
	15940.00	277807.	6654439.	90.32	7736.07	62.07
	15232.00	209203.	6645815.	90.34	7746.54	59.60
	15440.00	258)19.,	6636157.	90.34	7758,19	58.42
	15632.00	249394.	6627415.	90.33	7768.34	58•67
	15840.00	241395.	6618650.	90.29	7778.16	60.44
	16032.00	235559.	6611691.	90.24	7785.59	63.49
	1624 1.30	231599.	6635802.	90.17	7791.80	68.40
	16432.00	229745.	6632229.	90.10	7795.31	74.38
	16814-80	228928.	6601201.	89.94	7795.83	89.35
		3 ··· = 4	<del></del>	~~.	1172802	Q - + 5 Q

<sup>1)</sup> S-IVB/CSM Separation
2) End of Nominal IU Lifetime

### 37

## MARSHALL SPACE FLIGHT CENTER

TABLE 8

	EMIT	MASS	THRUST O	RBITAL FLIGHT LISTING VTHE	AZI	VVVE
	(SEC)	(KG)	(n)	(DEG)	(CEG)	(2/4)
	u24.63	30673.74	٥.	90.00	86.11	7377.54
	830.00	30458.08	€.	89.94	95.11	7278.Cl
	1030.00	30377.70	2.●	89+89	103.40	7376.13
	1230.00	30319.77	ŗ.	89-84	110.56	7373.18
	143 .00	30276.67	<u>}</u> •	89.37	116.23	7369.14
	1630.00	30242.05	ĵ.	89.77	123.427	7363.97
	1830.00	30213.47	ņ.	89,76	122.72	7357.73
	2030,00	30204.96	Ç.	89.76	123.61	7350.60
	2235.00	30203.17	٥.	89.77	122.99	7342.89
	2437.07	30201.39	<u>.</u> *•	89 + 80	120.84	7335.08
	2630.00	3,199.60	?•	29 <b>-</b> 84	117.13	7327.74
	2830.00	50197.81	હ્•	85 • 88	111.64	7321.48
	3337.30	30196.02	0.	59.93	105.06	7216.88
	323 / 50	3)194.23	1 <b>.</b>	89.98	97.08	7214.41
	3430.30	30138.28	ý•	9¢•¢3	88.48	7314.33
	3630.00	30107.44	# <b>•</b>	90.08	80.02	7316.69
	3830.00 4330.30	30194.72 39131.99	0.	90.13	72.41	7321.28
			٠,٠	90 - 17	66.12	7327.69
	4230.00	37399.27 30.96.55	ù•	90.20	61.38	7335.36
	4439430		ζ.~	95 - 23	58.22	7243.62.
	4633,30 4830,53	30:93.63 30:91.11	4. 0.	90 <b>-</b> 24 90 <b>-</b> 24	56.62 56.53	7351+62
	5030,00	30088.38	5.	90.24	57.98	7355.40 7365.55
	523: .00	30185.66		90.2	61.CQ	7271.20
	5430.00 5430.00	30782.94		90.16		7375.06
	5630.00	30 182 • 94 30 183 • 22		90.18	65.64	
Į)	5660,80	3079.72	• J.a.	90.10	71.67 73.18	7277.57 7277.69
47	5834,00	29323.82	)30 <b>.</b>	90.13	75.10 75.51	7384.55
	6730.00	29216.36	81.	89.94	86.08	7384.94
	6230.00	29155.68	45.	89.86	96.82	7382.30
2)	6387.80	29123.92	88.	89 <b>.</b> 79	103.30	7379.02
٤)	6430.00	29123.38		89.78	103.30	7377.85
	6630.00	29120.82		39.71	111.81	7371.27
	6830.00	29118.26	ů.	89.66	117.15	7262.12
	7030.00	29115.70	o.	89.64	122.88	7353.63
	7236.00	29113.14	,,	89.63	123.01	7343.10
	7430.00	29110.58	3.	89.64	123.62	7331.56
	76.0.00	29108.02	ò.	89.68	122.73	7320.77
	7830,00	29105.46	õ.	89.73	120.33	731C.17

- Begin LOX dump End LOX dump

B ALEAT

			ORB	ORBITAL FLIGHT LISTING			
	TIME	MASS	THRUST	VTHE	AZI	VVVE	
	(SEC)	(KG)	(N)	(CEG)	(CEG)	(2/4)	
	,,,,,,		•				
	8050.00	29102.90	۵.	89.79	116.37	7300.51	
	8430•00	29192.70	ö.	89.94	103.92	7289.13	
	8630.00	29395.22	<b>3.</b>	902	95.87	7287.66	
		29392.66	ð.	90.10	87.31	7285.5C	
	8830.00 9030.00	29090.11	ř.	90.18	78.97	7294.56	
	9231.50	29387.56	3.	90.24	71.55	7302.50	
		29085.01	F•	90.2)	65.47	7312.72	
	9430.05 9036.00	29382.46	J.	90.34	6J.94	7324.46	
		29379.91	č.	95.36	57.96	7236+85	
	9830.00 10030.00	29077.36	õ.	90.36	56.53	7349.04	
		29174.81	0.	90.35	56.61	7360.25	
4.5	17237•30 10495•30	29071.43	Ž.	90.30	59.19	7372.59	
1)	13495.30	14212,64	ō.	97.30	59.09	7372.59	
	13630.00	14166.01	0.	90.25	61.41	<i>7377</i> <b>-</b> 50	
		14162.98	<b>.</b>	90,19	66.23	7382.90	
	13830+30	14159.35	ž.	90.11	72.63	7386.02	
	1103 > 30	14155.73	o.	90.32	80.40	7386.92	
	11230.00	14152.10	Ď.	89.93	89.02	7385.72	
	11430.00	14146.47	Ď.	89.35	97.73	7382.56	
	11637.00 11847.00	14144.66	ń.	89.77	106.10	7377.28	
	12032.00	14141.17	5.	89.71	112.52	7370.83	
		14137.40	ű.	89.66	117.84	7362.25	
	12240.00	14133.92	o.	89.64	121.21	7353.C6	
	12432.00	14133.92	à l	F9.03	123.21	7342.05	
	12340.00	14120.66	Š.	89.65	123.60	7331.34	
	12832.00	14122.89	5.	39.69	122.45	7319.73	
	13040.00	14119.49	ĵ.	89.73	119-94	7309.67	
	13232.00	14115.63	٠ <u>٠</u>	89.80	115.61	7300.23	
	13449.00	14112-14	ÿ• 0•	89.87	110.12	7293.54	
	13632+00		Ň.	89.95	102.71	7289-12	
	13840.00	14138.37 14134.89	ù.	90.03	94.37	7288.C4	
	14 -32 - 00		ì.	90.11	85.95	7290.30	
	1424 1- 00	14171.11	o.	90.18	78.03	7295.46	
	14432,00	14097.63	0.	90.25	. 70.49	7303.97	
	14640.00	14'93.86	ÿ <b>.</b>	90.31	64 - 85	7313.93	
	14832.00	14190.37	7.	90.34	60.36	7326.20	
	15041.00	14 (86 - 60	o.	90.36	57.71	7338.C5	
	15232.00	14183.11	1.	90-36	56.44	7350.59	
	1544).00	14)79.34	0.	90.34	56.72	7361.17	
	15632.00	14.75.86	9.	90.31	58.62	737C+94	
	15840.00	14772.08	0.	90.25	61.90	7378.C3	
	16032.30	14368.60	U.s. Sta	90.18	67.14.	7383.40	
	16240+00	14,64.83	 Q.	90.10	73.49	7386.20	
	16432.00	14361.34		89.94	89.31	7385.86	
	16814.80	14 154 60	٥.	U767T	V / V / A		

38

<sup>1)</sup> S-IVB/CSM Separation 2) End of Nominal IV Lifetime

# TABLE 8 ORBITAL FLIGHT LISTING

TIME				EX ,	CY	'0, Z
(SEC)	(M)	(M)	(4)	(7/5)	(*/\$)	श्रीकेंग्रेड <b>ा</b>
624.63	6233399*	148316*	2165667.	-2561.23	410.53	የሚደታ <u>ይ</u> ቀነ
830.00	5530759.	228327	3595126.	~4244 <b>.</b> 85	359.58	7342.68
1030.00	4536999.	293523.	4788861.	-5646.65	289.35	6519.45 5352.34
1230.00	\$292971.	342993.	5717461.	-6735.97	203-04	3691.71
143 .00	1867466.	373981.	6331338.	-7453.12	105.37	2219.25
1030.00	339152.	384739.	6597244.	-7759.60	1.68	426.22
1830.00	1207922.	374627.	65.1338.	-7640.14	-102.38	1362.12
2133.00	2689025.	344144.	6049758.	-7103.29	-201.11	3112.21
223 4.00	4023466.	294594.	5268107.	-618C.59	-289.18	4667.86
2430.00	3138985.	229531.	4199855.	-4924.35	-361.83	5965.42
2630.00	5975586 *	151453.	2903775.	-3404.29	,-415.18	6936.16
2830.00	6488629.	64406.	1450597.	-1703-47	-446.39	7529.53
3330.00	6651)28.	-25516.	-00822.	86.28	-453.79	7715.39
3230.00	0454517.	-114989-	-1607761.	1869.63	-436.94	7485.08
3430.00	7909947.	-198726.	-3047922.	3551.48	-396.66	5851.54
3639. 0	5046662.	-272221.	-4323692.	5042.57	-234.56	5646.44
3830.00	3913979.	-331486.	-5366139.	6263.21	-254.96	1528.58
4030.00	2563849.	-373257.	-6118609.	7147.35	-16C.77	2961.EC
4230.00	1077783.	-395170.	-6539768.	7646.05	-57.26	1230.27
4430.00	466892.	-395903.	-6635933.	7730.33	50.10	572.45
4630.00	1986200.	-375258.	-6312520.	7393.23	. 155.55	1345.C4
4830.00	3397107.	-334192.	-5674549.	665C.75	253.3 <del>6</del>	1C C2.13
5030.00	4622009.	-274783.	-4726328*	5541.52	338.10	3440.05
·5230.00	5593193.	-200142.	-3518273.	4125.09	454.55	1582.25
5430+00	6256690.	-114235.	-2117182.	2478.81	450.20	1264-18
5630.00	6575431.	-21090.	~599633.	693.53	471.C4	1741.24
5665.80	6594762.	-4335.	-31423	357.68	472.10	1764.41
5830.00	6531422.	72>32.	951832.	1131.02	466.58	1657.51
6337.00	6126905.	163458.	2451388.	2894.38	436.25	1224.45
6230,00	5384233*	245472.	3815750.	4496.78	381.65	:352.35
6387.80	4586398*	301335.	4747204.	5588.73	223.69	415.39
6430.00	4344584*	314043*	4969385.	5849.29	306.41	132.03
6630.00	3066)13.	366958.	5850721.	6877.11	214.07	632.48
6830.00	1619095.	399310.	6410309.	7525.29	109.94	\$38.5C
7030.00	63477*	410529.	6619473.	7760.50	-C • 27	144,66
7230.00	1456858.	399398.	6468125.	7572.77	-110.55	850.08
7430+00	2918288	366703.	5965916*	6975.60	-214.54	348.65
<b>7</b> 630.00	4222130.	314189.	5141432.	6004.53	-207.86	858.51
7830.00	5298804.	244653.	404035	4714.53	-384.41	101-55

### TABLE 8 ORBITAL FLIGHT LISTING

			U.	KOTIWE EPIGUT PROTEGO			
	TIME				CX	CA	ÇΖ
	(SEC)	(M)	(M)	(4)	(8/9)	(2/4)	(2/4)
	(350)	<b>,</b> 10.7	<b>,</b>				
	8030.00	091380•	161785.	2722749.	-3176.48	44C.6C	-7C14-48
	8430.00	-6675592.	-25956.	-270306.	305.56	481.47	-7682.26
	*8630.00	-6437332.	123924.	-1785845+	2066.32	464.CC	-74C5.65
	8820.00	~585639C•	2)9913.	-32.6239.	3717.41	421.93	-6734.95
	9030-00	-4963459.	288187.	~44558Cl.	5172.48	357.28	-57C4.65
	923 2-00	-3805022.	351531.	-5467593+	6354.6l	273.26	-4367.46
	9430.00	-2444104.	396-76.	-6186875.	7199.95	174.06	-2752.18
	9630.00	-951291.	420478.	-6574.39.	7661.20	-64.74	-1060.82
	9830+10	592864.	422:76.	-6606394.	7710.47	45 • C C	735.07
	10030.00	2104960.	100989.	-6282154,	7341.46	161.10	2498.83
	10230.00	35 /2614 •	358155•	-5615957.	657J.70	265.45	4132.82
1)	10495.00	5048419.	271712.	-4269487.	5000.48	381.95	5947.74
1)	10495.00	5048419.	271712.	-4269487.	5000.48	381.95	5947.74
	12630.00	5657366.	216908.	-3416414.	4001.72	428.32	6664.12
	10830.00	b294746%	125916.	-2001960.	2342.17	477.49	7415.48
	11030.00	6585365.	-27037.	-477431.	557.28	500.82	7759.47
	1123 00	6512717.	72588.	1073261.	-1273.57	496.83	7675.79
	11430.00	60806204	109274.	2564608.	-3027.34	465.58	7168.79
	11430.00	5313018.	257117.	3914417.	-4612.92	408.72	6267.17
	11840-07	419236V+	33453.	5098413.	-6000·55	324.86	4952.13
		2944245.	388498•	5912068.	~6950.29	230.77	3488.16
	12032.00	1425605.	424322.	0455356.	-7579.20	115.80	1710.87
	12240.00	-53315•	435813•	6618651	-7761.19	3.38	-16.61
	12432.00	~1657287.	423784.	6427784.	-7518.80	-118.48	-1875.86
	12040.00		390668.	5903345	-6904.44	-225.04	-349C+52
	12832.00	-3049693.	332::92•	5012282	-5855.54	-327.81	-5034.43
	13040.00	-4374173.	262154.	3929575.	-4587.36	-496.03	-6196.30
	13232.00	-,38J795.	170790.	2539527•	+2964.84	-468.26	-7104.22
	134400	-6169962.	77263.	1124293.	-1316.92	-502.01	-7576.90
	13632.30	-0382653.	-28593•	-463852.	535.48	-510.97	-7667.84
	13840.00	-6654272.	125398.	-1917456.	2219.37	-493.30	-7359.82
	14/32-00	-6398739.	223001.	-337935?•	3918.10	-446.97	-6619.42
	1424 ) - 30	-5757405.		~4555828.	5290.15	-381.15	-5595.20
	14432.00	-4869800.	303492 <b>.</b> 373476 <b>.</b>	-5575839•	0484.14	-288-43	-4174.C1
	14640.00	-3639377•		-6233899•	7257.46	-187.32	-2641.19
	14832.00	-2314744.	419349•	-6596702.	7691.23	-66.64	-829.02
	1504 1.00	-752492-	445938.	-6590338.	7694.47	49.30	896.22
	15232.00	730726.	447594	-6212534.	7263.70	173.04	2719,97
	15440.00	2294103.	424376.		6483.23	279.38	427C • 44
	15632.10	3619437.	383774.	-5538711. -4495551.	5266.83	379.55	5711.06
	15840.00	4847516.	31192).		3860.78	452.92	6745.83
	16032.00	3727663.	231674.	-3294642.	2115.06	577.07	7482.C9
	16240.00	6352334.	131343.	-1897548.	386.38	530.72	7767.62
	16432.00	6593532.	-31294.	-337322+	~3033.92	498.13	7163.95
2)	16814.80	6078055.	169012.	2569930.	-3033,42	∃ /V♦AJ	,1030.3

S-IVB/CSM Separation
 End of Nominal IU Lifetime

#### TABLE 8

		U	REITAL FLIGHT LISTING			
TIME	ΧE	YΞ	ZΕ	CXE	CYE	cze
(SEC)	(M)	(M)	(r)	(2/4)	(M/S)	(8/5)
,						
624.63	-50119.	98370+	192758:	-2152-18	544.89	7035.57
830.00	-663121.	218661.	3308857.	-3694.07	621.31	6356.33
1737.00	-1537-)92.	347509*	4486762.	-5009-15	66C.65	5373.54
12303	-2650117.	480 <i>9</i> 65•	5440746.	-6374 <b>.</b> 71	657.25	4126.70
1430.00	-3946819.	657197.	6124708.	-6838.43	605.77	2678.27
163 .00	-5362875.	719020.	6503533.	-7263.54	4.204.13	1102.21
1830.30	-0828327.	805613.	6561682.	-7330.65	252.55	~521.58
2030.00	-8271141.	857709.	6296949.	-7038.47	160-54	-2113.16
2230.00	-9620825.	867518.	5723868:	-64C3.54	-67.27	-3552.80
243: •◊፻	-1.511382.	829295.	4877085.	-5459.00	-217.58	-4888.55
2032-30	-11786937.	739934.	37646∂∜*	-4252.54	-576.11	-5525.48
2850.00	-12499382.	599357.	2519889.	-2843.65	-827.C2	-6695-60
3070.00	-12915442.	413735.	112834	-1361.63	-1053.82	-7122.E2
323 -00	-13015606.	180017.	-310016.	302.71	-1240-36	-7202-11
3430-00	-12795+08.	-81733.	-1729131.	1890.30	-1371.64	-6531.39
3630.00	-12265554+	-363703.	-3060132.	3386.99	-1435.71	-6324.63
3830.00	-11451423.	650876.	-4238511.	4721.72	-1422.63	-5411-25
4335.00	-1:391953.	-927233.	-5267131.	5830.47	-1327.20	-4235.48
423 - 10	-9137932.	-1176272.	-5918942.	6659.19	-1148.63	-2853.64
4437-00	-7749737.	-1381432.	-63,39293.	7166.47	-E91.15	-1332.76
4030.00	-6294577.	-1528.16.	-6447719.	7325.78	-564.1C	252.78
4833.00	-4843316.	-16.3378.	-6239078.	7127-28	-181.82	1824.72
5030-00	-1407;39*	-1598293.	-57239584	6578.73	236.58	-22.04.74
5230.00	-2233502*	-1507u23.	-4928263.	5705.54	67CalC	4618.69
5430()	-12.3691.	-1330928.	~389197 <u></u> u.	4549.78	1093+04	5700.56
5630.JO	-428576*	-1072771.	-2667149.	3168-12	148C.51	£496.12
5666.80	-317101.	-1717763.	-2425946.	2894.56	1945.56	6667.93
5830.00	\$3394.	-742489.	-1314351.	1631.78	1809+25	6571.05
6 3	218553.	~354438.	984'L& <b>.</b>	12.44	2054.66	7093.35
623 .30	58528.	72845.	1567123.	-1696+36	2198.97	6861.67
6387.80	-292361.	423572.	2552524.	-2829.82	2232.67	6438.66
6430.00	-418405+	517768.	2821261.	-3143.38	2229.13	6891.42
663.400	-1188101.	956732.	3996972.	-4521.94	2138.77	5414.18
6837+00	-2212034.	1365329.	49700204	-5674.77	1927.38	4277.56
7030.00	-3439231.	1720086.	5694701.	-6546.52	1602.15	2941.45
7230-00	-4809157.	1999503.	6137879.	-7096.62	1176.78	1474.55
7437	-0254794*	2135428.	628^736.	-7301.03	670.96	-49.24
7630.00	-7736326.	2264162.	6119403.	-7152.88	109.34	-1554.56
7830.JO	-9093077.	2227324.	5664797.	-6662.25	-479.83	.2970.32
			•			

#### TABLE 8

	TIME	Xξ	ΥΞ	ZE	CXE	:Y €	ΕZE
	(320)	( M )	(M)	( M )	(2/4)	( 2 / 4 )	(+/5)
	8330.00	-1)349787.	1072420•	1941741.	-5855.C4	-1066.15	· ~4225.CE
	8430.00	-12243282.	.429256.	2849667.	-3462.78	-2107.06	-6058.13
	8630.00	-12790829•	966431.	1583875.	-1991.60	-2503.78	-654E.C5
	8830.00	-13333333-	435477.	251408.	-424 - 18	-2784.71	-6723.26
	9030-00	-12958986.	-138441.	108399 .	1165.23	-2930+72	-6577.52
	9230.00	-12573662.	-726927.	2358701.	2703.30	-2928.77	-6118.51
	943.1()	-11885927.	-1299665.	3512142.	4117+57	-2772.65	-5369.62
	9635,00	<b>-</b> 1J936431•	-1825903.	489521	5340.11	-2464.75	-4365+21
	9850.00	-9766706.	-2276369.	5244437+	6310.64	-2014-49	-3152.65
	10.03 1.00	-8432304.	-2023424.	5741146.	6979.70	-1440.36	-1793.73
	10230.00	-6997339.	-2345678.	5956421.	7311.60	-768.43	-351.11
1)	11495.01	-5060419.	-2923483.	3794055.	7201.17	215.69	1565.81
	10495.00	-5060419.	-2920483.	3794055.	7201.17	215.69	1565.81
	170311,07	-41:6545*	-2856494.	5519322.	6903.67	732.29	2496.12
	1:830.00	-2792827 <b>.</b>	-2634483.	1891372.	6178.94	1481.97	3759.22
	11/37.07	-1655464.	-2267443.	+028506+	5147.33	2175.68	4829-57
	11230.00	-753987.	-1770016.	1975616.	3859.80	2773.47	5654.77
	1143".))	-124327.	-1166042.	.785596.	2383.86	3239.70	6195.54
	11630.00	193547.	-48546) <b>.</b>	.518007.	784.97	3545.37	6427.78
	1184 .00	178340•	275914.	827762.	-928.62	3671.16	6331.24
	12. 32.00	-147321.	976890.	2011801.	-2450 • 17	3699.95	5946.34
	12240.00	-817739.	1730763.	317870J.	-3966.65	3325.31	5235.53
	12432-00	-1698934.	2300626.	+101641.	-5177.10	2896.77	4344.41
	12647.00	-2889503.	2840232.	+886969	-6219.41	2265.69	3176-69
	12832.00	-4152667.	3208948.	3382311 <b>.</b>	-6883.81	1557.07	1966.79
	13.40.00	-5631955.	3444184.	5647891.	-7263.95	691.40	579.28
	13232.00	-7031832.	3495786.	3634888.	-7273.25	-158.61	-711.30
	13440.00	-6514400.	3365909•	5345768.	-6920.50	-1087-56	-2053.61
	13032.30	-9786053.	3077339•	.84 <sup>7</sup> 61.	-6277.15	-1909.50	-3185.44
	13847.00	-1,992216.	2594437•	+004958.	~527J <b>.</b> 63	-2715.02	-4240.33
	14/325	-11395)52.	2910917.	3173152.	-4099.30	-3341.67	-5C14.43
	14240.00	-12597748.	1259057.	2064362.	-2627.89	-3856.53	-56CC.88
	14432.30	-12951330.	487197.	957444.	-1144.97	-4153.29	-5887.53
	14640,30	-13 26617.	-391791.	-274731.	520-88	-4260.03	-591C.C5
	14832.50	-12779757-	-1252407.	(39೧೮೧७. ⋅	2041.98	-4150.25	-5665.63
	15040.00	-12191368.	-2033616.	2517203.	3592.01	-3875-57	-5127.20
	15232.00	-1:376669.	-2717853.	3435209.	4863.50	-3299.60	-4401.29
	15440-00	-1,242362.	-3327289.	12574604	5993.75	-2539.94	-3405-13
	15632.00	-9011669.	-3736568.	1800492.	6769.80	-1703.32	-2335.66
	15840.00	-754á014 <b>.</b>	-3986536.	5159201.	7260.40	-685.07	-1071-47
	16032.00	-5139093•	-4022917.	5249090.	7371.20	310.91	136.93
	1624 (-07)	-4623435.	-3845234.	5085590.	7109.09	1393.15	1426.39
	16432.37	-33)9244.	-3485474.	4703374.	6527.73	2341.98	2541.62
2)	16814.80	-1103034.	-2273 66.	3361865.	4513,95	3898.77	4356.00

<sup>1)</sup> S-IVB/CSM separation
2) End of Nominal IU Lifetime

### TABLE 8

			ORBITAL FLIGHT LISTING				
TIMO	PHIP	PHIY	PHIR	Leng	CECL	LATT	
(SEC)-	(DEG)	(DEG)	(CEG)	(CEG)	(CEG)	(EEG)	
624,63	-135.86	3.36	-3.69	-61.24	31.41	21.58	
830,00	-123.12	2,64	-1.98	-45.83	31.27	21,44	
1330,00	-136.64	2.13	-2.55	-31.19	29.20	29,36	
1230 . 30 .	-150.14	1.49	-2.98	-17.42	25.44	25.58	
1430.00	-163.01	2.77	-3.24	-4.68	20.34	20.46	
1633.30	-177.56	-3.37	-3.33	7.09	14.28	14.36	
1830.00	169.52	77	-3.24	18.14	7.59	7.64	
2030+05	156.11	-1.50	-2.98	28.91	C = 6 1	C.62	
2236.00	142.73	-2.15	-2.55	39.42	-6.38	6.42	
2430.00	129.35	-2.68	-1.98	50.32	-12.10	-12.18	
2630.00	115.59	-3.08;	-1.31	"'61.84	-19.25	-15.37 .	
283 .00	102.63	-3.31	-0.56	74.22	-24.51	-24.65	
3030.00	89+29	-3.37	3.22	£7 <b>.</b> 59	-28.53	-26.68	
323⊍•00	75.496	-3.24	C.99	101.64	<b>-2C.</b> 96	-31.12	
3430.00	62.63	-2.94	1.71	116.57	-31.56	-31.74	
3630	49.31	-2.49	2 • 3 5	131.20	-30.31	-30.47	
3830√00	35.99	-1.90	2 • 86.	145.18	-27.28	-27.43	
4530.00	22.07	-1.20	3 - 22	158.22	-22.79	-22.52	
4230,00	9∙33	44	3.41	170.29	-17-17	-17.27	
4437.00	-4+22	* (0.36	3.42	-178.43	-10.77	-10.84	
4630.JO	-17.40	1.14	3.25	-167.66	-3.91	-2.54	
4830.00	-31.81	I.86	2.89	-157.05	3.12	3-14	
5030.0	-44.26	2.48	2 • 38	-146.26	10.04	10.10	
5230.00	-57.74	2.98	1.74	-134.97	16.55	16.65	
543. ასე	-71.25	5.31	C.99	-122.86	22.31'	22.44	
5630.00	-24.78	3.47	0.19	· -109.73	26.58	27.13	
5666.80	-87.27	3.48	C • 14	-107.20	27.68	27.64	
5836.00	-98.33	3.43	-0-63	-95.59	3C • 16	30.32	
6537.00	-111.09	3.21	-1.42	-80.72	31.56	21.72	
6230.00	-125.43	2.80	-2.13	-65.73	31.01	21.17	
6387.87	-136.17	2.37	-2.62	-54.22	29.23	25.35	
64300	-138.95	2.24	-2.73	-51.23	28.58	28.73	
6630.00	-152,43	1.56	-3.18	-37.66	24.52	24.66	
6830.07	-165.87	J.79	-3.46	-25.16	19.21	15.23	
7030.00	-179.28	-3.2	-3.55	-13.58	13.02	13.10 6.32	
7230-00	167.36	~v.84	<b>-3.45</b>	-2.67	6.28	°°°4 ~°°65	
7430.00	154.02	-1.61	-3.16 -2.71	7.90	-C.69	-7.46	
7630,00	140.72	<b>-2.30</b>	-2.71 -3.11	18.48	-7.62 -14.33	-14.30	
7630.00	127.43	-2.86	-2.11	29.39	-14.22	-14.50	

TABLE 8

5M17	PHIP	PHIY	ORBITAL FLIGHT LISTING PHIR	LØNG	CECL	LATT
(SEC)	(DEG)	(PEG)	(CEG)	(CEG)	(CEG)	(CEC)
3030.00	114,16	-3.28	-1.39	40.95	-20.20	-2C.32 -25.16
3430.00	87.67	-3.58	0.22	66.80	-25.CC	
3630.00	74,43	3.45	1. 4	81.C4	-31.16	-21.23 -31.68
1830-00	61.20	-3.14	1.67	95.68	-31.51 -30.01	-20.17
) )37.400°	49.80	-2.68	2.46	113.14 123.93	-26.EC	-26.55
7231-00	34.73	-2.34 -1.31	3.12 3.41	136.77	-22.18	-22.31
7433.30	21,48	-7.50	3.62	148.67	-16.48	-16.58
3630.00	8.20	1.56	3.31	155.84	-10.4e	-10.10
9831.00 10030.00	-17.94 -30.43	2.28	2.85	175.56	-3.16	-3.18
10233.03	-51.84	2.88	2.03	-178.85	3.87	3.89
17495,30	-54.53	2.79	2.11	-164.41	12.91	12.59
1:495.00	-54.53 -54.53	2.99	2.11	-164.41	12.91	12.59
1763770	~54.53	2.99	2.11	-156.66	17.20	17.30
17837.01	-54.53	2.99	2.11	-144.44	22.87	23.00
11,30.00	-54.53	2.99	2.11	-131.19	27.39	27.55
11230.00	-54.53	2.99	2.11	-116.94	30.40	30.56
11430.00	-54.53	2.99	2.11	-102.04	31.59	31.76
11430.00	-54.33	2.99	2.11	-87.07	30.83	31.00
11840.00	~47.03	-2.38	-12.32	-71.96	28.04	28.19
12 -32 - 10	10.57	-1.68	-136.39	-59.08	23.96	24.1C
12040.00	12.40		-176.33	-46.21	18.29	18.40
12432.00	-0.46	-0.00	-176.23	-35.21	12.25	12.33
12040.00	-14.36	Ú•90	-176.34	~23.92	5.19	5.23
12832.30	-27.15	1.69	-176.63	-13.79	-1.51	-1.52
13341.00	-41.99	2.45	-177.13	-2.76	-8.69	-8.74
13232.00	-53.74	3.72	-177.74	7.78	-14,96	-15.05
13440.00	-67.54	3.49	-178.53	19.92	-21.07	-21.19
13632.00	-80.26	3.74	-179.34	31.99	-25.75	-25.50
13849.00	-94.14	3.80	179.75	46.38	-29.46	-29.62
14032.00	-170.75	3.67	178,92	59.83	-31.30	-31.47
14247,00	-123.52	3.33	1788	75.07	-31.40	-31.56
14432.00	-133.22	2.84	177.39	88.89	-29.71	-29.87
146470	-146-99	2.16	176.79	103.10	-26.15	-26.30
14832.00	-159.72	1.41	176.39	115.30	-21.55	-21.68
15040.00	-173.53	₽.52	176,16	127.56	-15.50	-15.59
15232.00	173.70	34	176.14	138.20	-9.23	-5.29
15445.00	159.81	-1.26	176.33	149.31	-2.04	-2.06
15632.00	140.95	-2.04	176.70	. 159.49	4.70	4.73
15840.00	132.97	-2.78	177.30	173.60	11.81	11.69
16 -32 - 30	125.J2	-3.33	177.99	-176.19	17.92	18.03
16240.00	195.96	-3.73	178.35	-165.34	23.67	23.81
16432.00	92.95	<b>→3.90</b>	179.73	-152.48	27.84	27.59
16814.80	6å•99	-3.66	-178.53	-124.48	31.60	31.76

Manual control begins at 8995 and ends at 91.75

S-IVB/CSM Separation End of Nominal IU Lifetime

TABLE 9
;IB SPENI STAGE REENTRY DATA

	. TIPE	ALT	RRR	VTH*	VVV*	àŽ⇔
	(SEC)	(M)	(M)	(DEG)	(M/S)	(DEG)
	•					72001
	144.49	61995.	6435263.	4 2 4 2	2225 04	
	160,00	76872.	6450118.	63.42.	2325.84	75.70
	180.00	92831.	6466649.	66.62	2232-74	75.90
	293.00	105229.		70.90	2162.25	76.10
	220.00	•	6478420-	75-44	2106.96	76.30
		114083.	6487247.	80.17	2066.75	76.50
	240,00	119401.	6492538.	85.05	2042.31	76.70
	260.00	121189.	6494360.	90.00	2034.07	76,90
	280.00	119449.	6492535.	94.96	2042.21	77.10
	300.00	114179.	6487240.	99.84	2066.55	77.30
	320,00	105375.	6478410.	104.57	2106.64	77.51
	343,00	93026.	6466036.	109.10	2161.72	77.71
	360.00	77122.	6453138.	113.39	2229.96	
	380.00	57703.	6430665.	117.38		77.92
	400.00	35679.	6408618.	120.60	2295.09	78.15
	420.00	20304.	639323).		2132-45	78.68
	460.00			116.56	902.65	82.85
	480.00	11534.	6384456.	110.53	472.12	88.74
_	=	8478.	6381400.	108.46	445.51	89.33
5	500.00	5854.	6378776.	106.49	430.89	89.68
	520.00	3557.	6376479.	104.84	422,07	89.88
	• 540.00	1511.	6374432.	103.49	416.14	90.32
	556.17	0.	6372521.	102.59	413.00	90.08

TABLE 9
S-IB SPENT STAGE REENTRY DATA

	TIME	MASS	THRUST	VTHE ,	AZI	RANGE	VVVE
	(SEC)	(K(	' (N)	(DEG)	(DEG)	(M)	(M/S)
	144.49 160.00 180.00 200.00 220.00 240.00 260.00 380.00 340.00 360.00 380.00 400.00 420.00 480.00	40136 40591.38 405991.38 4055991.38 4055991.38 4055991.38 4055991.38 4055991.38 4055991.38 4055991.38 4055991.38	197848. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0	58-28 61-79 66-73 72-09 77-83 83-84 90-00 96-16 102-18 107-92 113-29 118-24 122-75 127-03 134-67 166-89 172-46 175-74	72.23 72.41 72.64 72.87 73.11 73.35 73.60 73.86 74.12 74.39 74.66 74.94 75.22 75.50 75.76 75.40 74.77	61940. 87395. 119921. 152261. 184466. 216581. 248649. 280713. 312818. 345006. 377322. 409804. 442402. 473817. 492460. 497648. 498194.	1979.29 1874.59 1790.43 1790.43 1674.33 1643.76 1643.46 1643.47 1673.46 1722.51 1789.21 1871.01 1950.86 1802.64 574.14 169.98 142.30
46	503.00 520.00 540.00 556.17	45591.38 45591.38 45591.38 45591.38	0. 0.	177.94 179.65 179.31	77.42 103.75 236.99	498587. 498629. 498621.	108.16 97.09 90.03

. table 9 s-ib spent stage reentry data

TIME	х		Z	DΧ	DY	DZ
(SEC)	(M)	(M)	(M)	(M/S)	(M/S)	(M/S)
144.49	6434124.	35609.	115707.	1002.54	118.63	2095.32
160.00	6448315.	37445.	147838.	838.32	118.35	20661
183.00	6463161.	39797.	189095.	846.70	117.15	2359.95
200.00	6474191.	42131.	230233.	456,40	116.19	2053.65
220.00	6481422.	44444.	271233.	266.78	115.16	2040+23
240.00	6484865.	46737-	312074.	77.63	114.37	2337.64
260.00	6484529	49007.	352731.	-111.25	112.92	2027.89
280.00	6480416.	51253.	353182.	-300.07	111.70	2016.95
300,00	6472525.	53474.	433401.	-489.01	110.42	2004.82
.320.0v	6460853.	55669.	473366.	-678.29	109.35	1991.47
340.00	6445391.	57837.	513052.	-868.37	107.64	1976.83
360.00	6426128.	59975.	552427.	-1058.37	106.15	1960.09
383.00	6403114.	62082.	591375.	-1239.17	104.59	1928.98
400.00	6377391.	64159.	628616.	-1261.20	103.13	1716.41
420.00	6359397.	66214.	653514.	-484.39	102.59	754.73
460.00	6348459.	70277.	673358.	-211.11	100.45	410.18
480.00	6344508.	72281.	681361_	-185.20	100.08	392.63
500.00	6341007.	74283.	689127.	-166-03	100,37	384.82
520.00	6337836*	76284.	696780.	-151.83	100.06	380.90
540.00	6334912.	78285.	704373.	-141.01	99.99	378.53
556.17	6332689.	799Cl.	710484.	-134-22	99.86	377.60

TABLE 9
S-IB SPENT STAGE REENTRY DATA

	TIME	ΧE	YE	ZE	DXE	DYE	DZE
	(SEC)	(M)	(M)	(M)	(M/S)	(M/S)	(M/S)
	144.45	61692.	-34.	62496.	1025.76	1.09	1692~75
	160.00	76267.	-7.	88380.	864.83	2.31	1663.17
	180.00	91689.	. 56.	.21580.	677.73	4.02	1657.20
	200.00	103385.	154.	54669.	491.93	5.88	1651.65
	220.00	111371.	292.	.87643.	306.79	7.87	1645.66
	240.00	115659.	470.	20492.	122.08	10.35	1639.19
	260.00	116256.	693.	53207 -	-62.41	12.27	1632.22
	280.00	113163.	962.	85778.	-246.88	14.66	1624.76
	300.00	106379.	1289.	J18194.	-431.53	17.18	1616.78
	320.00	*95899 <b>.</b> ·	1650.	350446.	-616.56	19.83	1608.27
	340.00	81713.	2074.	382521.	-802.16	22.63	1599.16
	360.00	63811.	2554.	414404.	-988.04	25.47	1588.65
	.380.00	42236.	3092.	445990.	-1165.43	28.22	1564.24
	400.00	17998.	3664.	475988.	-1188.44	27.31	1355.13
	420.00	1250.	4034.	493505.	-433.63	8-61	376.21
	460.00	-7898.	4138.	498017.	-168.03	0.13	25.67
48	480.00	-10987.	4134-	498326.	-142.09	-0.35	7.73
	500.00	-13624*	4127.	498390.	-122.64	-0.29	-j.37
	520.00	-15924.	4123.	498338.	-108.06	-0.15	-4.49
	540.00	17968.	4121.	498221.	-96.83	-0.03	-6.99
	556.17	-19473.	4121.	498097.	-89.68	0.03	-7.99

TABLE 9 S-IB SPENT STAGE REENTRY DATA

	TIME	CHIP	CHIY	CHIR	LØNG	DECL	LATT
	(SEC).	(DEC	(ces)	(DEG)	(DEG)	(DEG)	(DEG)
	144.49 160.00 180.00 200.00 220.00 240.00	-59.32 -59.32 -59.32 -59.32 -59.32 -59.32	C.00 C.00 C.00 C.00 C.00	0.00 0.00 0.00 0.00 0.00	-79.96 -79.71 -79.39 -79.08 -78.76 -78.44	28.53 28.60 28.69 28.78 28.86 28.94	28.09 28.76 28.85 28.94 29.02 29.11
	260.00 280.00 300.00 320.00 340.00 360.00 380.00	~59.32 +59.32 +59.32 +59.32 +59.32 +59.32 +59.32	C-00 C-00 C-00 C-00 C-00	0.00 0.00 0.00 0.00 0.00 0.00	-78.13 -77.61 -77.49 -77.18 -70.85 -76.53 -76.21	29.03 29.11 29.19 29.27 29.34 29.42 29.50 29.57	29.19 29.27 29.35 29.43 29.51 29.58 29.66 29.73
49	400.00 420.00 460.00 480.00 500.00 520.00 540.00	-59.32 -59.32 -59.32 -59.32 -59.32 -59.32 -59.32	C.00 C.00 C.00 C.00 C.00 C.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00	-75.89 -75.70 -75.65 -75.65 -75.64 -75.64 -75.64	29.61 29.62 29.62 29.62 29.62 29.62 29.62	29.77 29.79 29.79 29.79 29.79 29.79 29.79

TABLE 9 S-IB SPENT STAGE REENTRY DATA

	· TIME	JACH	RAG	ALP	999	VR ·	TACEL `
	(SEC)	(U)	(N)	(DEG)	(N/M2)	(M/S)	(M/S2)
50	144.49 160.00 180.00 200.00 220.00 240.00 260.00 300.00 320.00 340.00 360.00 400.00 400.00 480.00 520.00 740.00	6.36 6.49 5.58 4.835 4.35 4.35 4.35 6.45 1.75 6.61 5.75 0.32 0.32	68609. 7700. 416. 444. 11. 5. 4. 5. 11. 43. 401. 7379. 114016. 173629. 2382255. 512519. 495618. 462644. 474276. 468575.	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	14. 17. 3. 0. 0. 0. 0. 0. 3. 54. 832. 12537. 14654. 4924. 4996. 4996. 4996. 4996. 4996. 4996.	1978.27 1873.53 1789.33 1722.22 1672.86 1642.57 1632.26 1642.28 1672.29 1721.37 1788.11 1869.95 1949.85 1801.68 575.74 167.04 141.37 108.11 97.09 90.03	2.74 0.17 0.01 0.00 0.00 0.00 0.00 0.00 0.01 0.01 0.16 2.50 38.08 52.25 11.25 10.88 10.59 10.40 10.40
	i56 <b>-1</b> 7	0.26	464032.	-0.00	41304		

TABLE 10 .
COMPLETE AS-205 MISSION STATION LIST

Y		COMETETE	AS-205 MISSION S	STATION LIST		
,	STATION	GEODETIC LATITUDE (Deg. North)	LONGITUDE (Deg. East)	HEIGHT ABOVE ELLIPSOID (Meters)	APPLI FLIGHT BOOST	CABLE PHASE ORBIT
	ODOP					_
	Transmitter 1.3.7	28.444373	-80.579973	5.1	X	
	GLOTRAC.					
	Cape Kennedy Cherry Point, N.C. Eleuthera Grand Turk Bermuda	28.413386 34.881507 25.268375 21.434025 32.254488	-80.592263 -76.353404 -76.313049 -71.145182 -64.838675	11.0 3.0 31.8 18.6 87.1	X X X X	
	C-BAND RADAR					
	Mila Cape Kennedy Patrick Grand Bahama Grand Turk Bermuda Antigua Vanguard Ship Canary Island Ascension Pretoria Tananarive Carnarvon Mercury Ship Woomera Hawaii Point Arguello Resdtone Ship White Sands TELEMETRY	28.424862 28.481767 28.226553 26.636350 21.462889 32.347964 17.144031 32.700000 27.744861 -7.972761 25.943733 19.018056 24.897403 25.000000 30.819728 22.122092 34.582903 25.000000 32.358222	-80.664404 -80.576514 -80.599292 -78.267708 -71.132114 -64.653742 -61.792858 -48.000000 -15.602000 -14.401694 28.358489 47.304444 113.716080 125.000000 136.836990 -159.665380 -120.561150 -118.000000 -106.369560	12.0 14.0 15.0 12.0 28.0 19.0 58.0 0.0 36.0 143.0 1626.0 1375.0 62.0 0.0 151.0 1140.0 646.0 0.0 232.0	X X X X	x x x x x x x x x x x x x x x
	New Smyrna Beach Cape Tel 4 Mila - CIF Vero Beach Grand Bahama Grand Turk Bermuda Antigua Vanguard Ship Canary Ship Ascension Tarnanarive Carnarvon Mercury Ship Guam Hawaii Redstone Ship Guaymas Corpus Christi	29.066667 28.463713 28.542366 27.666667 26.628514 21.446016 32.348102 17.137333 32.700000 27.742797 -79.743140 -19.018056 -24.897356 25.000000 13.309244 22.125267 -25.000000 27.958406 27.65540	-80.916667 -80.653029 -80.643533 -80.350000 -78.299444 -71.147500 -64.653800 -61.774955 -48.000000 -15.598103 -14.392221 47.304444 113.716070 125.000000 114.734410 -159.667620 -118.000000 -110.720790 -97.380256	0.0 0.0 0.0 0.0 0.0 0.0 18.0 41.2 0.0 35.0 30.0 1375.0 64.0 0.0 127.0 1142.0 0.0 18.0	x x x x x	x x x x x x x x x x x x x

TABLE 11

# AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY, REVISION PERFORMANCE CHARACTERISTICS

### -S-IB STAGE

Average Longitudinal Sea Level Thrust (lbs)

	H-1 Engine	Turbine	<u>Total</u>
Engine #1	204233.	618.	204851.
Engine #2	199748.	633.	200381.
Engine #3	201948.	594.	202542.
Engine #4	198849.	596.	199455.
Engine #5	200981.	584.	201665.
Engine #6	202797.	622.	203419.
Engine #7	200659.	613.	201272.
Engine #8	202030.	618.	202648.

Total Average Sea Level Thrust (F): 1,616,123.

Flight Time Interval: 0.0 - 140.11 (IECO)

$$\dot{W} = Wt (t = 0) - Wt (t = 140.11) - * Waux / 140.11$$

$$\mathring{W}$$
 = 6210.77 lb/sec \*Waux: Frost = 1100 lbs  
 $Seal$  Purge = 6 lbs  
 $Seal$  Fuel Additive = 27 lbs  
Total = 1133 lbs

 $I_{SP} = 260.21 \text{ (sec)}$ 

#### S-IVB STAGE

High Thrust Level Flight Time Interval: 149.4 - 454.8 seconds Low Thrust Level Flight Time Interval: 454.8 - 614.8 seconds

### AVERAGE VALUES

	High Level Thrust	Low Level
Vacuum Thrust (1bs) Flowrate (1b/sec)	226019. 533.97 423.28	176087. 407.92 431.67
Specific Impulse	423.20	431.07

NOTE: The 0.4% S-IB stage bias is not included in these data.

### TABLE 12

### AS-205 L/V TRAJECTORY DISPERSION ANALYSIS, REVISION 1 VEHICLE WEIGHT BREAKDOWN (POUNDS)

Spacecraft* Instrument Unit S-IVB Stage Inert Useable Reserve Propellant Injection Weight	36,600 4,280 25,285 <u>1,452</u>	67,617
J-2 Thrust Decay Propellant S-IVB Cutoff Weight	140	67,757
S-IVB Propellant Consumed S-IVB APS Propellant Consumed Ullage Cases (214) and LES (8986) S-IVB "90% Thrust" Weight	228,164 6 9,200	305,127
S-IVB GH2 Start Tank S-IVB Buildup Propellant Consumed Ullage Propellant Consumed S-IVB Stage Weight at Separation	5 304 <u>L67</u>	305,803
Ullage Propellant Consumed S-IVB Separation Components S-IVB Aft Frame Hardware S-IB/S-IVB Interstage S-IB Dry Weight S-IB Residuals and Reserves S-IVB Frost Consumed S-IB Frost Consumed S-IB Seal Purge Consumed S-IB Fuel Additive Consumed S-IB Gearbox Lubricant Consumed Inboard Engine Thrust Decay Prpt Consumed	9 5 31 6,478 84,401 10,949 100 1,000 6 27 717 2;152	
To Separation S-IB Mainstage Propellant Consumed Vehicle Liftoff Weight	1,643 <u>878,460</u>	1,291,78

CSM (9,655 SPS Tanked) 32,758 SLA 3,842 36,600

TABLE 13

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT S-IB/S-IVB SEPARATION
COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

DISPERSION GROUP		FLIGHT TIME (SEC)	ALTITUDE (M)	SPACE-FIXED VELOCITY (M/S)	SPACE-FIXED FLIGHT PATH ANGLE (DEG)	GROUND RANGE (KM)	SPACE-FIXED FLIGHT AZIMUTH (DEG)	VEHICLE MASS (KG)
S-IB Propulsion	+RSS	3.33	1574.	18.53	1.333	3.19	0.034	1095.
S-IB Propulsion	-RSS	2.68	1406.	19.53	1,212	2.38	0.021	929.
S-IB Non Propulsion	+RSS	0.00	1386	34.48	1.476	3.70	1.800	. 141.
S-IB Non Propulsion	-RSS	0.00	1384.	25.54	1.464	2.51	1.798	141.
S-IVB Propulsion	+RSS	0.00	426.	. 10.34	0.093	0.25	0.012	1046.
S-IVB Propulsion	-RSS	0.00	423.	10.27	0.093	0.25	0.012	1046.
S-IVB Non Propulsion	+RSS	0.00	37.	0.89	0.008	0.02	0.001	91.
S-IVB Non Propulsion	-RSS	0.00	37.	0.89	0.008	0.02	0.001	91.
Combined Positive	RSS	3.33	2140.	40.50	1.991	4.89	1.800	1524.
Combined Negative	RSS	2.68	2018	33.76	1.903	3.47	1.798	1409.

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TABLE 13 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT S-IB/S-IVB SEPARATION

COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

		FLIGHT	VE	HICLE ATTITUI	Œ	(1) VEHICLE ATTITUDE RATE		
DISPERSION GROUP		TIME (SEC)	PITCH (DEG)	YAW (DEG)	ROLL (DEG)	PITCH (DEG/S)	YAW (DEG/S)	ROLL (DEG/S)
S-IB Propulsion	+RSS	3.33	0.065	0.006	0.007	AND wall date	490 to 0.000	prin dell'obli
S-IB Propulsion	-RSS	2.68	0.042	0.006	0.005	WE #W ****	<b>₩</b>	** ** ***
S-IB Non Propulsion	+RSS	0.00	1.020	1.014	1.829	AND AND DOWN	and yes side	***
S-TB Non Propulsion	-RSS	0;00	1.013	1.014	1.829	tive step total	east days who	mole alone should
S-IVB Propulsion '	+RSS	0.00	0.003	0.000	0.000	the state state	.00 and 144	nips while spile
S-IVB Propulsion	-RSS	0.00	0.002	0.000	0.000	No. of the	with day link	
S-IVB Non Propulsion	+RSS	0.00	0.000	0.000	0.000		*	*** ***
S-IVB Non Propulsion	-RSS	0.00	0.000	0.000	0.000	NAME AND ADDRESS OF THE PERSON	pro 944 444	till san mr
Combined Positive	RSŞ	. 3.33	1.022	1.014	1.829	game state state  4	- 1865 - 1880 - 1865 Sterebrich is a merichigan dien Sterebrich Stere provinsi	that and the state of the state
Combined Negative	RSS	2.68	1.014	1,014	1.829	has the side	nderstadgegergeren frequende gjithe de derste berker. Her fan stade freder	and the ship

<sup>(1)</sup> The S-IB stage attitude rates have been omitted in order that more realistic values, reflecting conditions at physical separation, can be determined.

TABLE 13 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT S-IB/S-IVB SEPARATION

COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

		FLIGHT	SPACE	FIXED POSITION	VECTOR	SPACE FIX	ED VELOCITY	VECTOR
DISPERSION GROUP		TIME (SEC)	(M)	(M)	Z (M)	. XDOT (M/S)	YDOT (M/S)	ZDOT (M/S)
S-IB Propulsion	+RSS	3.33	1613.	410.	4494.	47.03	0.54	28.75
S-IB Propulsion	-RSS	2.68	1468.	318.	3421.	51.47	0.18	27.13
S-IB Non Propulsion	+RSS	0.00	1418.	3003.	3721.	46.27	66.17	51.85
S-IB Non Propulsior	-RSS	0.00	1419.	3086.	2519.	47 <b>.</b> 46	65.97	46.30
S-IVB Propulsion	+RSS	0.00	421.	1.	260.	7.79	0.02	7.76
S-IVB Propulsion	-RSS	0.00	418.	- 1.	259.	7.72	0.02	7,72
S-IVB Non Propulsion	+RSS	0.00	36.	0.	23.	0.67	0.00	.0.67
S-IVB Non Propulsion	-RSS	0.00	36%	0.	23.	0.67	0.00	-0.67
Combined Positive	RSS	3.33	2189.	3031.	840.	66.44	66.17	59.79
Combined Negative	RSS	2.68	2084.	3102.	249.	70.44	65.97	54.22

TABLE 13 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT S-IB/S-IVB SEPARATION

COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

	•	FLIGHT	EAR	TH FIXED POSI	TION .	EART	H FIXED VELO	CITY
DISPERSION GROUP		TIME (SEC)	X (M)	Y (M)	Z (M)	XDOT (M/S)	YDOT (M/S)	ZDOT (M/S)
S-IB Propulsion	+RSS	3.33	1586	24.	3209.	46.41	0.71	28.93
S-IB Propulsion	-RSS	2.68	· 1427.	8.	2394.	50.70	0.24	27.28
S-IB Non Propulsion	+RSS	-0.00	1402.	3003.	3727	45.77	66.17	<b>52.2</b> 3
S-IB Non Propulsion	-RSS	0.00	1402.	3086.	2528	46.98	65.97	46.74
S-IVB Propulsion	+RSS	0.00	423.	0.	257	7.87	0.00	7.67
S-IVB Propulsion	-RSS	.0.00	420.	0.	256	7.81	0.00	7.62
S-IVB Non Propulsion	+RSS	0.00	37.	0.	22.	0.68	0.00	0.66
SaIVB Non Propulsion	-RSS	0.00	37.	0.	22.	0.68	0.00	0.66
Combined Positive	RSS	3.33	2159.	3003.	4925.	65.66	66.17	60.20
Combined Negative	RSS	2.68 <sup>.</sup>	2044.	3086.	3491.	69.56	65.97	54.66

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TABLE 13 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT S-IB/S-IVB SEPARATION

COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

DISPERSION GROUP		FLIGHT TIME (SEC)	GEODETIC LATITUDE (DEG)	LONGITUDE POS. WEST (DEG)	EARTH FIXED VELOCITY (M/S)	EARTH FIXED PATH ANGLE (DEG)	EARTH FIXED FLIGHT AZIMUTH (DEG)
S-IB Propulsion	+RSS	3.33	0.009	0.023	19.20	1.562	0.042
S-IB Propulsion	-RSS	2.68	0.006	0.031	20.25	1.414	0.022
S-IB Non Propulsion	+RSS	0.00	0.028	0.026	3i <b>.</b> 15	1.786	2.225
S-IB Non Propulsion	-RSS	0.00	0.027	0.037	21.41	1.763	2.217
S-IVB Propulsion	+RSS	0.00	J.001	0.002	10.63	0.082	0.001
S-IVB Propulsion	-RSS	0.00	0.001	0.002	10.55	0.082	0.001
S-IVB Non Propulsion	+RSS	0.00	0.000	0.000	0.92	0.007	0.000
S-IVB Non Propulsion	-RSS	0.00	0.000	0.000	0.92	0.007	0.000
Combined Negative	RSS	3.33	0.029	0.035	38.12	2.374	2.225
Combined Negative	RSS	2.68	0.028	0.048	31.31	2.261	2.217

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TABLE 14

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT J-2 ENGINE CUTOFF SIGNAL COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

DISPERSION GROUP		FLIGHT TIME (SEC)	ALTITUDE (M)	SPACE-FIXED VELOCITY (M/S)	SPACE-FIXED FLIGHT PATH ANGLE (DEG)	GROUND RANGE (KM)	SPACE-FIXED FLIGHT AZIMUTH (DEG)	VEHICLE MASS (KG)
S-IB Fropulsion	+RSS	4.43	14.	0.00	0.002	7.66	<b>0.0</b> 50	276.
S-IB Propulsion ·	-RSS	3.63	0.	0.00	0.001	6.68	0.042	314.
S-IB Non Propulsion	+RSS	1.07	16.	0.00	0.002	12.18	0.065	300.
S-IB Non Propulsion	-RSS	1.56	6.	0.00	0.002	10.28	0.055	207.
S-IVB Propulsion	+RSS	21.52	16.	0.00	0.002	76.91	0.468	408.
S-IVB Propulsion	-RSS	19.33	.14.	0.00	0.000	70.00	0.424	436.
S-IVB Non Propulsion	+RSS	0.47	8.	0,00	0.002	2.59	0.015	8.
5-IVB Non Propulsion	-RSS	0.45	. 6.	0.00	0.002	2.63	0.015	22.
IMU Dispersions	+RSS	0.03	601.	1.60	0.026	0.32	salet aan 1886 St	and week while
IMU Dispersions	-RSS	0.03	601.	1.60	0.026	0.32	det mil 340	ga sirya - h
Combined Positive	RSS	22. 00	602.	1.60	0.026	78, 29	0.475	5 77.
Combined Negative	RSS	19.73	601.	1.60	0.026	7.1.11	0.430	576.

TABLE 14 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT J-2 ENGINE CUTOFF SIGNAL COMBINED S-IB AND S-IVB STAGE THREE SIGNA DEVIATIONS

	FLIGHT		VEHICLE ATTITUDE		VEHICLE ATTITUDE RATE		
	TIME (SEC)	PITCH · (DEG)	YAW (DEG)	*ROLL (DEG)	PITCH (DEG/S)	YAW (DEG/S)	*ROLL (DEG/S)
+RSŞ	4.43	1.187	0.035	glaf sittle way	0.002	0.000	AL 107 00
-RSS	3.63	1,608	0:041	and day state	0.000	0.000	कर को का
+RSS	1.07	1.17,3	1.601	<del>乘 抓 **</del>	0.003	0.002	***
-RSS	. 1.56	1.469	1.652	هم چند خدن	0.002	0.002	***
+RSS	21.52	0.123	0.022	— =>> ex	0.000	0.000	, M. M. T.
-RSS	19.33	0.466	0.02.7	Judy Soft Mass	0.001	0.001	
+RSS	0.47	0.548	0.540	400 MQ 160	0.002	0.002	pp pa qu
-RSS	0.45	0.598	0.540	our set no	0.002	0.002	***
+RSS	0.03	As se no	ar ar va	NW OW NO	agas tan 199	**** ME COL	right that both
-RSS	0.03	<del></del>	gad mad diet	and pap day	que ses seà	# # <del>*</del>	** = =
`RSS	722.00	1.761	1.690	-rian- +ri-	0.004	0.003	* ***
RSS	19.73	2.306	1.739	ent eur en	0.003	0.003	1. 
	-RSS +RSS -RSS +RSS -RSS +RSS -RSS -RSS	TIME (SEC)  +RSS 4.43  -RSS 3.63  +RSS 1.07  -RSS 1.56  +RSS 21.52  -RSS 19.33  +RSS 0.47  -RSS 0.45  +RSS 0.03  -RSS 0.03	TIME (SEC) (DEG)  +RSS 4.43 1.187  -RSS 3.63 1.608  +RSS 1.07 1.173  -RSS 1.56 1.469  +RSS 21.52 0.123  -RSS 19.33 0.466  +RSS 0.47 0.548  -RSS 0.45 0.598  +RSS 0.03  -RSS 0.03  -RSS 0.03	TIME (SEC) (DEG) (DEG)  +RSS 4.43 1.187 0.035  -RSS 3.63 1.608 0.041  +RSS 1.07 1.173 1.601  -RSS 1.56 1.469 1.652  +RSS 21.52 0.123 0.022  -RSS 19.33 0.466 0.027  +RSS 0.47 0.548 0.540  -RSS 0.45 0.598 0.540  +RSS 0.03  -RSS 0.03  -RSS 0.03 1.761 1.690	TIME (SEC) (DEG) (DEG) (DEG)  +RSS 4.43 1.187 0.035RSS 3.63 1.608 0.041RSS 1.07 1.173 1.601RSS 1.56 1.469 1.652RSS 21.52 0.123 0.022RSS 19.33 0.466 0.027RSS 0.47 0.548 0.540RSS 0.45 0.598 0.540RSS 0.03RSS 0.03RSS 0.03RSS 0.03	TIME (SEC) (DEG) (DEG) (DEG) (DEG) (DEG/S)  +RSS 4.43 1.187 0.035 0.002  -RSS 3.63 1.608 0.041 0.000  +RSS 1.07 1.173 1.601 0.003  -RSS 1.56 1.469 1.652 0.002  +RSS 21.52 0.123 0.022 0.000  -RSS 19.33 0.466 0.027 0.001  +RSS 0.47 0.548 0.540 0.002  -RSS 0.45 0.598 0.540 0.002  +RSS 0.03  -RSS 0.03 1.761 1.690 0.004	TIME (SEC) (DEG) (DEG) (DEG) (DEG) (DEG/S) (DEG/S)  +RSS 4.43 1.187 0.035 0.002 0.000  -RSS 3.63 1.608 0.041 0.000 0.000  +RSS 1.07 1.173 1.601 0.003 0.002  -RSS 1.56 1.469 1.652 0.002 0.002  +RSS 21.52 0.123 0.022 0.000 0.000  -RSS 19.33 0.466 0.027 0.001 0.001  +RSS 0.47 0.548 0.540 0.002 0.002  -RSS 0.45 0.598 0.540 0.002 0.002  +RSS 0.03  -RSS 0.03  -RSS 0.03  -RSS 0.03 1.761 1.690 0.004 0.003

<sup>\*</sup> Not Applicable

TABLE 14 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

FRAJECTORY DISPERSION ENVELOPE AT J-2 ENGINE CUTOFF SIGNAL COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

		FLIGHT	SPACE F	IXED POSITION	VECTOR	SPACE 1	FIXED VELOCITY	VECTOR
DISPERSION	•	TIME	X	Y	Z	XDOT	YDOT	DOT
<del></del>		(SEC)	(M)	(M)	(M)	(M/S)	. (M/s)	(M/S)
S-IB Propulsion	.+RSS	4.43	2429.	540.	8993.	8.93	0.20	2.82
S-IB Propulsion	-RSS	3.63	2840.	394.	7658.	10.78	0.24	3.40
S-IB Non Propuls	+RSS	1.07	3148.	656.	.1696.	11.70	0.35	3.68
S-IB Non Propulsion	-RSS	1.56	3698.	558.	9955.	13.84	0 <b>.3</b> 8	4.37
S-IVB Propulsion	+RSS	21.52	25019.	4683	3622.	90.08	2.02	29.43
S-IVB Propulsion	-RSS	19.33	28490.	4279	6313.	98.86	2.29	33,55
S-IVB Non Propulsion	+RSS	0.47	852.	151.	2676.	3.17	0.25	1.00
S-IVB Non Propulsion	-RŚS	0.45	842.	151.	2696.	3.16	0.24	1.00
IMU Dispersions	+RSS	0.03	654.	725.	207.	4.10	3.05	0.65
IMU Dispersions	-RSS	0.03	654.	725.	207.	4.10	3.05	0.65
Combined Positive	RSS	22.00	25356.	4817•	84956.	91,42	3.68	29.82
Combined Negative	RSS	19.73	28889.	4396,	77387.	100,54	3.85	34.02

\*

\*

TABLE 14 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT J-2 ENGINE CUTOFF SIGNAL COMBINED S-IB AND S-IVB STAGE THREE SIGMA DEVIATIONS

		FLIGHT	EAR	TH FIXED POS	ITION	E	ARTH FIXED	VELOCITY
DISPERSION GROUP		TIME	X	Y	Z	XDOT	YDOT	ZDOT
	·	(SEC)	(M)	(M)	(M)	(M/S)	(M/S)	(M/S)
S-IB Propulsion	+RSS	4.43	1846.	498.	7635.	7.36	1.43	2.10
S-IB Propulsion	-RSS	3.63	2102.	452.	6655.	8.70	1.17	2.50
S-IB Non Propulsion	÷RSS	1.07	2827.	998.	12133.	11.41	0.41	3.15
S-IB Non Propulsion	-RSS	1.56	3350.	<b>834.</b>	10235.	. 13.62	0.41	. 3.77
S-IVB Propulsion	+RSS	21.52	19943.	.5567.	76163.	85.,38	7.86	22. 88
S-IVB Propulsion	-RSS	1933	:22703.	4983.	69552	77.,80	7.14	:2608
S-IVB Non Propulsion	+RSS	0.47	723.	196.	2578.	2.92	0.30	0.82
S-IVB Non Propulsion	-RSS	0.45	708.	201.	2621.	2.89	Ö.28	0.81
IMU Dispersions	+RSS	0.03			w) do (m	44 to so	* ****	<del>~</del> ≈ ~
IMU Dispersions	-RSS	0.03		***	240 AND PA		N	*==
Combined Positive	RSS	2200	,2024,0.	5681.	<sup>.</sup> 77543,	86,50	8,.01	:2321
Combined Negative	RSS	19.73	£23056.	5076.	7.0664.	79.51	7., 25	26.,48

<sup>\*</sup> Does not include effects of IMU dispersions.

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TABLE 14 (cont'd)

AS-205 L/V TRAJECTORY DISPERSION ANALYSIS

TRAJECTORY DISPERSION ENVELOPE AT J-2 ENGINE CUTOFF SIGNAL COMBINED S-IB AND S-IVE STAGE THREE SIGMA DEVIATIONS

	DISPERSION GROUP		FLIGHT TIME (SEC)	GEODETIC LATITUDE (DEG)	LONGITUDE POS, WEST (DEG)	EARTH FIXED VELOCITY (M/S)	EARTH FIXED PATH ANGLE (DEG)	EARTH FIXED FLIGHT AZIMUTH (DEG)
	S-IB Propulsion	+RSS	4.43	0.007	0.070	0.00	0.002	0.052
	S-IB Propulsion	-RSS	3.63	0.006	.080	0.00	0.001	0.045
	S-IB Non Propulsion	+RSS	1.07	0.009	.108	0.00	0.002	0.068
	S-IB Non Propulsion	-RSS	1.56	0.008	.128	0.00	0.002	0.058
	S-IVB Propulsion	+RSS	21.52	0.053	.734	0.02	0.002	0.493
	S-IVB Propulsion	-RSS	19.33	0.054	.808	0.01	0.000	0.448
	S-IVB Non Propulsion	+RSS	0.47	0.002	.028	0.00	0.002	0.016
•	S-IVB Non Propulsion	-RSS	0.45	0.002	.027	0.00	0.002	0.016
	IMU Dispersions	+RSS	0.03	del Ary and			~~~	, en en en
	IMU Dispersions	-RSS	0.03		g and and dust			
*	Combined Positive	RSS ·	22.00	0.054	0.746	0.02	0.004	0.501
*	Combined Negative	RSS	19.73 ·	0.055	0.822	0.01	0.003	0.454

<sup>\*</sup> Does not include effects of TMU dispersions.

FIGURE 1
- S-IB STAGE ALTITUDE VS GROUND RANGE

371539 001 001

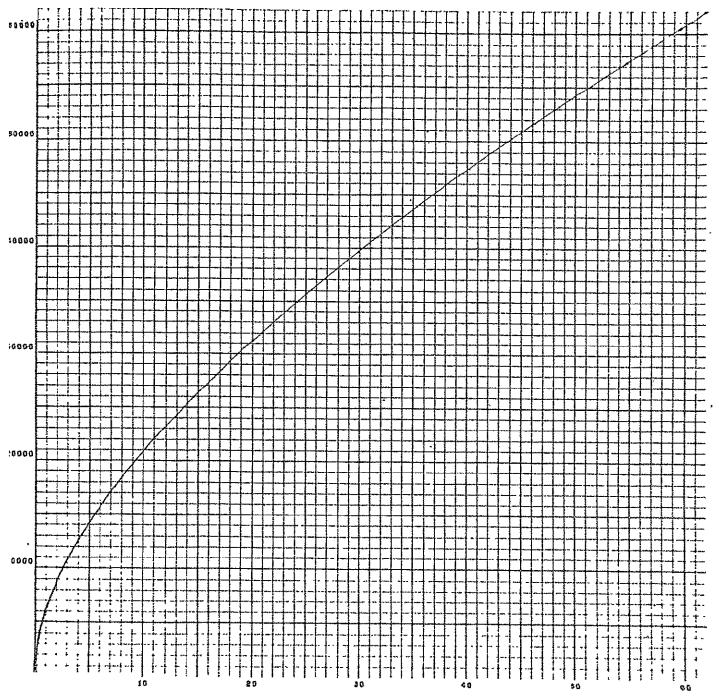


FIGURE 2
S-IB STAGE SPACE FIXED VELOCITY VS TIME

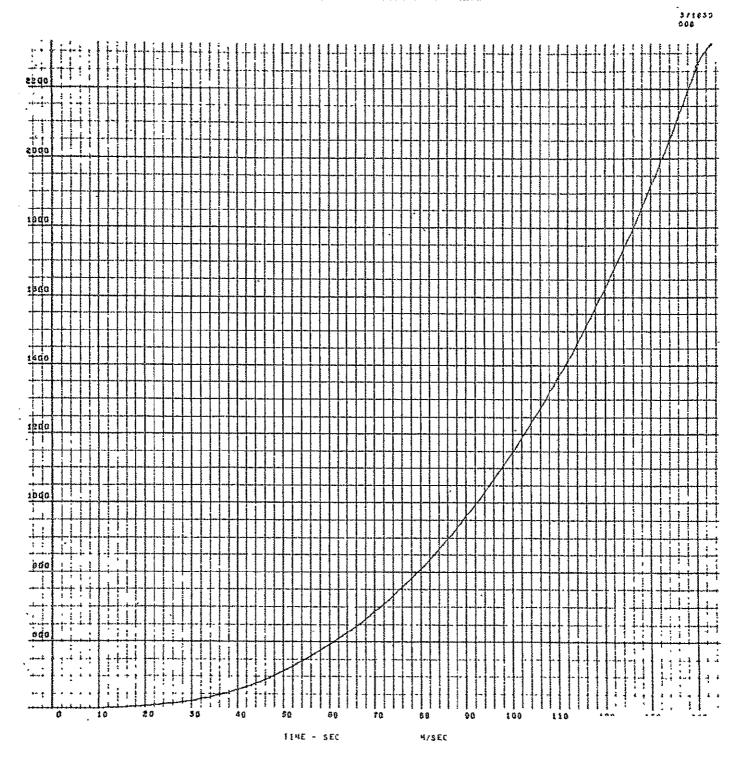


FIGURE 3

### S-IB STAGE SPACE FIXED PATH ANGLE VS TIME

008 668

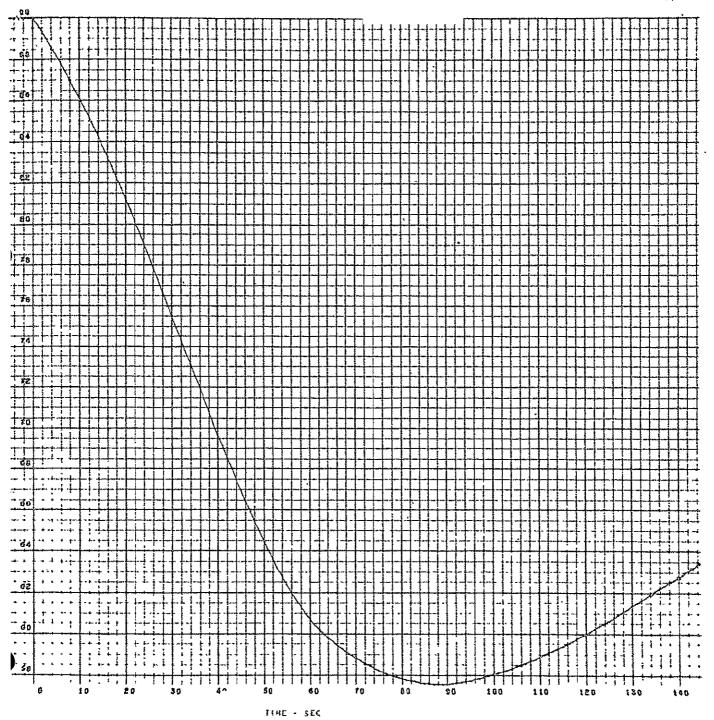


FIGURE 4
S-IB STAGE LONGITUDINAL ACCELERATION VS TIME



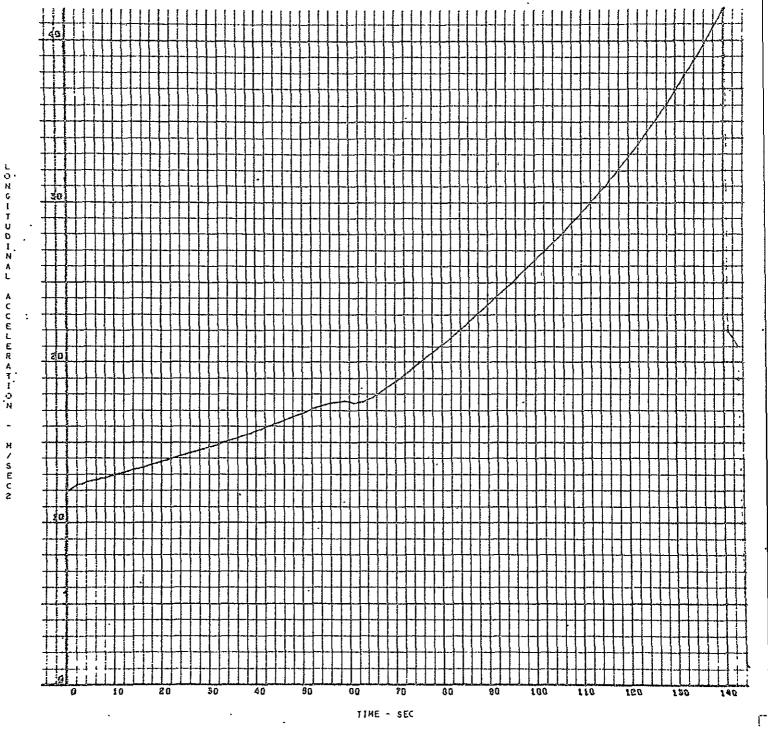


FIGURE 5
S-IB STAGE DYNAMIC PRESSURE VS TIME

5/1830

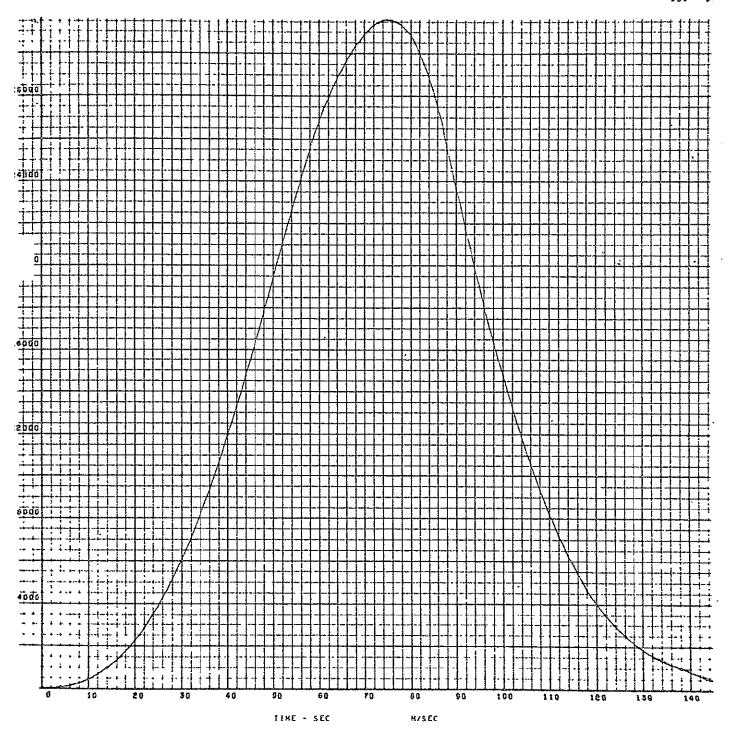
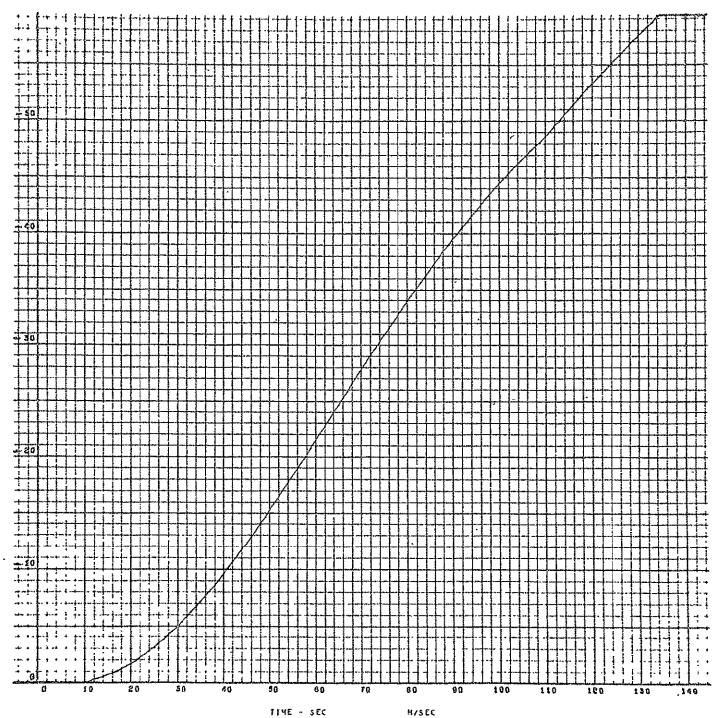
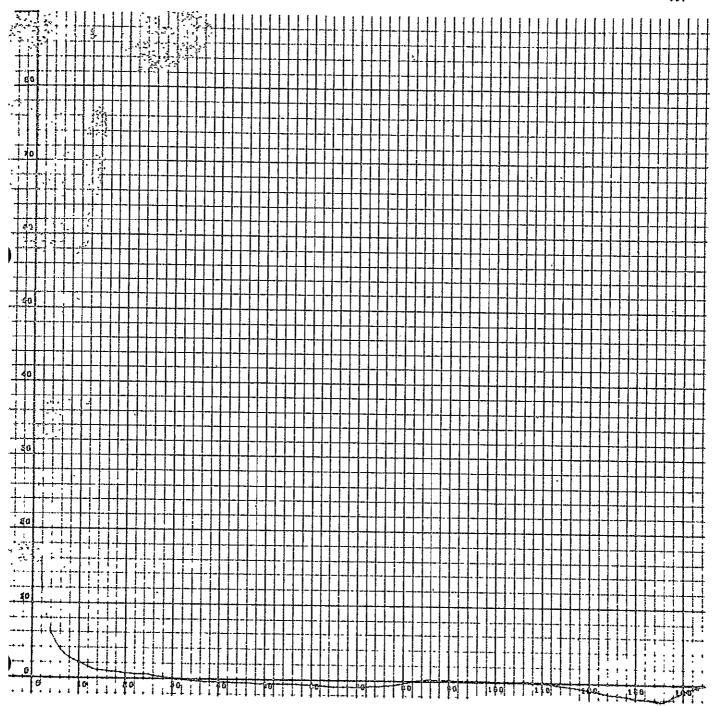


FIGURE 6
S-IB STAGE PITCH ATTITUDE COMMANDS VS TIME



371630



TIME - SEC

H/SEC

FIGURE 8

#### S-IVB STAGE ALTITUDE VS GROUND RANGE

3/1850 GDZ GG.

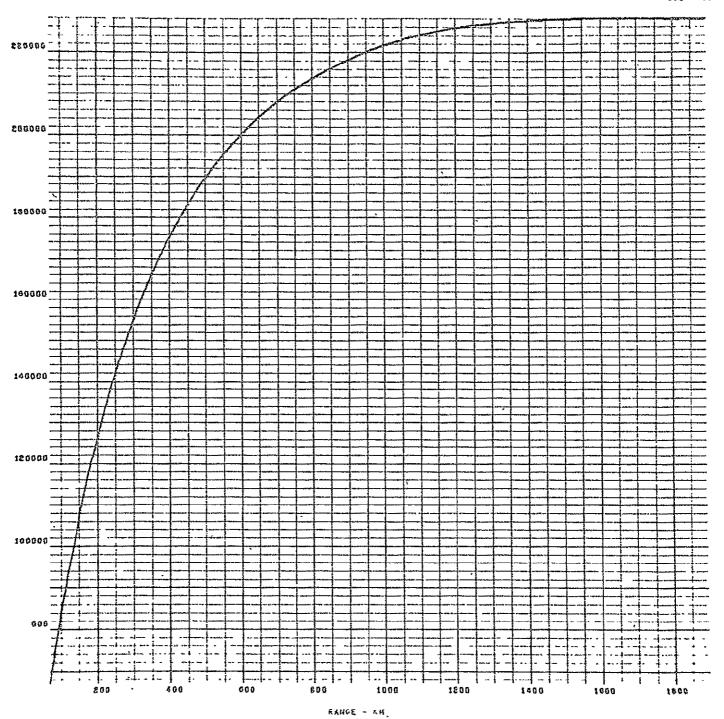


FIGURE 9
S-IVB STAGE SPACE FIXED VELOCITY VS TIME



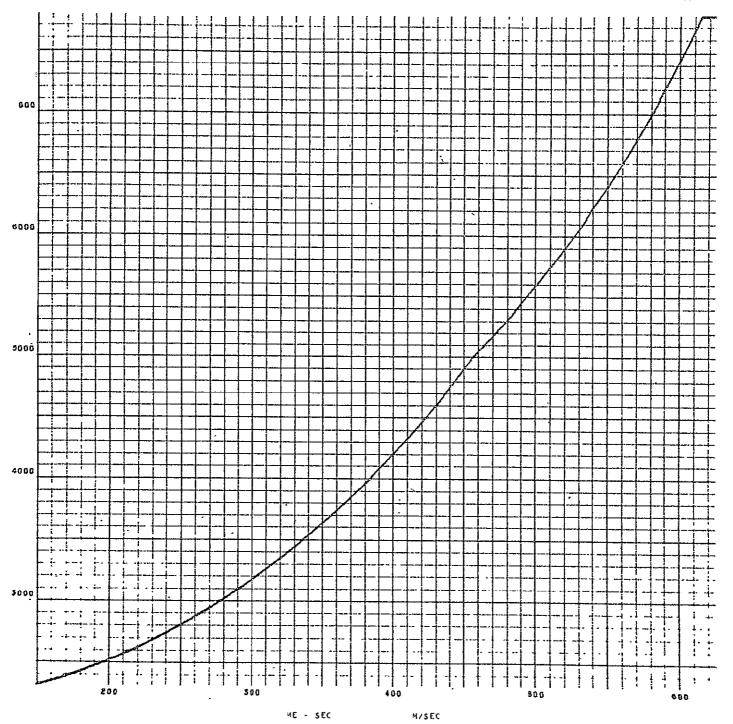


FIGURE 10 S-IVB STAGE SPACE FIXED PATH ANGLE VS TIME

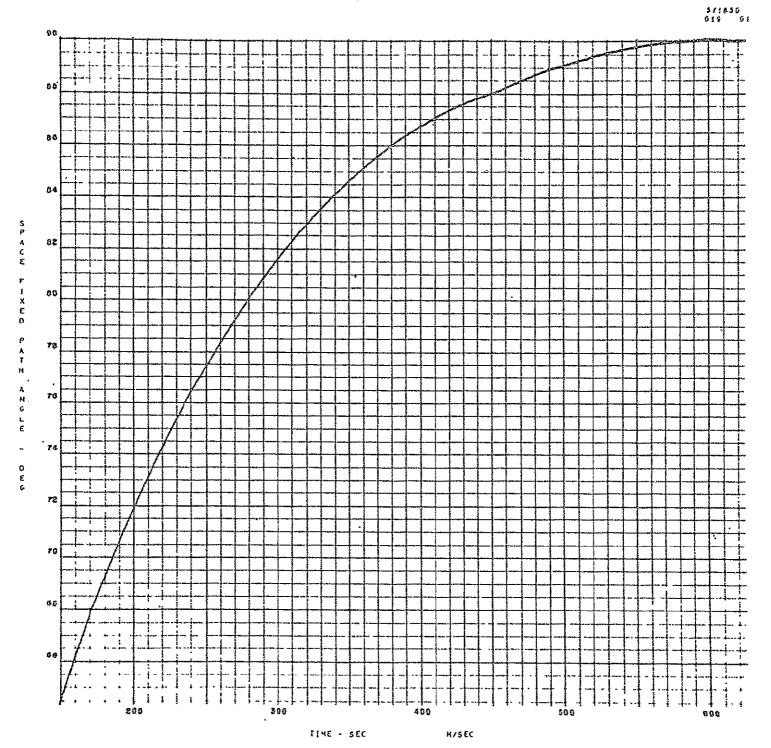


FIGURE 11 S-IVB STAGE LONGITUDINAL ACCELERATION VS TIME

LONGITUDILAL

4 / S E C 2

371830 -004 054

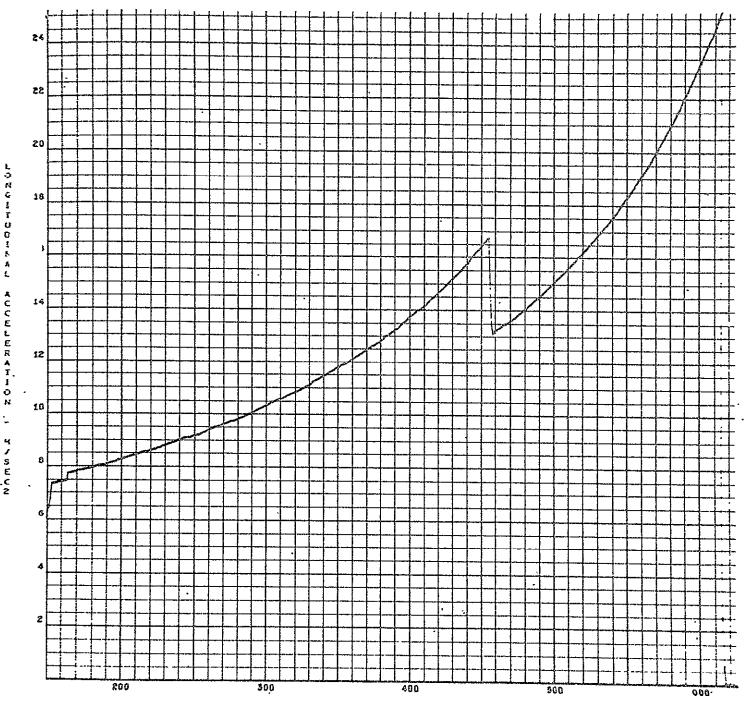


FIGURE 12

## S-IVB STAGE PITCH ATTITUDE COMMAND VS TIME

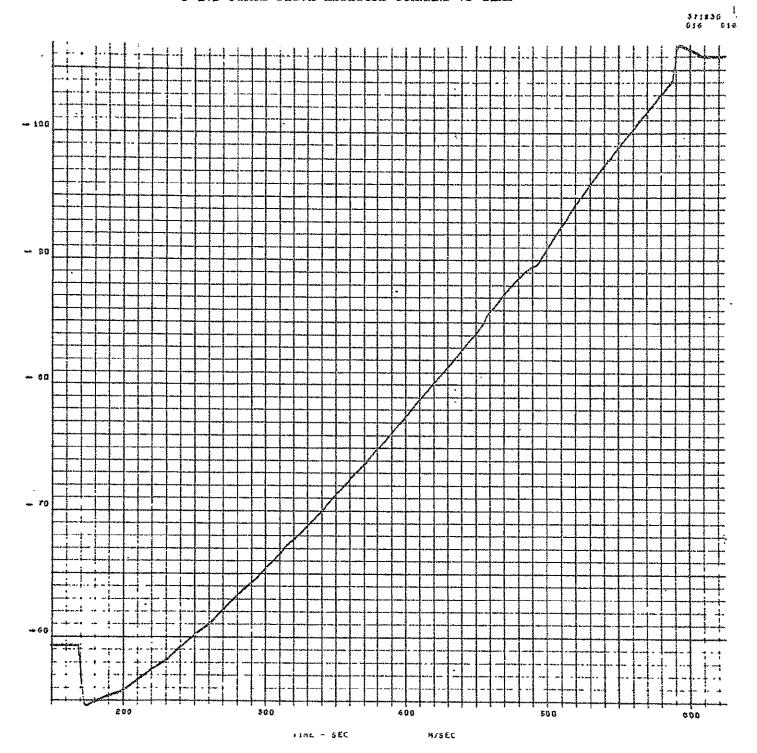


FIGURE 13 S-IVB STAGE YAW ATTITUDE COMMANDS VS TIME

371830 G17 011

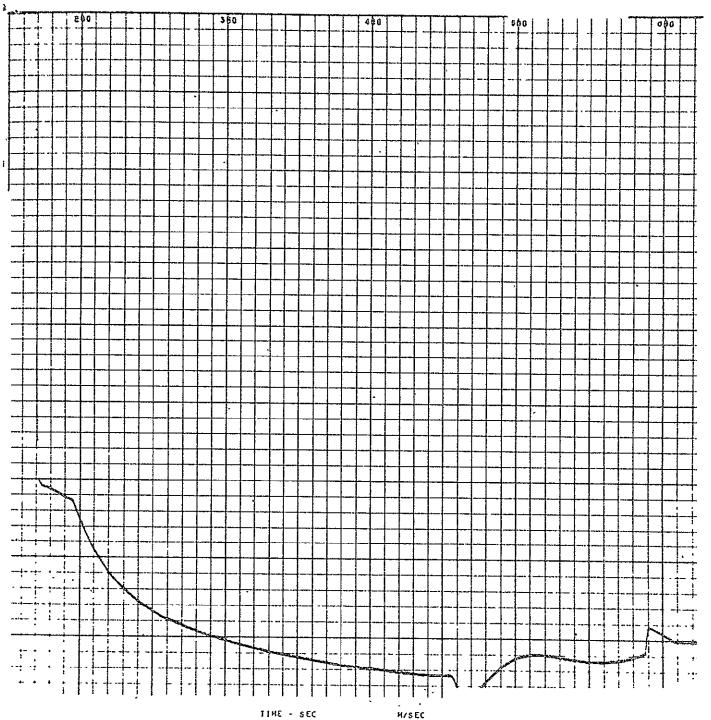


FIGURE 14
S-IVB STAGE-SPACE FIXED PATH ANGLE VS SPACE FIXED VELOCITY

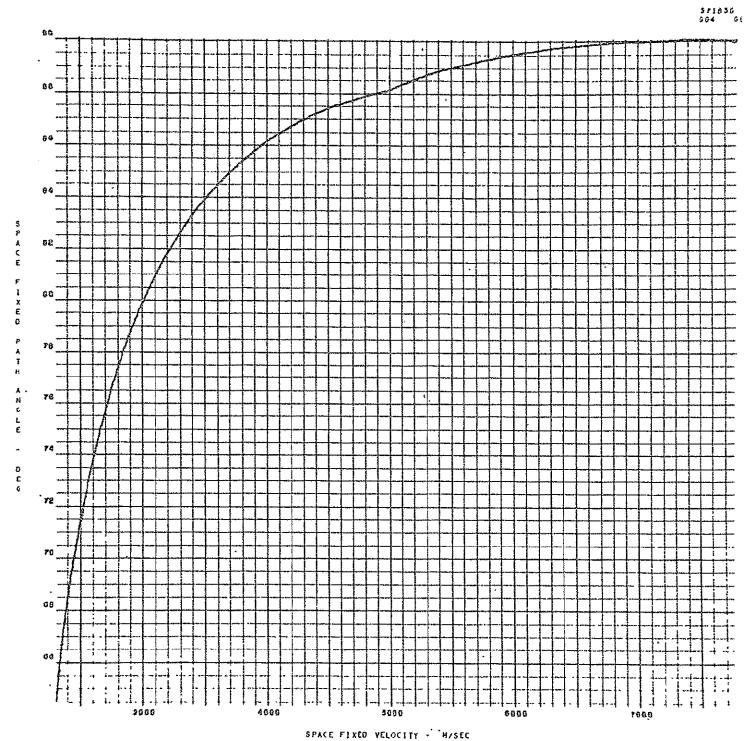
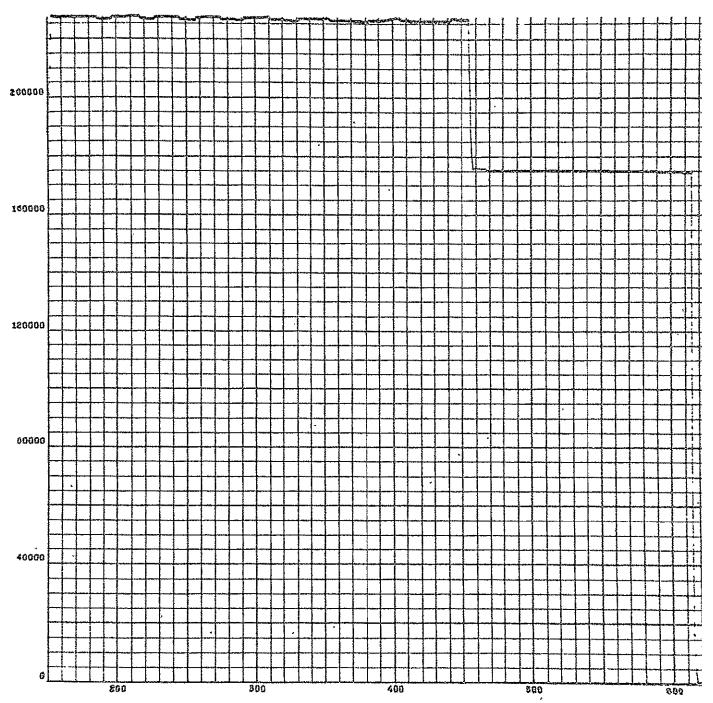


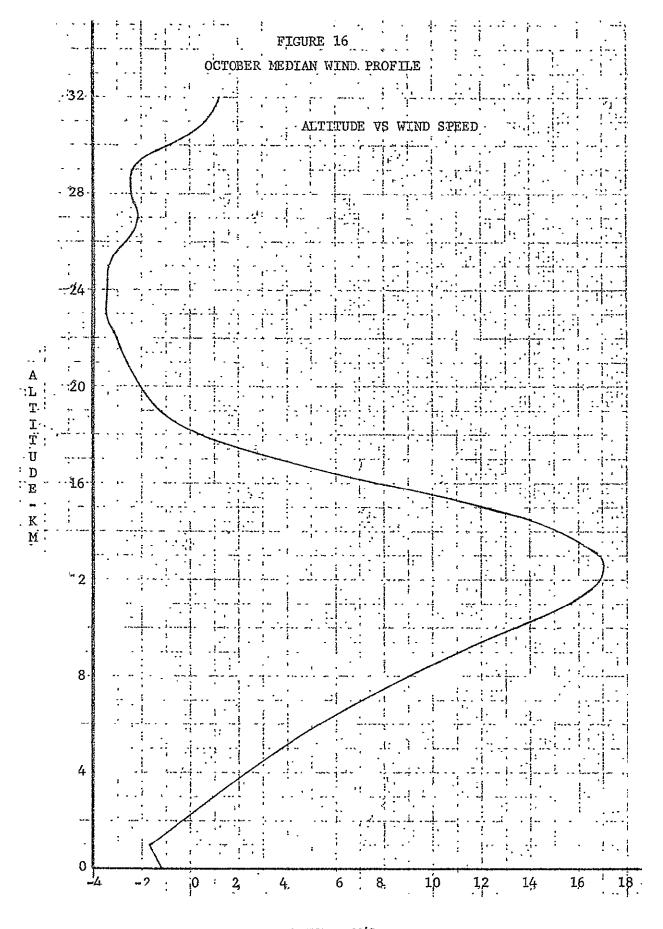
FIGURE 15

### S-IVB STAGE THRUST VS TIME

371650 L



TIME - SEC



WIND SPEED - M/S

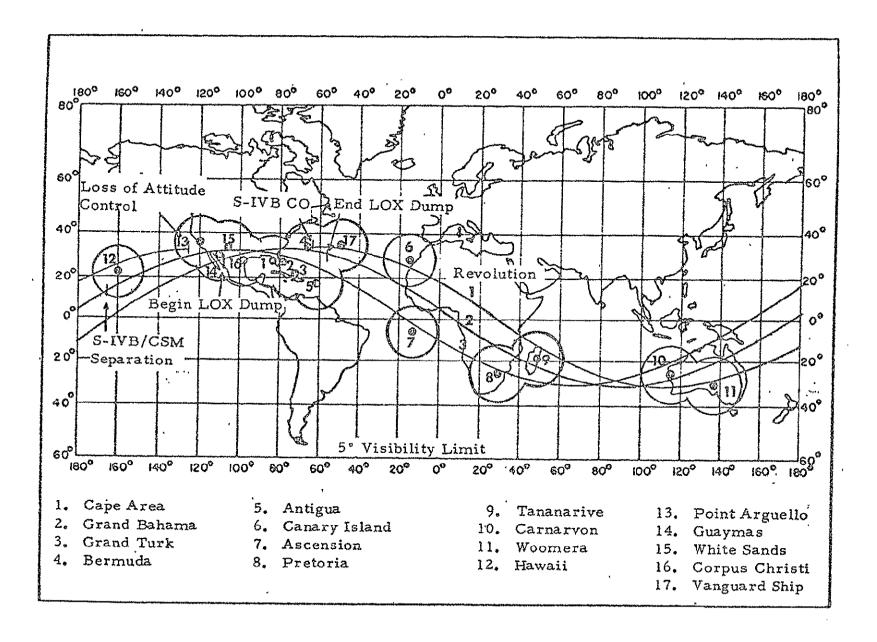


FIGURE 17 SA-205 ORBITAL GROUND PROJECTION ,

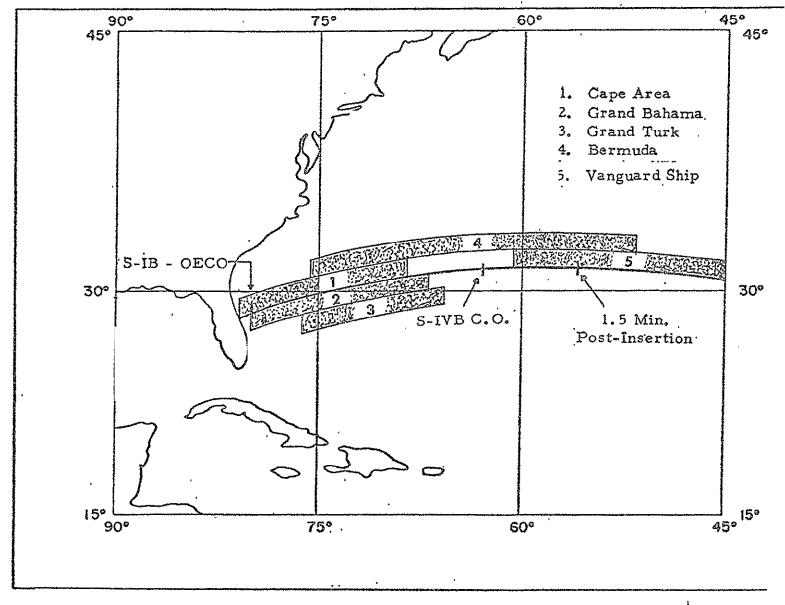
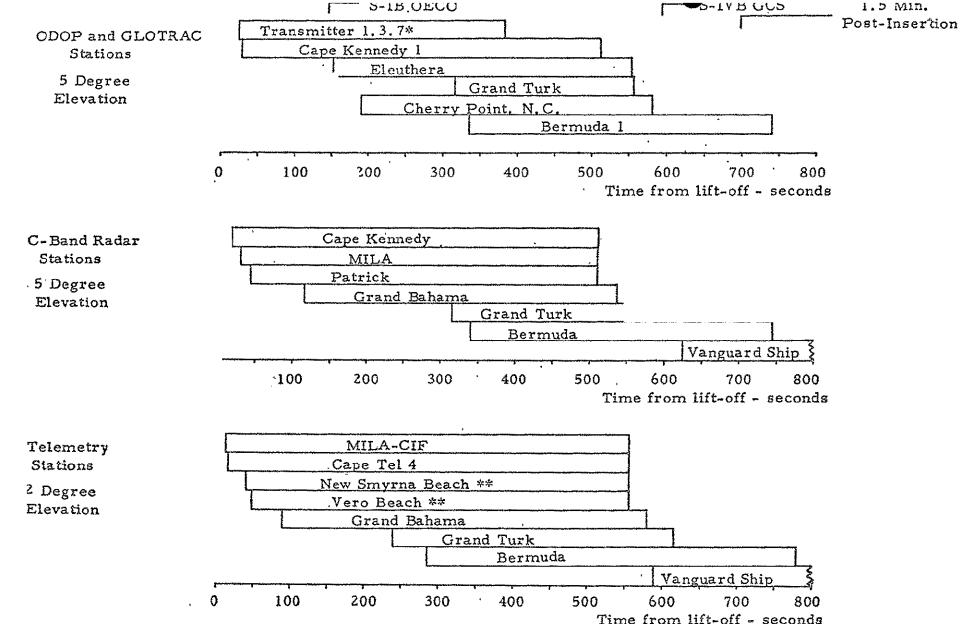


FIGURE 18 LAUNCH PHASE GROUND PROJECTION WITH COMMUNICATION COVERAGE ABOVE 5 DEGREE ELEVATION ANGLE





<sup>\*</sup> Tracks S-IB Stage Only

FIGURE 19 LAUNCH PHASE MISSION TIMELINE WITH TRACKING AND TELEMETRY COVERA

<sup>\*\*</sup> Measures Signal Strength Only

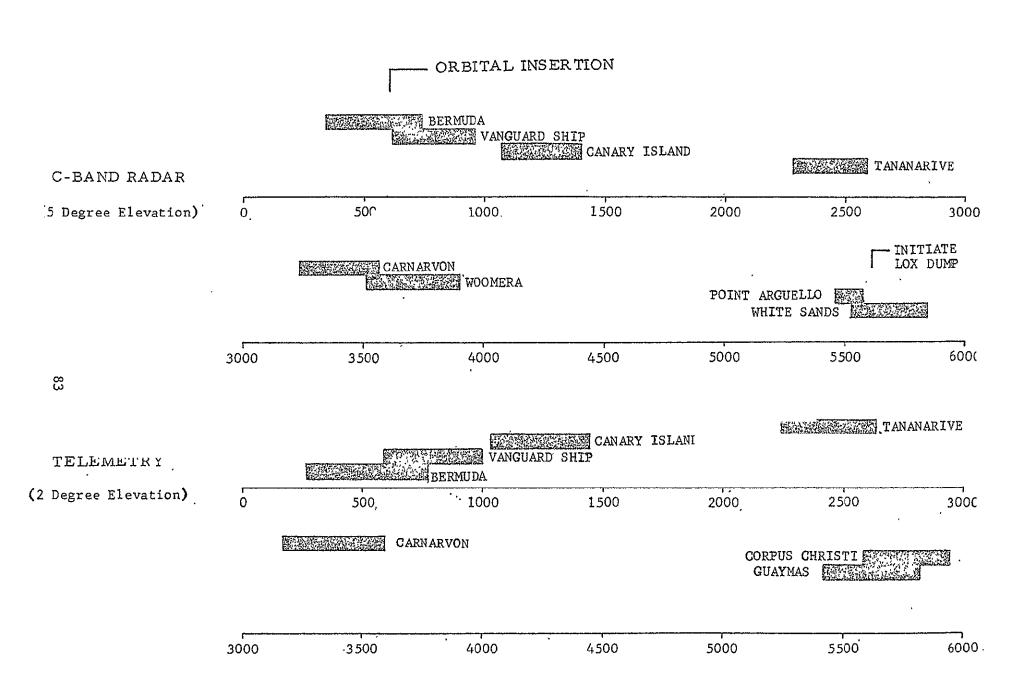


FIGURE :19 (cont'd)
ORBITAL PHASE MISSION TIMELINE WITH C-BAND RADAR AND TELEMETRY COVERAGE

- COMPLETE LOX DUMP MILA GRAND BAHAMA GRAND TURK BERMUDA VANGUARD SHIP PRETORIA C-BAND RADAR TANANARIVE (5 Degree Elevation) 5500 6000 6500 7000 7500-8500 8000 r S-IVB/CSM SEPARATION CARNARVON KANTAN WOOMER HAWAII' POINT ARGUELLO. WHITE' SANDS 8500 9000 9500 10000 10500 11000 11500 CAPE TEL 4 GRAND BAHAMA GRAND TURK BERMUDA ANTIGUA VANGUARD SHIF Example CANARY ISLAND TELEMETRY TANÀNARIVE 8500 7000 5500 6000 6500 7500 8000 (2 Degree Elevation) CARNARVON' EMERICAN TOTAL HAWAII GUAYMAS CORPUS CHRISTI 8500 9000 10000 9500 10500 11000 11500

FIGURE 19 (cont.id)
ORBITAL PHASE MISSION TIMELINE WITH C-BAND RADAR AND TELEMETRY COVERAGE

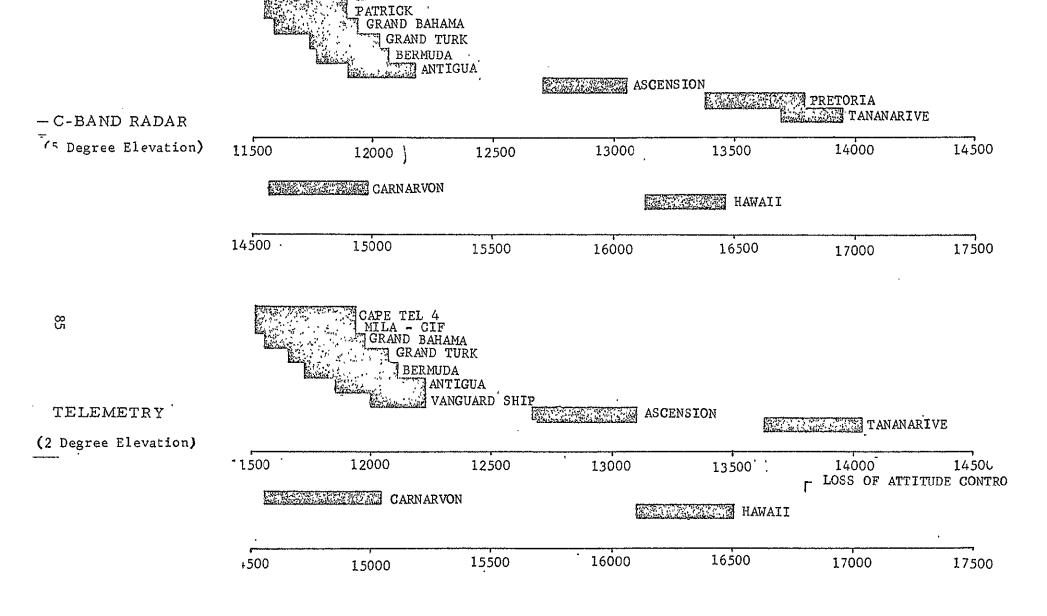


FIGURE 19 (cont'd)
ORBITAL PHASE MISSION TIMELINE WITH C-BAND RADAR AND TELEMETRY COVERAGE.

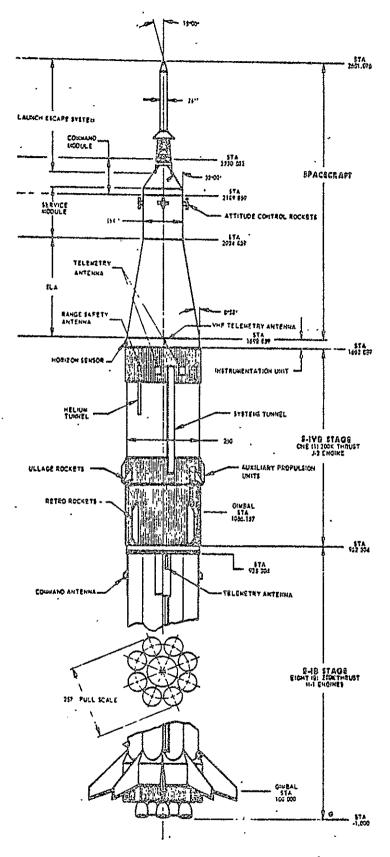


FIGURE 20 AS-205 VEHICLE PROFILE

# APPENDIX A

"ORBITAL TIMELINE AND VENTING SEQUENCE"

TABLE 1A

AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY ORBITAL VENT SEQUENCE & IMPULSE HISTORY

	FLIGHT TIME (HR:MIN:SEC)	TIME FROM TB4 (SEC) ART	FLIGHT TIME (HR:MIN:SEC)	TIME FROM TB4 (SEC)	EFFECTIVE · IMPULSE (LB-SEC)	TANK OR SPHERE AFFECTED AND DESCRIPTION OF EVENT
	0:10:15.2	0.4	0:31:15.2	1260.4	0.	LH <sub>2</sub> Tank (LH <sub>2</sub> Blowdown)
	0:10:15.4	0.6	Open Permane	ently	. 0.	LH2 Tank (Secondary NPV Open)
	0:10:45.0	30.2	0:11:15.0	60.2	9237.	LOX Tank (LOX Blowdown)
	0:54:04.8	2630.0	0:59:04.8	2930.0	0.	LH <sub>2</sub> Tank (LH <sub>2</sub> Blowdown)
89.	1:34:26.8	5052.0	1:46:27.8	5773.0	4700.	LOX Tank (LOX Dump)
∞	1:34:36.8	5062.0	Open Permane	ently	0.	LOX Tank (NPV Valves Open)
	1:34:40.8	5066.0	1:44:43.8	5669.0	0.	LH <sub>2</sub> Tank (LH <sub>2</sub> Blowdown)
	1:42:26.8	5532.0	2:30:14.8	8400.0	0.	Helium Sphere (Partial Helium Dump Through LOX Tank NPV)
	3:17:31.8	11237.0	4:41:22 <b>.1</b>	16267.3	0.	Stage Control Sphere (Helium Dump)
	4:30:14.8	15600.0	4:50:18.4	16800-0	. 0.	Helium Sphere (Complete Helium Dump Through LOX Tank NPV)

<sup>\*</sup> Based on nominal residuals. The  $\pm$  3  $\sigma$  impulse values are 24,500 and 80,000 lb-seconds, respectively. Based these impulse values and the associated mass data, the approximate  $\Delta V^{\dagger}s$  imparted to the configuration are 8.1, and 11.8 meters/second.

TABLE 2A
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL TRAJECTORY
ORBITAL ATTITUDE TIMELINE

	MANEUVER	TI	ME*
	PERINEOVEK	SECONDS	HR:MIN:SEC
1.	Maintain cutoff inertial attitude for 20 seconds after initiation of time base four (tb4).	tb4 + 0:0	0:10:14.8
2.	Initiate maneuver to align the S-IVB/CSM along the local horizontal (CSM forward, position I down) and maintain with respect to local reference.	tb4 + 20.0	0:10:34.8
3.	Begin orbital safing sequence by enabling J-2 engine dump. Maintain vehicle attitude as defined in paragraph above.	tb4 + 4860	1:31:14.8
4.	Begin manual control of S-IVB attitude from the spacecraft. Maneuvers in roll, pitch and yaw will be based on maximum commandable rates of 0.3°/second in pitch and yaw, and 0.5°/second in roll.	9000	2:30:00
5.	End manual control of S-TVB attitude from the space-craft. The I. U. will return to its programmed timeline whenever the spacecraft relinquishes attitude control.	9180	2:33:00
6.	Initiate maneuver to pitch nose down 20° from the local horizontal (position I down) and maintain orbital rate.	9 <u>7</u> 80	2:43:0
7.	Initiate inertial attitude hold using gimbal angles at the specified initiation time. Maintain inertial attitude.	10275	2:51:15
8.	Nominal CSM physical separation.	10500	2:55:00
9.	Initiate maneuver to align the S-IVB/IU along the local horizontal, tail leading and roll to position I up. Maintain orbital rate.	11820	3:17:00 <sup>:</sup>

<sup>\*</sup> Times other than those noted at time base four (tb4) are ground ellapsed time from launch vehicle GRR.

APPENDIX B

"GUIDANCE PRESETTINGS"

#### TABLE 1B

### - AS-205/CSM-101 LAUNCH VEHICLE FLIGHT TRAJECTORY S-IB STAGE UNBIASED PITCH PROGRAM

For the time segment  $T \leq 10$ :

$$\chi_y = 0.$$

$$\chi_z = 0$$

$$\chi_z^{J} = 0.$$
  
 $\chi_x = -.488692 \text{ rad } (-28.0 \text{ degrees})$ 

For the time segment  $10 < T \le 63.8$ :

$$x_y = A_0 + A_1 T + A_2 T^2 + A_3 T^3$$

$$A_0 = -.139871 \times 10^{-1} \text{ rad}$$

 $A_0 = -.139871 \times 10^{-1} \text{ rad}$   $A_1 = +.320258 \times 10^{-2} \text{ rad/sec}$   $A_2 = -.222830 \times 10^{-3} \text{ rad/sec}^2$   $A_3 = +.114730 \times 10^{-5} \text{ rad/sec}^3$ 

$$\chi_{x} = \chi_{z} = 0$$
.

For the time segment  $63.8 < T \le 109.8$ 

$$\chi_y = B_0 + B_1 T + B_2 T^2 + B_3 T^3$$

$$B_0 = +.337135 \times 10^0$$
 rad

B1 = -.108052 x 10<sup>-1</sup> rad/sec B2 = -.384334 x 10<sup>-4</sup> rad/sec<sup>2</sup> B3 = +.350965 x 10<sup>-6</sup> rad/sec<sup>2</sup>

$$\chi_{x} = \chi_{z} = 0$$
.

For the time segment  $109.8 < T \le 134.3$ :

$$\chi_{V} = c_0 + c_1 T + c_2 T^2 + c_3 T^3$$

 $C_0 = +.210868 \times 10^0$  rad  $C_1 = -.978481 \times 10^{-2}$  rad/sec  $C_2 = -.101180 \times 10^{-4}$  rad/sec  $C_3 = +.103404 \times 10^{-6}$  rad/sec

$$\chi_{\rm X} = \chi_{\rm y} = 0$$
.

For the time segment > 134.3:

$$\chi_{\rm V} = -1.0352499$$
 rad

$$\chi_{\rm X} = \chi_{\rm Z} = 0.$$

T = Time from time base one

TABLE 2B

AS-205/CSM-101 LAUNCH VEHICLE FLIGHT TRAJECTOR
S-IB STAGE UNBIASED PITCH ATTITUDE COMMAND

TIME (T) (SEC)	PITCH ATTITUDE OMMAND, X <sub>C</sub> (DEG)	TIME (T) (SEC)	PITCH ATTITUDE COMMAND, Xc (DEG)
0.	0.00	81.8	-35.05
10.	0.00	83.8	
11.8	<b>-</b> .31	85.8	-36.19 -37.31
13.8	53	87.8	-38.40
15.8	83	89.8	-39.47
17.8	-1.21	91.8	-39.47 -40.52
19.8	-1.66	93.8	-41.53
21.8	-2.19	95.8	-42.52
23.8	<b>-2.7</b> 8	97.8	-43.48
25.8	-3.44	99.8	-44.41
27.8	-4.15	101.8	<del>-</del> 45.31
29.8	-4.93	L03.8	<b>-46.18</b>
31.8	-5.76	L05.8	-47.02
33.8	<b>~</b> 6.65	L07.8	<del>-47.82</del>
35.8	<b>~</b> 7 <b>.</b> 58	109.8	<b>-</b> 48.59
37.8	-8.56	111.8	-49.56
39.8	<del>-</del> 9.58	113.8	-50.49
41.8	-10.64	115.8	-51.41
43.8	-11.73	117.8	-52.32
45.8	-12.86	119.8	-53.21
47.8	-14.02	121.8	-54.10
49.8	<b>-15.21</b>	123.8	-54.97
51.8	-16.42	125.8	-55.82
53.8	-17.65	127.8	-56.67
55.8	-18.89	129.8	<b>-57.50</b>
57.8	-20.16	131.8	-58.31
59.8	-21.43	.133.8	-59.12
61.8	-22.71	134.3	-59 <b>.</b> 31
63.8	-23.99		
65.8	-25.23		
67.8	-26.51		
69.8	27.79		
71.8	-29.04	1	Ţ
73.8	-30.28		1
75.8	-31.51		1
77.8	-32.71	₩	¥
79.8	-33.89	IGM Starc	-59.31

T = Time from time base one

TABLE 3B

^C-205/CSM-101 LAUNCH VEHICLE OPERATIONAL FLIGHT TRAJECTORY.
IGM PRESETTINGS

LVDC SYMBOL	INITIAL VALUE	UNITS	DESCRIPTION
منه منه منه منه	25.	sec	Time from time base 3 to initiata IGM guidance.
Tli	286.3	sec	Time to go for first IGM stage.
T3i	159.8	sec	Time to go for second IGM stage.
т <sub>3</sub>	326.5	sec	Average value of $m/\tilde{m}$ at initiation of second IGM stage.
Vex1	4151.0	m/sec	Average exhaust velocity (go x Isp) for 1st IGM stage phase of flight.
·Vex3	4233.2	m/sec	Average exhaust velocity (go x Isp) for 2nd IGM stage phase of flight.
. VT	7780.67	m/sec	Guidance Cutoff Signal (GCS) criteria. Terminal velocity for IGM equations.
TVX	6600405	m T	
YVT	0.	m	Desired terminal position vector components in IGM coordinate system.
ZVT	0.	m _	1 Total Decrease Byolding
XVT	0.	m/sec ]	
YVT	0.	m/sec -	Desired cutoff velocity components in IGM coordinate system.
ZVT	7780,67	m/sec	2. Zori Godzanace System.
ΧVG	- 9.15	m/sec <sup>2</sup>	•
Ϋ́VG	0	m/sec <sup>2</sup>	Terminal gravitation vector components in IGM coordinate system.
zvg	0.	m/sec <sup>2</sup>	zi zon coorarnace system.

TABLE 3B (Cont'd)
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL FLIGHT TRAJECTORY
IGM PRESETTINGS

LVDC YMBOL	INITIAL VALUE	UNITS	DESCRIPTION
	25.	sec	Value of T3i to initiate IGM $\Delta \overline{V}$ guidance.
one and less and said	5.	sec	Value of T3i to freeze IGM $(\epsilon')$ .
TRL	1.8	N/D	Mission dependent constant multiplier for terminal range angle equation.
TR2	0.	N/D	Mission dependent constants for Ni
TR3	0.	N/D	and N4 equations.
جد جد سه شد	56.89	sec	Time from time base 3 to initiate thrust misalignment correction.
ĨŹ	0.	sec	N2 and N4 multipliers for $\chi_y'$ and $\chi_z'$ .
Δ TNOM	R-ASI	sec	Nominal computation cycle length.
TLD	0.	N/D	IGM staging flag (burn switch).
AP1 AP2 AP3 AP4 AP5 AP6 AP7 AP8 AP9	42597028 E+00 .50108032 E-01 .90334849 E+00 39490759 E-02 .99835259 E+00 57239996 E-01 90472851 E+00 27949930 E-01 42507067 E+00	N/D N/D N/D N/D N/D N/D N/D	Transformation matrix from navigation coordinate system to the (X4, Y4, Z4) coordinate system  Az = 72.  ØL = 28.521963.  ØN = 119.0  i = 31.605  Time from time base 3 to sample F/M for IGM staging.
FLAG	N/A	N/D	Number of F/M decreases required for IGM staging criteria.

TABLE 3B (Cont'd)
AS-205/CSM-101 LAUNCH VEHICLE OPERATIONAL FLIGHT TRAJECTORY
IGM PRESETTINGS

LVDC SYMBOL	INITIAL VALUE	UNITS	DESCRIPTION
ACC	N/A	m/sec <sup>2</sup>	F/M comparison tolerance for IGM staging.
T10	286.3	sec	Constants for updating second stage
AO	N/A	sec -	time-to-go for perturbed EMR shift time.
AI	n/a	M/D	LLIE .
PCO	0.	sec	Backup time for IGM staging.
τ10	536.0	sec T	
CO	35.	sec	
MO3	129761.6	kg -	Constants for artificial T3 mod
* MI	239.7877	kg/sec	
* M3	183.1797	kg/sec	
* M03	134374.	kg	Mass at J-2 start command minus mass to be jettisoned.
* MI	585943.	kg	Mass at first motion.
* 11	7600000.	kg 'm/sec <sup>2</sup>	S-IB thrust:
* F2	995355.2	kg m/sec <sup>2</sup>	S-IVB average first burn thrust, from 90% thrust to IGM staging.
* <b>F3</b>	775441.26	kg m/sec	S-IVB average second burn thrust from IGM staging to cutoff.
* M <sub>2</sub>	2820.8	kg/sec	Average S-IB flow rate from first motion to IECO.

<sup>\*</sup> Biased values for accelerometer failure cases from R-ASTR-NG-115-68.

# APPENDIX C

"TRAJECTORY LISTING, ENGLISH UNITS"

	***************************************			TABLE 1C			
			S-IB	POWERED FLIGHT TABL	ES		
	TIME	ALT .	RKR · ,	VTH+	V	AZ o	AH:
	(SEC)	(FT) ,	· (FT)	(DEG)	(FT/S) .	(DEG)	FT/FT2-RAD)
	<del></del>	· 1 1-3 v	<del>20707561.</del>	90+00			····
	0.00	11,3 •	20909961.	90.00	1341.76	90.00	0.
	5 v 0 0 ······	21-3	20910061-	88 •-22	1342-25	89,99	
	10.00	540.	, 20910388.	86.15	1344+26	89.98	1254+
	20.00	1126. 2004.	20910974. 20911852.	83 - 82	1349 • 28	89.97	
	<del>25,00</del>	3-2-1-1-	20711852 · · · · · · · · · · · · · · · · · · ·	81+23	1360 • 10	89.93 89.83	27256 • 76578 •
	30.00	4781.	20714629.	in a contract of	1409.44	89.45	180520•
	35.00	6749.	20711027	72.55	1453 • 41		375176
	40.00	9152.	20718999.	69.65	1513.34	88.92	709167•
	-45+00	1-202-3	20 ° 2 1-8 7 0 <del></del>		1590 • 96		1243714
	50.00 (	15398.	20925244.	64.48	1687.63	87,68	2050507
	-55+00	<u>l-</u> 9-3β6	20929151	62.37	1803-61		32072461
	60.00	23767 ·	20933611.	60 - 71	1936-40	86.07	4781528+
	<del>65</del> +00 <del></del>	2877-2	209386-1-4-	59,60	2081 • 37	85 • 2 <u>1</u>	
	70.00	34313.	20944153.	58.80	2246+45	84.32	9225805+
···	<del>7</del> 5+00	40428	20950264	5 ë v 2 2	2434 • 82	83 • 42	12087703
	80.00	47158+	20956991.	57 • 8 <u>2</u>	2647•34	82.53	15336926+
	<del>05</del> +00	54547*	<del>20964376.</del>	57-59	2884-27	8 t60	
	90.00	62637,	20972460.	57.56	3147.73	80.88	22365033.
•	<del>95</del> • 0 0	7-1-454-	20961-270-	<del></del> 5 7- <b>.</b> 7 2 <del></del>	3438 • 74		25613234 <del>v</del>
	100.00	81026	20990834	58.01	3756+80	79.41	28508397•
	-105-00	91380	21001-181	58•4U	4102+23	78.77	31006685
	1 10 • 00 '	102549.	21012339	50.86	4475 • 85	78.20	33089589.
	120.00	127436.	21024338.	59+37 59+97	4879.49	77.48	34757560+
	-} 25 * CO		21037201.		5314×70 5844.50	778 £4	481692026 
	130.00	155854.	21065586.	61.32	6291.05	76.41	37796354.
	134+50	169835	21079550		6.780 • 82	76 s-1-0	38293432
	135.00	171435.	21081148.	62,04	6837+35	76.07	38353944.
<u> </u>	140.00	187973	21097665		7426-73	75,78	38770758
	140+11	188349.	21098040.	62.72	7440+19	75.77	38775172+
<del>-2)</del>	143.11	198657. 203126.	21112797	63•40	7627•97 7631•53	75.70 75.70	39022199•
-3)	_1-44-a-4 R	203396	21113067	63+42	7630+72	75.70	39027177

1) Inboard Cutoff
2) Outboard Cutoff

3) Separation

TABLE 1C

#### S-IB POWERED FLIGHT TABLES

	TIME	MASS	THRUST.	VTHE	AZ I	HANGE	VVVE
	(SEC)	(LB)	(18)	(DEG)	(DEG)	(FT)	(FT/S
		1291782.73		N/A	N/A	0.	0+00
	0.00	1291782.73	1562203.	N/A	N/A	0•	0.00
	5 <u>.</u> 00	1241107.83	1607473	0 - 40	n/a		41.45
	10.00	1230432+94	1617234.	0.45	N/A	~3·	. 90 • 1
	15 • 00	1-1.99440+14	1632437+	O • 28	N/A	<u>~</u> 7.	145 • 28
	.20.00	1168447.34	1644747	U-79	. 54+85	10.	207+29
	25+00	<u>1137-263-84</u>	1-655890-	2-09	66-75	34	276.66
	30.00	1106120+34	1667434.	3 • 90	69 + 42	115.	353+46
	35+00	1074901+05	1660861	6.15	70+29	287.	438.40
	40.00	1043681.74	1694984.	8.74	70+61	602.	532+36
	45.00	1012429+84	1710295	11-58	7.0 + 8 2	111.7.*	636.33
	56 • OO	981177.94	1726048.	14.59	71.02	1902.	751 • 37
	85.00	945924 69	1741473	1-7-7-0	71+18	3035	877.98
	60.00	918671.44 -	1754691.	20.84	71.30	4596.	1013+71
	65×00	La7314+09-	1768246	24•04 <u></u>	7 <u>1.0 4 0</u>	6662	1153+38
ç X	70.00	655956.76	1782925.	27 + 25	71.50	9322	1308 • 86
<del>x</del>	75.00	824555.04	1796101	3U • 3 3	71.057	1 2 6 8 1	1485 - 79
	80.00	793166-66	1807262	33.25	71.63	16849.	1685 - 71
· <del></del>	<u></u>	761868-81	1816407	35.95	71.69	21941	1909.59
	90.00	730570.95	1822644.	38.54	71+75	28077	2158+83
	95+DQ		1826569	41 • U3	7.1 + 8 U	35399	2434.48
	100.00	668238.37	1828101.	43.36	71.85	44049.	2737.13
		6371-27 26	1827945	45.52	71.89	S4165	3067.55
	110.00	606016.19	1826472	47.51	71.93	65887.	3426.78
	-15-00	575010-75	1-8-2-3-6-1-1	44.36	71-77	79359	3816.26
	120.00	544005.36	1817070	51+12	72.02	94742.	4238 00
	_1-25 - 00	5-1-3 1-6-1 - 6-2	1.014073	52.81	7.2.05	112211	4694.33
	130.00	482317.86	1807687.	54.41	72.09	131948.	5187+81
	134_50	454782,24	1.800289	\$5.7.7	72.13	151813	5666.25
	135.00	451722.75	1799344.	55.92	72.13	154150.	5721.54
	140-00	421127,64	1-78-70-93	57.28	72418	179019	<u> </u>
1)	140.11	420458.86	1786680.	57.31	72.18	179598	
<u>-2</u> \$_		409321-08	786845	57.98	77.22	195703	6494.26
-7	144.41	407542.83	46942.	56.26	72.23	202810.	6494.74
_31	144.49	407516.02	44478	58.28	72.23	203242	6493.74

<sup>1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Separation

S-IB POWERED FLIGHT TABLES								
TIME	x ´	Y	Z	DX	DA.	'nΖ		
(SEC)	. (FT)	· (FT)	(FT)	(FT/S)	(FT/S)	(FT/S		
-5.00	20909874	55930		00.0	414+63	1.57.		
0.00	20909877.	58002.	-11792.	0.00	414.41	1-276+0 1276+1		
5.00	20909974	60074		40.83	41.4.03	127601		
10.00	20910296.	62143.	969.	88 • 87	413.66	1275 • 9		
<del></del>	20710874+	642-1-1-	7349,	143+56	4·13·52	1276.3		
20.00	20911743.	66278.	13737.	205 • 12	413.37	1279 # 4		
25.00	<del></del>	<del>68344</del>		273+89	413,32	1287.0		
30.00	20914493.	70411.	26618.	. 349.62	413.24	1301.3		
35-00	20916446	7 2 4 7-7	33179	432+35		1324.7		
40.00	20918829.	74541.	39884.	522 • 12	412.57	1359 0 1		
45+00	20921678	7 6 6 0 2 6	46793.	618.73	4-12-02	140606		
50.00	20925027.	78661.	53975.	721:82	411+54	1468 • 9		
55+00	20928945.		61509	830+29		1547-4		
60.00	20933333.	82771 .	69477.	940 - 31	410.52	1642.2		
	20938297.	84823.	77956	-1045 • 15	410.01	175206		
70.00	20943792.	86071.	87037.	1154-09	409.48	1883.3		
75-00	20949852	68417-	96826.	1.271-25	408.87	203509		
80.00	20956518.	90960.	. 107435.	1340.66	408.26	2211.5		
#5 × 00	20763832.	<del></del>	1-1-8-9-6-6	1530+33	40.7+65	<del>2410+</del> 5		
90.00	20971832.	95037•	131587.	1670.26	406.99	2636.8		
95+00	20980542	97070.	145398.	1814+64	406.24	2892.5		
100.00	20989987.	`99099• (*` `	160560.	- 1963.97	405.44	3176.7		
105+00	21000190v	101-124	177214.	2116+13	404243	3489.7		
110.00	21011176.	103143.	195506.	2277 • 29	403,27	3832.0		
115-00	21022971		215588	2440+73	401-77	4205+6		
120.00	21035587.	107163.	237621.	2600.06	400.54	4614.5		
1-25 • 00	21049034*	1 0 9-1-4-1	261794.	2773.04	398,41	5060+8		
130.00	21063320.	- 111146.	288295.	2941.88 -	395.94	5546.7		
134.50	210769U4	112423	31-430-7-	3095+64	393,58	6020-1-		
135.00	21078456.	113120.	317331.	3112.82	393.31	6074.9		
140-00	<del>2-  0 9 4 4 6 2</del>	1-15079.	3491-16,	3293+60	390,48			
1) 140-11	21094025.	115122.	349847.	3297.84	. 390.42	6657.99		
2)	21104769.	<del></del>	<del>378143</del>	3321+30	387-42	6855.95		
144.41	21109071.	116798.	379075.	3291 • 53	389.21	6674.20		
3) 144.49	21109331	1-16829v	<del>37</del> 9618,	3289 + 1-9	38%-20	4874×42		

<sup>1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Separation

				TABLE 1C		•	
			S-1	IB POWERED FLIGHT TABL	ES	,	
	TIME	XE	YE	Ζε	DXE	DYE .	DZE
	(SEC)	(FT)	(FT)	(FT)	(F1/S) ·	(FT/S)	. (FT/S
		,				,	
	<del></del>	1-1-4 -	-0.	<del></del>		0.00	
	0.00	114.	· -0.	<b>-0</b> .	⊷n•00	-0.00	0 • 0
	10+00	214.			4169		
		541 • · · · · · · · · · · · · · · · · · ·	+2.	-1.	90•15 145•28	-0.33 	-0.48
	20.00	2005	<b>+5</b>	· -0 •	207 • 28	*D•29	
	25.00	3212	-5• ,	28,	276.49	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	
	30.00	4782.	~7.	109.	352.69	-0.14	23.72
	35+00	6750	-8	280.	435 - 92		46.60
	40.00	9153.	-10.	593.	526.25	+0.55 ·	80.45
	45-00	1-2024			623+49		1-27-+18
	50.00	15399.	-19.	1890.	727 • 30	-1+31	188 - 67
	55.00	19307+	27	3020-	834-61		266031
	60.00 .	23768.	-35.	4580.	94/+61	-1.92	360.03
		287.72			1053+59	-2 - 15	469.30
<b>#</b>	70.00 75.00	3431·2 • 40424	÷57•	9306.	1163+86	-2 - 31	598 • 78
ŏ	80.00	47152.	-81.	16845	1409 • 81		750°04 924°14
		54537	-94.	21949	<u></u>	~2.47	72701- 1121447
	90.00	62619.	<del>-</del> 106.	28105	1687.97	-Z•35	1345.89
•	95-00	71424	-117	35457.	1835+19		
	100.00	80980.	-127.	44148.	1987.79	*1 <b>.</b> 7 7	1881 • 64
	105+00	91-31-1	-1-35	54321	21.45 • 67	-1.42	2192.25
	110.00	102445.	-141.	. 66119.	2309+03	÷0.99 ·	2532.04
	<del></del>			79693	<del>2477+23 /</del>	~0,45	<del>29020,9</del> 4
	120+00	127222.	-145.	95207.	2647.93	0 • 25	3308.95
	125+00	140893		1-1-2644,	2820 • 95	······································	3752 - 1-9
	130.00	155416.	-139.	132795.	2996+57	0.95	4234.85
	135.00	170864.	-134. -133.	152896 155263.	3157+1-2		4705-21
		187202	-133.	122403.	3175 • 10 	1.54	4759.70
	1) 140.11	187573.		181049	2010 01	2+36	5326+11 5338+97
	2) 1 1 3 - 1 1	197705	-123.	197385	3396124	2+38	5535.42
	144.41	202136.	-110+	204600.	3307.62	3.55	5553.45
	3)-44-4-48	203402.		205039	3365.34	3.57	5553.6A

<sup>1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Separation

TABLE 1C

			. s-m	POWERED FLIGHT TABLE	ES		
	TIME	. PHIP	РНІҮ	рнія	OPHIP	үінчо	DPHIR
	(SEC)	(DEG)	· (DEG)	(DEG)	(DEG/S)	(DEG/S)	(DEG/S)
	<del></del>	0.00	<u>8-00</u>	2-3-0-0	-0.00	0.00	
	0.00	-0.02	0.01	27.99	· -0.00	0+00	-0.00
		0 • 0 2	<u>-0.+0</u> 2	28 +0-1			
	10.00	0 + 0.2	~0.02	. 28.00	0.00	0.00,	-0.01
		7D 4-55	U+05	24-68			
	20.00 25.00	-1.40	<b>4.07</b>	19.17	-0+23	. 0.00	. ~0.86
	<del>-</del>	-2-74	U • 0 7 - ·	<del></del>	<del></del>	-0-00-	<del></del> -0-6-8-
	.30.00 35.00	=4 + 4 9 = 4 + 4 9	0.05	9.13	<b>-</b> 0 • 3 9 ′	-0.01	~1+00
	40.00		0:02				
	45+00	~9 • 03 ~1-1 • 68	~u.03 ————————————————————————————————————	-0.19	-0.51	-0.01	-0.06
	50.00	-14.51	~U.U3	, +U,01	-0.55	G • 00	
	55.00-		~0.03		+0.58 59		-0+00
	60.00	-20,39	· -0.04	-0.01	-0.5x	-0.00	
	65.00		~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~		-0.6Z		+0.00 
	70.00	-26.50	*0.05	-0.02	-0.59	-0.00	*0.00
	75-00					0:00	-0.00
20	80.00	-32.59	. =0.05	<b>-0.01</b>	-0.62	0.00	0.00
	85.00	~35,72		0.00			0.00
	. 20.00	-38.84	-0.05	0.00	<b>~</b> 0.59	0.00	-0.00
	95.00	-11.52	-0-04			0.00	
	100.00	<b>-43</b> ,93	-0.04	J . 0 ii	<b>-0.4</b> 6	0.00	₩0.00
					-0.42	0.00	0.00
	110.00	``-48 <sub>+</sub> 15	~u.06	-0.00	~D•39	~0.ua	'-0.00
	115+00	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	-0.00	-0.47		
	120.00	-52.72	-0.07	-0-00	-0.45	-0.00	-0.00
	125.00	~55+00			-0+42	0+01	0.00
	130.00	-57.08	*U.15	~0.00	-0+41	-0.00	0.00
	134+50	-58-89			-0+40	# O = O O =**	0.0 <u>.</u>
	135.00	-59 · O8	, ~0.15	-0.00	-0-37	O。UD	-0.00
<del></del>	140.00	59-31		-0.0.0-	-0.01		0.00
1	) 140.11 ·	+59+31 -59+32		-0.00 -0.00	-0.01 0.01		0.00
	144.41	-59,31	<b>~0.</b> 07	-0.00	0.00	0.00	-0.00-
3	144.49	-59-31	-0.07	-0.00 -0.00	0.00	0.00	-0.00 -0.00

Inboard Cutoff
 Outboard Cutoff
 Separation

TABLE 1C

	S-IB POWERED FLIGHT TABLES									
	TIME	CHIP	CHIY	снія	LONG	DECL	LATT			
	(SEC)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)			
_		······································			· ·					
	-5 <sub>V</sub> 00		<del>-0+00</del>			<del>28+36</del>	<del>28 • 5 2 -</del>			
	0.00	0.00	0.00	-28.0Q	-80.56	28+36	28.52			
	5+00	0.0-4	-0-0-0-0	28+00		28.36	<u>28.52</u> -			
	10.00	0.00	0.00.	-28.00	-80.56	28.36	28.52			
	15,00		-0.00	2-3 + 2 0		-24.36-	28+52-			
	120.00 .	1.66	0.00	-18.20	<b>~</b> 80∙56	28.36	20.52			
	25.00	<del></del>	<del></del>	<del>-13-20</del>	<del></del>	<del>-2.8                                    </del>	28-62-			
	30.00	- 4.93	0.00	-8.20	-80.56	28+36	28.52			
	35+00	7 q-l-l-	-0+00-	3 + 20		-28 + 36-	28.52_			
	40.00	- 9.58	0.00	0.00	-80.54	28 • 36	28.52			
	45-00	1-2 v-3 0	-0.00-	<b>—</b> 0→00 <b>—</b> —		-28.36-	28-52-			
	50.00	-15.21	0.00	0.00	-80.56	28.36	28.52			
	55v00	-18 v 27	<del>~b~0</del> 0~	<del></del>		-28-06-	20+62-			
	60.00	-21.43	0.00	0.00	#80+S5	28.36	28.53			
	65+DO	24 +6-2	~ა.00-			-28 • 37	28+53-			
	70.00	- 27.79	0.00	0.00	<b>≈80+53</b>	28.37	28.53			
	75+00	30+90	~D-∈0:0~	-0.00-		-28.37-	28+53			
102	80.00	- 33.89	0.00	0.00	-80.51	28,38	28.54			
	35.00	-36+76	<del>-000-</del>	-0-00-	8D+5O	-26-30-				
	90.00	-39.47	.0 • 0.	O • O t	-80.48	28.39	28.55			
	<del>95</del> v00	42 0 0-3	-0.00-	-0.90	———- B U + 4 6	-28+-39	28,55-			
	100.00	-44.41	0.00	0.00	·-80·43	28.40	28.56			
		46.60	~0.00-	0,00		-28 + 41	28.57-			
	110.00	-40.59	0.00	0.00	-80.37	28.42	28.58			
	<u>  \5.80</u>		<del>-0+00-</del>	<del>-0.00</del>		-2-8-4-3-	<del> 28 - 5</del>			
	120.00	- 53.21	0.00	0.00	<b>⇔80•28</b>	28.44	28 • 60			
			-0.00-			-28+46-	28.62-			
	. 130.00	- 57.50	0.00	0.00	-80+17	28.47	28 • 63			
	134.50		<del>-0</del> -00-	-0·-00		-28+49-	-28+65-			
	17,777			2.00		8.5 4.6	20 (5			

0.00

-3×00-

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\_\_\_\_\_

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-80 · 10

-80.03

-79.96

-79.96

<del>-79-98---</del>

28.49

-28-51-

28.51

28.63-

28.53

28.53\_

28.65

---28-57-

28.67

28:69

-28,69\_

1) Inboard Cutoff

135.00

140-00

144.41

1) 140.11

3) 144.49

- 59.32

<del>~59,√32</del>

- 59.32

\_59.32

- 59.32

-58-32

2) Outboard Cutoff

3) Separation

. S-IB POWERED FLIGHT TABLES								
(SEC)	(0)	. (L8)	(DEG)	(LB/F12) .	(FT/S)	(FT/52		
-5.00	0.00		N/A		0.00	0.0		
0.00	0.00	5.	N/A	9.	3.99 41.88	39+1 41+2		
500	<u>0.04</u>		4.70		70 • 2 4	42-4		
10.00	0.08	2689. 6798.	2 • 07 L • 00	24.	145.34	43.8		
15.00	0.13	10202.	. C.48	47•	207.42	45 - 2		
20.00 	0.10	10202.	<u> </u>	00	376.92	45.7		
30.00	0.32	17285.	0.01	125.	353.72	48+1		
35.00	0.39	25675	0.32	182	438,45	49 47		
40.00	0.48	32842.	-0.41	248+	532.03	. 51.4		
46,00	0,58	40755		324	635 • 1 S	53.3		
50.00	0,69	49651+	-0.58	405	748 + 69	55+2		
<del>-55-00</del>	0~02	67421.	-0.62	486		57.5		
40.00	0.96	118843.	-0.67	559÷	1005.20	5,00 6,000		
<u> </u>	<del></del>	1.657-7-9-,	· -0.94	645 *	1287.91	61.6		
70.00	1,29	150205.		660	1458457	65+2		
75.00		130+26. 110071.	-0.37	645.	1661.07	69.1		
60.00 85.00	1,76				1-8-7-9-6-3	7304		
90.00	2.28	70179.	0.14	495 •	2161.46	77 • 4		
95+00	2,53	52568.		397 *	2480.74	8 1.0 P		
-100.00	2.31	× 30561.	0.03	310+ . **	2744.80	86+4		
-105-00		2-7-250 ·	O + O 2	235	70 0 7-3 0 7-	9: *3		
110.00	3.41	18399,	-0.14	172+	3424.96	96.3 20147		
415.00	3+7-2			122*	3813+28	197+6		
120.00	4 + 0 5	4892.	<b>~0</b> , 9 9	- 85 • - 58 •	4234.94 4691.19	10/+6		
1-25.00				39*	5184.61	12101		
130.00	4.79	-1650.	- i + 7 [ 	28.	5663.00	128 0		
	5.24 5.32	-31-2-1 · · · · · · · · · · · · · · · · · · ·	-2:32	27 •	5718.28	128 - 8		
135.00	5.34 60U	-3200+ -5197			6294138	37.4		
140.11	6.01	-5163.	-1.08	18+	6309.63	137+6		
-142011	<del></del>	3.726		13	6490403	62+		
144.41	6.35	~3143,	-0.04	11+	6491,40	309		
144.49		-3110			490.39			

<sup>1)</sup> Inboard Cutoff
2) Outboard Cutoff
3) Separation

·			TABLE 2C	······································	<del></del>
		ULLAG	AND J-2 BUILD-UP TABL	LES '	
TIME	ALT	RKR	VTH•	V V V •	A Z o
(SEC) .	(FT)	(FT)	(DEG)	. (FT/S)	, (DEG)
144.49	203396.	21113067	63,42	7630.72	75.70
1.45 + 81	207888.	21117552. 21125146.	63.68 	7613•32 7588 <del>+</del> 32	75.71 
149.41	219882.	21129530,	64.35	7592 • 17	75.74

			TABLE 20			,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	
ULLAGE AND J-2 BUILD-UP TABLES							
TIHE "	MASS	THRUST	VTHE	AZI '	RANGE	YVYE	
(SEC)	(63)	. (LB)	. (DEG)	(DEG) ,	(FT)	(FT/S)	
	<del></del>		·				
145+81	305744,49	9240. 98266.	58.56 	72.25 72.27	210468.	6473•12 ·	
149+41	305127.38	188491.	59.34	72.29	230162.	6443.46	

	· · · · · · · · · · · · · · · · · · ·		TABLE 2C		···	
		ULLAG	E AND J-2 BUILD-UP TAB	LES	•	
TIME	X	Υ.	Z	οx	ργ.	DZ
(SEC)	(FT)	(FT)	(FT)	(FT/S)	(FT/S)	(FT/S)
144.49 145.81 148.08	21109331 • 21113649 • 21120946 • .	116027. 117343.	377618. 388699. 404313.	3289.19 3248.09 3179.83	389+28 389+02 388+71	6874:42 6874•68 6878•97
149+41	21125151.	118742.	413466.	3149.42	368.48	6897•19

			TABLE 2C `		······································		
ULLAGE AND J-2 BUILD-UP TABLES							
TIME	ΧĘ	, YE	ZE .	DXE	DYE	OZE	
(SEC)	(FT) ;	.(FT)	(FT)	(FT/S)	· (FT/S)	(FT/S)	
144-49	202402.		205039.	3365.34	3+5.7	5653·66	
145.81	206821• 214295•	-105. -95.	212375. 224989.	3325 • 25 3258 • 77	3 • 9 2 	5553 • 73 5557 • 72	
149.41	218606.	-89.	232386,	3229.55	4.88	5575.68	

	_
TABLE	2C

ULLAGE AND J-2 BUILD-UP TABLES								
TIME .	PHIP	PHIY	рній	DPH1P	. DPHIA	DPH18		
(SEC)	(DEG)	(DEG)	(DEG)	(DEG/5)	(DEG/S)	(DEG/S)		
144.49	#58.31	<u>~D.07</u>	-0.00	0.00	0.00	~0.00		
145.81	-59.30 ·	-U.06 -U.05	÷0+00	0.02	0.01	20 • 0 		
148.08	-59.03	-0.10	0.06	8Q+Q,	<b>-</b> 0+06 .	0.05		

TABLE 2C

ULLAGE AND J-2 BUILD-UP TABLES								
TIME	CHIP	CHIY	CHIR	LONG	. DECL	LATT		
(SEC)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)	(DEG)		
1"				······································				
144,49	52.12			~79.86	2-6-53	<u>2</u> 8-6-7-		
145.81	-59,32	0.00	0.00	<b>≈79.94</b>	28.54	28,70		
-1-4-80-8			————O⊸ O O	7-9·+·9·0		28.57.1.		
149,41	-59.32	0.00	0.00	79.88	28.56	20.72		

<u></u>	TABLE 2C									
TIME	MACH	DRAG	age-and-j-	BUTT-D-UP ALP	TABLES QQQ -	. VR ,	TACEL			
(SEC)	(U)	(L8)	ł	(DEG)	· (LB/FT2)	(FT/S)	(FT/S2)			
	6.43.6	493.		0-0-2		6490+3 <del>7</del>	0+9;			
145.81	6.39	412.	•	8.30	9 <b>*</b>	6469.76 - 6439,28	0 • 9 3 3 <del>- 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - </del>			
148 <del>,08</del> 149•41.	6,46	304¥ 257₄	. 1	1 • 4 1	. 5.	. 6440.08	19.85			

TABLE 3C

		S-IV	B POWERED FLIGHT TAB	LES		
TIME	ALT	RKR	VTH•	AAA	AZ?	RANGLI
(SEC)	(FT)	· . (FT)	. (DEG)'	(FT/S) .	(DEG)	(DEG
1) 149,41	219872	21127530	64.35	7592-17	75-74	0 > 6
150.00	221821.	21131466.	64.46	7596+02	75.74	0.64
j-5 6 <del>+ 4</del> -l	242491,	<del>211:521:05</del>		7657+7-3	75,73	O+7 *
163.11	263332+	21172964.	66.70	7733.82	75.73	0 : 84
<del></del>	284131	24193679.		7822+90	75.72	0.95
190.00	340625.	21250071.	70.60	8103+45	75.93	1 - 2 5
210-00	392110-	213014504	7-3-1-0	<u> </u>	76.17	1.63
230+00	433911.	21348142.	75.36 77.39	5789•49 91.97•03	76.44	2 • 0 l
250-00 <del>.</del> 270-00	519517•	21390 <del>417</del> 21428518.	79.22	9647.68	77.04	2 • 6 3
290-00	553813.	21-462693-		10143+37	77.36	3.26
310.00	584413.	21493168.	82.28	10686+32	77+79 -	3.74
		21520:34	83-54	1-276.79	7.8-(15	4,27
350.00	635489.	21543944.	84.64	11919 84	78.42	4 • 8 1
370 v 00	656459	2-15644-1-9	85.58	1 2 6 1 7 . 9 7	7-8 - 8-1	5 - 3 9
390.00	674728.	21582949.	86.38	13374.60	79.22	5 = 00
4-1-000		21598 <del>477</del>	₩ 7-+05	1-4198+98	7.9. 66	——
430,00	704347.	21612301.	87.59	15094+53	80.12	7,37
<del></del>	7-14377	21624163		1.6072-29	80-61	8-12
2) 454.75	719013.	21626764.	88 - 10	16318+15 .	80.73	8,31
<del>470+00</del>		2-1 6-3-4-3-20- <del></del>	88+49	16969-89	8·1-•-1-3	
490.00	734549+	21642079*	88.94	17858+79	81+68	. 9.78
510+00	740424		39,27	1 8 8 1.9- 0 5	32,26	10+68
530.00	. 744446.	21651647.	89.56	19863.18	82.86 83.61	11 • 63
560+00	746925	21653887	89.94	21001.68 22248.47	84.20	13.71
. 570,00 590.00	747901.	21654830. 21654955.	95.01	23522+31	84.93	130/1 130/114~85
610.00	747885.	21654554.	90.04	25198+92	85.71	16.07
3) 614-63	747893*	21654501	90.01	25,140,72 25527.13	85.90	، ده د مدمدا ــــــــــــــــــــــــــــــــــ
4) 624.63	7478/1.	21654485.	90.00	25549 • 23	86,31	17,00

<sup>1) 90%</sup> Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

TABLE 3C

	S-IVB POWERED FLIGHT TABLES							
	TIME ^	MASS	THRUST .	VTHE	AZI	RANGE '	YVVE	
	(SEC)	(FB)	(LB)	(DEG)	(DEG)	(FT)	(FŢ/S)	
	1) 149,41	305127.30	188491	59.34	72.29	230162	6943446	
	150.00	304835.79	197217.	59.46	72 - 29	233401.	6446.00	
	156.41	301593 + 43	227243-		72.36		6493499	
•	163+11	297808.85	227233.	62.18	72+43	306917.	6556+66	
	170+00	285139.55	227248.	63.55	72•49	346830	6632.93	
	190.00	274437.63	. 226516.	66.98	72+94	467742 6	6882.35	
	210+00	<del>263711,34</del>	227401	70.05	7-3 - 41	596265	7179+92	
	230.00	252992.89	227263.	72.82	73.89	732849.	7522+60	
	250+0Q	242268.85	226405	75.31	74.39	876006+	7913.55	
	270.00	231544,65	227313.	77.52	14.89	1032313.	8350.93 -	
	290.00	2208/2+1-2	276251	7.9 . 48	75 <b>.</b> 39	1196423	8836.06	
	310.00	210170+11	227167.	81.19	75+90	1371038.	9370+70	
	230.00	<del></del>	225705	<del></del>	7 6 • 4·1	1-556.942	9954 <del>+63</del>	
	350.00	185794 469	226543.	83.96	76 • 93	1754973.	10592+57	
	37-0+00	1-7 8-1 0 9 + 2 5	2262-38 <b></b>		77-+46	196608B		
	390+00	167440.08	226170.	85.98	78 • 01	2191303.	12040+21	
			2-2-5-1-9- <del>2</del>	<del></del>	78 • <del>5</del> 7	24317-7-7	12862•07	
H-4	430.00	146089.71	226014	87.36	79 • 15	2088814.	13755.77	
<del></del>	450+0 <u>0</u>	135422.21	224655	87.84	79.75	<u>2963850+</u>	14732 ~ 00	
• -	2) 454•75·	132909.98	226627.	87.93	79.89	3031997•	14977 654	
	470+00	1-26482+26		88.36	80-36	3257.436	15628+35	
	490.00	118352.99	175194.	86.85	81.00	3567718.	16516+16 -	
	5-10+00	1-1-0225+04	1-7-5 1-5 3	89-22		3895767	1.7475 • 70	
	530.00	102096.88	175109.	89.53	82.34	4243091•	18519.32	
	550-00	93974+96	175048.	89.78	8 <u>3</u> ~Q&	961-1421+	<u>19657.47</u>	
	570.00	85854,08	174961.	89.94	83 • 82	5002720.	20904.03	
	590+00	77/38-29	174805.	9 O • O-I	84.64	5419265	2227.7.673	
	610+00	69632.61	174503.	90.04	35 • 47	5863750.	23804 • 27	
	3) 614+63	67757+00	174397	90-01	85.47	5971037	24162 <u>~46</u>	
	4) 624.63	67617.40	٥.	90 • 0 O	86.11	6204635.	24204,52	

1	\Qn <sup>n</sup>	. Thi	·net

<sup>2)</sup> Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

		· · · · · · · · · · · · · · · · · · ·	TABLE-3C	····		
			IVB POWERED FLIGHT TAI	BLES		•
TIME	ΧĘ	· YE	ZE	DXE	DYE	DZE
(SEC)	(FT)	(FT)	(FT)	(FT/S)	(FT/S)	(FT/5)
1 <del>}148+41</del>	<del>218484</del>		2-3-2-3 8-6- <b></b>	3-2-2-9+55 <u>-</u>	4-88	<del>5575</del> ♦68
150 • 00 	220508. 	-66. -51.	235678. 27[858.	3217.72	5 • 0 2 5 • 9 0	5585 • 45 5706 • 87
163+11 -170+00	261106+ ** 28122 <del>2</del> +	-8. -44.	310542.	2979 • 16 2860 • 19	6 • 9 8 7 • 8 2	5840 • 74 5984 • 56
190.00 -210.60	335320 • 383468 •	473.	475026.	2554•44	37.26	6390+63
230.00 · -250.00-	425827• 462480	3378.	747695. 897523	1974 • 59	111.50	7257•97 7729•28
270 • 00 -2.50 • 00	493455. ~~ 518750.	9632.	1057050.	1407 • 29	203.82	8228 • 97 
310+00 -330+00	538305. 552026.	19888	1407725.	833 • 17 	311.61	9328 • 39
350.00 -370.00	559773 •	34781.	1805340.	235-18	436.09 505.49	10580+98
390+00 -#-10+00	556500. 544917.	55036. 67436.	2256674. 2504992.	-407•98 -753•89	580.16	12019.31
430.00 450.00	526189.	81515. 77414.	2770002.	-1122.90	748.20	13689.43
2) 454.75 470.00	492375•	101475.	3123098,	-1617 • 37	843.08 866.85	<del>14629→24</del> 14864•71
.490.00 -490.00	420765.	1347.90	3354520. 3672172.	-2464.57	933.90 1025.05	1.5475.52. 16299.04
530.00 530.00	366305. 300797.	156236	4359874.	-2989. <del>72</del> -3570.89	1222.94	17.181-057- 18130-59
570.00	223075. 131904.	233117.	4732589. 5126629.	-42 <u>1-1-68</u>	1457 <sub>1</sub> 16	19154÷54 20265÷22
570.00 610.00	25900	263578. 296811.	5543874 5986469.	-6589·11	15.90 • 9.7——— 1735 • 33	21478403 22808+23
3) 614.63 4) 624.63	-127925 · · · · · · · · · · · · · · · · · · ·	304934 322736.	609.2403. 6324103.	-7060•94	1771-+33 1787-69	23139•66 23082•59

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

TÁBLE 3C S-IVB POWERED FLIGHT TABLES

	TIME	. X	<b>Y</b> '	Z	ρx ·	DY	ΟZ
	(SEC)	(FT)	· (FT)	(FT)	(FT/S)	(FT/S)	(FT/S)
•	•					<u> </u>	
	1) 149.41	21-125-15-1-	<del>-113742</del>	413466	3149+42	-388+48	<del>6897+19</del>
	150.00	21127005.	118971.	417538.	3137 • 04	388.35	6907+08
	156+41	<del>····21-146706 •</del>	-1-21-454	4621-95	3012+1-1-	38615	7029.85
	163.11	21166463.	124033.	509/47.	2885•68	383.84	7165+02
	1 <del>-70 - 0</del> 0	<del>2   1</del> 859   <del>1 - 1</del>	-1-26669	55961-2 <sub>+</sub>	2769-62-	381+04	7310°07
	190.00	21237785.	134453.	709904.	2432+33	399.25	7719 • 47
	5 <del>-1-0 - 0 0</del>	<del>212832</del> 40+	<del>142650 -</del>	866505	2116-48-	<del>-421 - 18</del>	-8144.19
	230+00	21322461.	151327.	1035818.	1803.94	447.04	8590•76
	250+00	2-1-355432	-1-6055-1	1-212309	1492.064	_475.80	9062×61
	?70•00·	21382155.	170374.	1398500.	1179 • 69	506.64	9561 . 86
	5-40 <del>+ 00</del>	21-40-250-2	-180038	1-5 9 4 9.9-2	864-+8Q	_539-4-9-2	-10092+26-
	310+00	21416599.	191985.	1802423.	538 • 67	575-10	10657+23
	<del>}34+88</del>	<del></del>	<del>-203655</del> -	<del>202-1-5-1-/</del>	2 065-7	-612-28	-1-1-258428-
	350+00	21424757.	216493.	2253036.	<b>~137.99</b>	651.83	11901+21
	<del>}70+00</del>	21418433	-2 2 9 9-4 5	2497-862		693+-78	-1-2589.+-07-
	390.00	21404763.	244261.	2756919.	-873.92	738.35	13325.58
14	410.00	2-1-383-332	-259499	3031273	1272-48-	785.98	-14119490-
7	430.00	21353667.	275721.	3322114.	-1698·8Š	836.87	14975+25,
		21-31-51-64-	292949	-3630753	<del>-~2167-37</del>	-851 - 64	15901-87
	2) 454.75	2,304646.	297267.	3706635.	-2271 • 79	905.27	16133+87
	<del></del>	21-266992	-311346	-3957S16+-	2672.18	-940 a-93	-16731.082
	490.00	21208002.	330638.	4300105.	-3232+36	988.23	17536+01
		2-1-1-37488 ←-	350878 <del></del>	4659328	3828.18	4036+05	_15396 oH2_
	530.00	21054460.	372099.	5036376.	-4465.49	1086+64	19319+56
	<del>- 550+00</del>	<del>26°57633</del> .	<del>-3-11-2-11-4</del>	<del>5432578</del>	5218.54	-1-1-1-7-2	-20313.50-
	570.00	20645646.	417802.	5849459.	<b>~6002•92</b>	1201,93	21389,59
	590·00	207-1-6967	442486+-	6288002	86.0°.38	1267-34	_22562.53_
	610.00	20569450.	468501.	6752624.	<b>~7878.05</b>	1335.92	23845.76
	3) 614.63	20532408+	-47-4-7-3-1	6863839	_=8112+87	1353.28	<u> 24165 ° 77</u>
	4) 624.63	20449802•	488243.	7105207+	-8402+99	1348.19	24090115

<sup>1) 90%</sup> Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

TABLE 3C

		S-IV	B POWERED FLIGHT TABLE	S	•	
TIME	PHIP '	· PHIY	рнік	DPHIP	орніч	DPHIR
(SEC).	(DEG)	. (DEG)	(DEG)	(DEG/S)	(DEG/S)	(DEG/S)
.1) 149,41	-59-03-	=0 <u>10</u>	0.06	O_DA		0-05
150.00	-58-99	-0+14	0+09	0 • 0 6	<b>*0.08</b>	0.07
156.41	59.OS.	.=0,35	03.9	<del>-</del> _0 • 03	0.02_	0 • 0 •
163.11	-59.14	#0.31	0.45	-0.01	-0.00	0.0
1.70.00	-58.87-	.m.Q • D.8	0+66	O + 4 H	O a.H 4	. و و و ســـــــــــــــــــــــــــــــ
190.00	-55+24	Z+73	-0+33	-0+05	ប្លុស្បាប	-0.09
210.00	-54-39-	-3-03-			0.3	Q.O.
230.00	-57.96	3,28	0 * 26	−ប•ជ8	10.01	<b>"0.0</b>
250.00	×59.92-	-3.42-		<u> </u>	O • O \	
270.00	-61.86	3.50	-0.07	-0-12	0.00	' 0.0!
290±00	-64.00-	-3-55-	0-11		O • O O	
310.00	-66.26	3.59	-0+67	-0.12	0.00	~O • O •
330.00	6854-	36-3	Q3-J			O + O +
350.00	<b>~70+97</b>	3.65	-0.22	-0-12	0+00	<b>~0+</b> 0!
370.00	7.339	-3.67	u <u>2</u> 2 2		0.00_	.0 م 0 . ـــــ
390.00	<b>⇒75.8</b> 5	3.68	. 0 • 3 0	-0.13	0.00	<b>~</b> Q • O •
410.00-	-78 - 53-	-3-69	U • S 3			
430100	-81.13	3.69	-0.05	-0-13	0.00	0.0
450.00	-8.3 - 8.3 -	-3-6-9-			0	الاموسي
2) 454.75	₩ 8 4 4 8	3.69	-0.05	-0-14	<b>~0.ò</b> ¤	<b>~</b> 0 ⋅ 0 :
47.0.00	-1 P- 66m	3-+8-1	~U * 7 4			0 + 0 .
490.00	-69.23	3.63	0 + 2 9	-0.08	<b>#0.01</b>	0.09
510.00-	× 9.2 + 24-	3-o-5 Z	D.U2	0 • 18		: 0 a 0 مثنسب
530.00	<b>∞</b> 95.57	- 3 + 5 2	-0.45	⇔0•14	0.00	0.0
550,00-	Ed- 22.	3-54			0.00	OO
570.00	101.32	3 + 5 2	-0.69	,-0•1 <sup>4</sup>	<b>~</b> 0.00	+0.0
5.90.00	404.56	3· <sub>*</sub> -3 9	0 , 25	=0.•.7.0	<u>≈0</u> ~08	0.1.
610.00	: 05.88	3,37	-0.20	0.02	0.00	*0+0
3) 614.63	165.37	-3+3-7	U.+ 3.6	0.0		0.0:
4) 624.63	105.86	3.36	-0.69	0.00	<b>-8.00</b>	<b>~</b> 0 + 0 3

<sup>1) 90%</sup> Thrust
2) Mixture Ratio Shift
3) Guidance Cutoff
4) Orbit Insertion

. TABLE 3C

S-IVE POWERED FLIGHT TABLES

TIME	CHIP	CHIA	CHIR	LONG	DECL	LATT
(SEC)	(DEG)	. (DEG)	(DEG)	(DEG)	(DEG)	(DEG)
		, <del>*****</del>	<del></del>	79.88	<del>28+56</del>	-20.72
150.00	59.32	0.00	0.00	-79 - 87	28.56	28.72
156.41	5.93.2-	-0-y O O	<del>0+00</del>	·7.9 · 7 6	28 • 5 9	20.J.5
163+11	-59.32	0.00	0.00	*79 • 65 <u>.</u>	28.62	28.78
170+00	<del></del>	7 <u>5</u>	-0.00-	-7.9 + S 3	28+65	-28.81-
190+00	-55.40	.3.11	0.00	•79 • 17	28.75	28 + 91
210.00	<del>56,61-</del>	<del>-3-,-4-5</del>	<del>- 0+00</del>	<del>~78 + 7 ?</del>	<del>28 + 05</del>	2 <del>9+01</del>
230.00	-58.18	.3.69	0.00	-78-38	28.96	29+12
250.00		<u> </u>	0+00	•7-7-• <del>9-4</del>	29.07	29 • 23
270.00	- 62 · 14.	.3.93	0.0∪	•77•47	29.18	29 • 34
290.00		4-00-	0.00	·76 •-9 J	29.29	ــــ5 إلـ م 29 ـــ
310.00	-66.55	·4 • O5	0.00	•76 • <del>4</del> 4	29+41	29•57
330.00	<del></del>	<del>-4</del>	<del>-0+0-0</del>	<del>-75-07</del>	<del>2-9 + 5.3</del>	-29-70-
350.00	71.22	-4-13	0.00	•75•27	29 4 6 6	29.82
370+00	<del>7</del> 3 • 65-	►4 e 1-6	0+00	·74 · 62	29	29+95
390.D0	-76.12	.4.19	0.00	•73•42	29.92	30.08
410+09	78.61	<del>-4 6-2</del> -1		•-7-31-8- <del></del>	30+05	30 - 51
430.00	- 61 - 42	.4.23	0.00	•72•38	30.19	30 • 35
450.00		<del>-424</del>	<del>0</del> -00	-7-15-2	<del>30,33</del>	<del>30+45</del>
2) 454.75	- 84.78	.4.24	0.00	-71+30	30.36	30.52
470-00		4-3-7		·70 • 60—-	30-+47	-30+43
470.00	- 89.46	.4.19	0.00	•69 • 62	30 + 60	30.77
510-00		·4 »-1-0	0 o o o	-68+59	3074	30 - 90
530.00	- 95.90	4.12	0.00	•67 <b>+</b> 49	30.87	31.04
550.00	<del></del>	-4	<del></del>	-6-6-6-3-3	<del>-3-100</del>	
570.00	101.64	-4 - 1 3	0.00	*65 • O8	31.12	31.29
570-00	105,65	~3- <sub>4</sub> ~9	—-u~ 0 <del>0</del> —-	-6375	3-1-+-24	-31-040-
610.00	- 106.03	-4 • 0 1	0.00	-62+33	31.34	31.51
3) 614.63	106.03	·4-0-1		-61-99	3.1-+3.7	_31-53
4) 624.63	- 106.03	4.01	000	-61 + 24	31.41	31.58
		•		- In the state of		
1) 90% Thrust 2) Mixture Ratio	Chi ft					
3) Guidance Cutor		<del></del>	<del></del>			

TABLE 3C

		' s-IV	B POWERED FLIGHT TABL	ES .		
TIME	иасн .	DRAG .	ALP	କ୍ ୱ ଜ	٧R	TACEL
'(SEC)	(U)	(LB)	(DEG)	(LB/FT2)	(FT/S) .	(FT/52)
-1)149+41-		<del>257.</del> -	<del></del> 1-+4-1	<del></del>	<u> </u>	-1.9.65-
150.00	6.54	239.	1.59	5.	6442.61	20.79
	6 o 6 d	405.	_2 - 9.7	24	_6490.55_	_24.23
163.11	7 4 2 6	43+	4 • 4 1	1+	6553.17	24.55
170.00	<del>7</del> •50		6 8	c	6629+41	-25 064-
190.00	6.90	2 •	13.67	0 +	6878.72	26.56
<del>210-00-</del>	-5.78	<del>0-</del> -	1-6-00-	<del></del> 0	<del>7-1-7-5</del> +-2-2	-27-0-7-4
230,00	4.53	٥.	17.64	0 •	7510.84	28 • 90
250.0U-	4 • O 2·	<del></del> 0-y-	-18 - 67	0	<b>7909•75</b>	-30.07-
270.00.	3.85	0.	19.42	٥٠	8347.09	31.58
290,00-	—3 + 8 B—	<del></del> 0-+-	19.75-	0	_ 8832 . 19	-32.96-
310.00	4 • 0 0	0.	19.77	۵۰	9366+82	34.78
330+00		0	1-9-5-7	Q	9 9.5 0 7-3	-36+45-
350,00	4.35	0 •	19.02	0 •	10588.67	38.66
370°00	<u>-4,58</u> -	0 +-	48.35	0	-11282.80	-40.87-
390.00	4.84	υ <b>.</b>	17.48	۵.	12036,31	43.46
410±00	51 3	0 +	16.31	0 +	-12858 · 17	-46043-
430.00	5 • 45	. 0•	15.09	0 •	13751.87	49.78
450+00-	<del>-5+80</del>	<del>0</del> -	13+69-	0	-1-47-281-0	-53-83-
2) 454.75	5.89	0,	13.34	0 •	r4973.65	54.86
470.00-	<del></del> 61-2	<del></del> 0- <del></del> -	12.03	O ^	-15624.46	-44.57_
490.00	6 • 4 4	D •	11+11	0 •	16512.27	47.63
510+00-	<u>-6.7.9</u>	U +	<u>-9 • 4 4</u>	0	-17471 082	-51 • 13 -
530.00 <sub>.</sub>	7 • 18	D •	7 • 46	<b>0</b> •	18515.45	55 - 18
<u>\$50.00</u>	<del>-7.62</del>	<b>Q-</b>	<del></del>	0	19653.61	59,93
570.00	8.10	0 •	4 • 3 5	0 •	20900.18	65,57
590÷00	-8.63-	0-	_2.39	0 4	22273.89	.7.2 · 35
610.00	9 • 2 2	0+	2 - 40	٥.	23800.44	80.63
_3)614+63	<del></del>	<del></del> 0- <del></del>	-2+69	0	24178 - 64	82.81_
4) 624.63	9438	О.	3,36	0 •	24200.71	0.00
				<del>,</del>		*******

1) 90% Thrust
2) Mixture Ratio Shift
3) Guidance Gutoff
4) Orbit Insertion

## TABLE 40

			:		
TIME	ALT	RRR	ORBITAL FLIGHT LISTING	<b>VVV</b> *	<b>4</b> 24
, (SEC.	(FT)	(FT)	(CEG)	(FT/S)	(CEG)
53	747871.	21654485.	96463	25549.21	86.31
830,00	750149.	21656916.	89.94	25550.82	94.84
1530.40	755282.	21664262.	89.89	25543.40	102.68
1233.00	763324.	21676,65.	89.85	25531.59	109.44
1430.00	774347.	21691572.	89.21	25515.83	114.75
1630.00	788357.	21709810.	89.78	25496.74	118.53
1830.00	855161.	21729068.	89.77	25475.13	120.81
203 ) . 00	824248.	21749979.	89.77	25452.Cl	121,64
2230.00	844734.	21769003.	89.79	25428.57	121.06
2430,00	865381.	21787499.	89.81	25406.16	119.06
2630.00	834691.	21802767.	89.85	25386.13	115.59
2830.00	901,74.	21814688.	85.09	25369.80	110.63
3037.00	913)52.	21822731*	89.93	25358.28	104.24
3230.00	919459.	21826564.	. 8 <b>9.</b> 98	25352.40	96.7C
3430,20	919610.	21826044.	90.03	25352.64	88.57
3633.30	913406.	21821215.	90.38	25359.00	8C.56
3839.00	901361.	218123-)2.	90.12	25371.07	73.37
4030.00	884529.	21799699.	90.16	25388.G¢	67.46
4230.00	864373.	21783971.	90.19	25408-80	63.02
4430.00	842578-	21765855.	90.21	25431.95	60.07
4630.00	820829.	2174624 ,.	90.23	25456.C4	58.57
4830.0D	80067( •	21726130.	90,22	25479.62	58.5C
5 (30 - 00	783526-	21706623.	9C.21	25501.41	59.84
5230.00	708814.	21688037.	90.19	25526.30	,62.65
5450.00	758253.	21673836.	90.15	25535.45	66.99
5630.00	751321.	21662568.	90.10	25546.29	72.85
5666.80	750373+	21660912.	90.09	25547.7€	74.09
5830.00	748245.	21656238.	90+23	25571.32	8C.C7
<b>6030.00</b>	750797•	21657252.	99.95	25573.82	88.18
6230.00	758∂7ċ•	21666025*	89.87	25565.24	96.46
6387.80	769203.	21678146.	89.81	25553-77	102.59
6430.00	772A9G.	21682118.	89.79	25549.67	104.13
6630+00	791389.	21704692.	89173	25526.40	110.61
6837+00	813126.	21731234.	89.68	25497.82	115.62
7030.00	838720.	21760882.	89.66	25465.37	119.09
7230.00	867046.	21791941.	89.65	2543C.59	121.08
7430.00	896921.	21822053.	99.66	25395.25	121.64
7630.00	926834.	21851335.	89.59	25361.19	12C.8C
7830.00	955002.	21876490.	89.74	25330.32	118.56

#### TABLE 4C

	RBITAL FLIGHT LISTING					
TIME	ALT	RAR	VTH*	V V V *	AZN	
(sec)	(FT)	(FT)	(CEG)	(FT/S)	(CEC:	
8030,00	979523.	21896861.	89.43	25304.41	114.87	
8437.00	.010476.	21919662.	85.55	25273.60	103.10	
8630.00	.014212.	21921096.	90.02	2527€•7€	95.55	
8830+00	.0 19247.	21915750.	90.10	25276.65	87.45	
9333.00	995785.	21903911.	90.17	25291.28	79.57	
9230.00	974742.	21886161.	90+23	25313.63	72.59	
9430.00	947673.	21863373.	90.28	25342.42	66.86	
9630.00	916563,	21836532.	90.32	25375.86	62.61	
9830+00	133669.	21807268.	90.34	25412-C1	59.81	
10030+00	51223.	21776749.	90.34	25443+66	58,50	
10230.00	21268.	21745687.	90.33	25483.72	58.57	
10495.00	88130.	21710357.	90+28	25524.49	60.67	
10495.00	88136.	21710357.	. 90.28	25524.49	60.87	
~10630,00	74865.	21694446*	90 • 25	25541.67	63.04	
10830.00	63183.	21675289+	90.18	25561.64	67.54	
11030.00	51596.	21662425*	90.11	25574.35	73.57	
11234.00	48981.	21656686*	90.442	25579.32	80-91	
11430.00	52549.	21658466.	89.94	25576.46	89+07	
11030.00	63446.	21667689*	89 - 80	25566+03	97-32	
11840.00	74609.	21684788.	89.78	25547.54	105.24	
12032,00	91988.	217-26119.	89+72	25524.66	111.28	
12240.00	15130.	21733929.	39 • ¢8	25494.61	116.26	
12432.00	8,396,92.	21762458.	.89 ₌ 66	25463.28	119.40	
12647.00	869472.	21794634.	.89.65	25427.09	121.26	
12832.00	-898164.	21823854*	99.67	25393.29	121-61	
13/4/200	929114.	21853245.	89.70	25358.21	120.54	
13232.00	935848	21876890*	89.75	25329.04	118.20	
13440.00	989802.	21897436.	89-81	25302.79	114-15	
13632.30	998401.	21910335.	89.88	25284.98	109.60	
13849.00	1019946.	21918654.	99.95	25273.90	102-01	
14/32-00	1012671.	21919404.	90 • • 3	25271.96	94.61	
14240.00	1096582.	21913213.	90.11	25279.98	86.17	
14432.00	992888.	21901328.	90.17	25293.82	78.68	
14640.00	970357.	21882412*	90 - 24	25317.71	71.57	
14832.00	943960.	21860193*	90.28	25345.77	66.28	
15040.00	911440.	21832149.	90.32	25380.80	62.07	
15232.00	879931.	21803655.	90.34	25415.50	59.60	
1544-1-00	846525.	21772169.	90.34	25453.38	58.42	
15632.00	818223.	21743489.	90.33	25486.67	58.67	
15840.00	791978.	21714764.	90.29	25518.90	60.44	
16/32.00	.772831.	21691901.	90+24	25543-61	63,49	
16240.00	758197.	21672581.	90.17	25563.66	68.40	
16432.00	750476,	21660858.	90.10	25575.18	74.38	
) 16814.80	751076.	21657486.	89.94	25576.87	89.35	

2)

 <sup>5-</sup>TVB/CSM Separation
 End of Nominal IU Lifeti:

#### TABLE 4C

#### ORBITAL FLIGHT LISTING

		TIME	MASS	THRUST	VTHE	AZI	VVVE
		(SEC)	(L8)	(L8)	(DEG)	(DEG)	(FT/S)
		624-63	67617.40	· · · · · · · · · · · · · · · · · · ·	90.00	80.11	24204.52
		·830•00	67149.87 6697 <b>1.</b> 36	, <u>C</u> •	89.94 99.69	95.11 103.40	24206.08 24199.89
		1030.00 1239.00	66843.64	ე. მ.	89.84	113.56	2419C+23
		1430.00	66747.31	0.	89.80	116.23	24176.96
		1637.00	66672.30	۲. د	89.77	123.27	2416C+CC
		1830.00	66609.30	). ).	89.76	122.72	24135.53
		2931.00	66590.53	5.	89.76	123.61	24116.13
		2230.00	66586.59	5.	89.77	122.99	24090.85
		2430.00	6582.65	Õ.	89.80	123.64	24065.23
		2630.00	66578.71	5.	89.84	117.13	24041-14
		2830.00	66574.76	0.	89.88	111.84	24020.60
		3030.00	66570.82	0.	89.93	105.06	24005.52
		3230.00	o656¢•88	0.	89.98	97.C8	23557.40
		3450.00	66443.52	ប÷	90.03	88.48	23997.14
		3630.00	. 06375.53	9.	90.18	8C.02	24004.87
,_		3830.00	66369.53	0.	90.13	72.41	24019.95
120		4930.00	66363.53	0.	96.17	, 66 <b>-12</b>	24040.59
		4230.00	o6357.52	5.	96.23	61.38	24066.14
		4430.00	66351.52	<b>⊙</b> •	90.23	58.22	24093.23
		4630.00	66345.52	્ €∙	90.24	56.62	24120.14
		4830.00	66339.52	0	90,24	56.53	24145.C2
		5030.00	66333.52	9•્	90.22	57.98	24166.5C
		5230.00	66327.52	<b>⊙</b> ∜	96.20	61.00	24183.73
		5430.00	66321.52	€.	90.16	65.64	24196.40
		5630.00	66315.52	₽.	90.11	71.87	24204.62
	1)	5666.80	66314.41	_ა.	90.1)	73.18	24205.67
		5830.00	64647.95	75.	90.03	79.51	24227.53
		6030-00	64411.04	18.	89.54	88.68	24228.61
		6230.00	64277.26	10.	89.86	96.82	24220.15
	2)	6387.80	64237.25	20.	89.79	103.30	24209.38
		6430.00	64206.06	Ŏ.	89•78 . 89•71	104.93 111.81	24205.56 24183.95
		6630.00	64203.41	Ç.	89.66	117.15	24157.21
		6830.00	64194.77	<u>٠</u> ٠.	89.64	120.88	24126.08
		7030.00	64189.13	ö•` 0•	89.63	123,01	24091.55
		7230.00	64183.48 64177.84	ε.	89.64	123.62	24C55.CC
		7430+00 7630+00	64172.20		89.68	122.73	24018.26
		7830.00	64166.55	ζ.° 2•	89.73	120.33	23983.51
		1020.00	U71UU	\* <del>0</del>	0,415	* ** * * * * *	

<sup>1)</sup> Begin LOX Damp
2) End LOX Damp

#### TABLE 4C

			-	RBITAL FLIGHT LISTING		
	TIŅE	MASS	THRUST	VTFE	VZI	V V V E
	(SEC)	(LB)	LB)	(DEG)	(CEG)	(FT/S)
	8930.00	64160.91	٥.	89.79	116.37	23953.12
	8430.00	64149.62	C.	85.54	103.92	23914.46
	8630.09	64143.98	C.	90.12	95.87	`235GS•66
	8830.00	64138.34	<b>ĕ•</b>	90.1)	87.31	23515.68
	9930.00	64132.70	0.	90.18	78.97	23932.28
	9230.00	64127 <b>.</b> C6	a.	90.24	71.55	. 23958.34
	9430.00	64121.46	<b>⊙.</b>	90.30	65.47	23591.88
	9630+00	64115.83	<b>3.</b>	90.34	63.94	24030.39
	9830.02	64110.21	0•.	90.36	57.96	24071.04
	13/3/00	04174.59	્રે•	90.35	56.53	24111.61
	10230.00	64)98.97	Q.	95.35	56.61	24147.79
1)	10495.00	5409T.52		90.30	59.09	24188.29
	10495.00	31333,52	••	90.37	59.09	24188.29
	10630.00	31232.03	?∙	93.26	61.41	24204.41
	17830.00	31224+03	<u>٠</u> .	90.19	66.23	24222.13
	11)30.00.	31214.03	5.	90.11	72.63	24232.36
	11230.00	31208.03	, <u>(</u> 1) •	90.12	89.40	24235.37
	1143 1.00	31203.03	?∙	89.93	89.02	24231.35
	11637.00	31192.03	0. '	29 • 35	97.473	24220.99
	11840.00.	31183.63	0.	89.77	106.10	24203.68
	12032.00	31175.95	3.*	39.71	112.52	24182.52
	12240.00	31167.63	٠ <u>.</u>	39.46	117.84	24154.37
	12432.00	31159.95	<b>}•</b>	89.64	121.21	24124.21
	12640.00	31151.63	<b>9.</b>	89.63	123.21	24088.09
	12832.00	31143,95	₽•	89.05	123-60	24052.94
	13340.00	31135.63	0.	89.48	122.45	24C14.88
	13232.00	31127.95	1.	69.73	119.94	23981.86
	13440.00	51119.63	٥.	89.30	115.51	23950.90
	13032.00	31111.95	0.	89.87	11:412	23929.95
	13840-00	31103.63	∂•	89.95	102.71	23914.45
	14,32.00	31795.95	j•	90.3	94.87	23910.90
	14240.00	1 187 • 63	ō.	90.11	85.95	23918.32
	14432,00	31''79.95	્રે•	97.5	78.03	23935.24
	14640.00	31,71.63	0.	90.25	73.49	23963.14
	14832.30	31 )63 + 95	Ĵ• ,	90.37	64.85	23995.85
	15040.00	31755+63	o.	90.34	63-36	24636.08
	15232.00	31747.95	ર્ગ•	, 90.35	57.71	24074.98
	15440.00	31.39.63	2•	90.36	56.44	24116.11
	15632.00	31.131.95	ე.	90.34	56.72	24150.81
	15840.00	31023.63	ρ.	90.31	58.62	24182.87
	16032,00	31015.95	્ર•	90.25	61-93	24206.12
	16247.00	31707.63	<b>0.</b> .	91.15	67.14	24223.76
	16432.00	30999.95	٥ <b>٠</b>	90.10	73.49	24232.94
2)	16814,80	30984.64	<b>9.</b>	89.94	89.31	24231.81

S-IVB/CSM Separation
 End of Nominal IU Lifetime

TABLE 4C

#### ORBITAL FLIGHT LISTING

	Orbital flight listing								
TIME				£Χ	CY	£Ž			
(SEC)	(FT)	(FT)	(F1)	(F1/S)	(FT/\$)	(F1/5)			
024.63	2:445802.	488243.	7165227.	-8402.59	1348.19	24090-15			
830-00	1.145536.	749134.	11798322.	-13926.66	1179.73	21389.27			
1,30.00	14865165*	943001.	15711291.	-18525.76	\$45.31	17560.19			
123 `.00	13873710.	1125365.	18758074.	-22099.66	666.15	12766.67			
1430.00	6126865.	1226971.	25772166.	-24452,51	345.71	72 € 1 . C 2			
1630.00	1112702.	1262268.	216445C2•	-25458.Cl	5.5¢	1464.95			
1830.00	-1962998.	1229392.	21329849.	-25166.C6	+335.88	-4534.53			
233500	-8822261.	1129380.	19848287.	-23304.75	-655.88	-10210.67			
2230.00	-13277349*	907503*	1726381c.	-20277.53	-948.76	-15334.50			
2430.00	-15860187.	752955.	13779052+	-16155.55	-1167.12	-19571.60			
2630.00	-19674943*	496884.	95268C±•	-11168.53	-1362.15	-22756+43			
2830.00	-2:288154.	212946.	472917	-5588.81	~1464.54	-24702.18			
303 -00	-21823959.	-83714.	-26516:.	283.41	-148E.EC	-25312.55			
3230.00	-21176237+	~377261*	-9274835.	el33.95	+1433.52	-24957.36			
343: .00	-19389591.	-651987.	-9999743.	11651-85	-1201.27	-22476.61			
3630.00	-10507291*	-893113.	-14155342.	16543.87	-1698.94	-15187.75			
3833.00	-12881295.	-1.87353+	-17665444*	20548.59	-636.49	-14857.56			
4030.00	-0411579*	-1224595.	-20074177.	23449.31	527.45	-5716.22			
4236.00	-55365222*	·1296490 ·	-21455933.	25185.46	-187.87	-4036.31			
4430.00	1531799.	1298895.	-21672999.	25361.97	164.36	1676.13			
4654.30	651643L*	12311612	-20710366.	24256.01	510.25	7706.64			
4830,00	11145366.	1396429.	-18617287.	21820.05	831.25	13130.34			
5030.00	15164373.	~901524.	-15535342*	18180.84	1109.26	17847.52			
5230.00	16350372.	<u> 656933•</u>	-11547891.	19939.77	1326.72	21195.21			
5431.70	25527204#	-374786.	-6946134.	8132.59	1477.02	24160.69			
9630.00	21572936.	-7116C.	-196729c+	2275.36	1545.41	25397.79_			
5666.80	21636360.	-14205.	-1030937.	. 1172.48	1548.88	25473.77			
5830.00	21426551.	237965.	3122810.	-3710.70	1530.77	25854.30			
6034460	2.1 1394.	535625*	8C-2613.	-9495.98	1431.25	23702.28			
6231.00	17664807.	8:5356.	12518891.	-14753.22	1252.79	20841.17			
6387.80	15046253.	988633.	15574817.	-16335.73	1062.64	17767.02			
643C.00	14254214.	1;32292.	16365397.	-19191.57	1665.28	16827.27			
6030.00	1,059099.	1203929.	19195275.	-22562.69	702.34	11917.59			
6834.00	5311990.	1310730.	21031158.	-24689.28	360.70	6355.51			
7030.00	273673.	1346881.	21717433.	-25460.95	-C.89	474.66			
7730.00	-4779710.	1310361.	21227885.	-24845.04	-362.7C	-5412+65			
7430.30	-9574437.	1203394.	19573216.	-22885.83	-705.18	-10584.54			
7630.00	.13632134.	1030003.	16808216.	-19699.90	-1616-64	·1594C.CC			
7830.00	·17384530*	802566.	13255742.	-15467.62	-1261.19	-20015.67			
	· <del></del> -					•			

#### ORBITAL FLIGHT LISTING

	SMIT			Z :	cx	DY	CZ
	(SEÇ)	(FT)	(FT)	(F1)	(FI/S)	(FT/S)	(F7/S)
	8030.00	-19984845.	530790.	8932904.	-10421.54	1445.54	-23013.39
	8430,00	-219J1549.	-85158.	-880831.	1.402.48	1579.63	-25204.26
	8630.00	-21119856.	-396732.	-5859372.	6779-27	1522.32	-24296.76
	8830.00	-19213879.	-688691.	-10519157.	12190.23	1384.27	-22096.31
	9337+00	-15284315.	-945495.	-14618773.	16970.09	1172.19	-18716.C5 ·
	9230.00	-12435638.	1153319.	-17930304.	23343.45	-896.52	-14328.94
	9430,00	-3013713.	1300775.	-20298160.	23621.89	-571.65	-9160-69
	9030000	+33ر 121د⊶	1379522.	-21568377.	25135.18	-212.43	-3480.37
	,983C . 10	1945090.	-1384765.	-21676163,	25296.81	160.75	2411-64
	10 (3 ())	69 16137.	-1315587.	-20617742.	24086.17	528.53	8198.26
	10230.00	11491516.	-1175 5.7.	-18425756.	21557.42	870.89	13562.42
	1) 13495.00	10563054.	-591444.	-14007504.	16433.77	1253.11	19513.59
	10495.00	10563(154.	-891444.	-14007504.	10405.77	1253.11	19513.59
	17637.00	1856,7912.	-711534.	-11208738.	13129.01	1475.26	21863.92
	17634.00	2,652,56.	-4131 19.	-6568111.	7083.94	1566.58	24329.CC
	11030.00	216 (553).	-90071.	-1506375.	1805.40	1643.12	25457.57
	11230,00 4:437.00	.21357184.	238149.	3521197.	-4178.38	1630.01	25183.C4
		.9949541.	535361.	P414968.	-9932.21	1527.50	23519.64
	11637.00	17431163.	843536.	12842576.	-15134.26	1340.95	20561.57
123	11840.00	13756102.	1097538.	1672737 / .	-19680.84	1065.81	16247.13
ដើ	12032,00	9659598.	1273288.	19395549.	-228J2+31	757.13	11444.69
	1224JA 12432.11	4677181.	1392134.	21178990.	-24865.13	379.92	5613.09
	12040.00	-174919.	1429332.	21714732.	-25463.22	11.39	-52.53
	12837,00	-5414327-	1390267.	21,65565.	-24667.98	-388+73	-6154.40
		-9976526.	1281727.	1930793 .	<b>-22</b> 652 <b>.</b> 36	-738.33	~11451.84
	13040.00 13232.00	-1435 1961.	1092167.	16444495.	-19211.08	1075.48	-16517.17
	13440.00	~17653526.	800036.	128923)7.	-15) <i>5</i> ).39	1332-11	-20329.08
	13032.00	-2:242657.	500334.	8331782.	-9727.16	1536.29	-23307.80
	1384 1,00	-21596632.	253486.	3688625.	-432).60	1647.01	-2485E.6C
	14,32,50	-21804411. -2J993233.	-93809.	-1539229.	1756.81	1676.40	-25156.97
	1424).00	-18869125.	-411412.	-6297867.	7283.41	1618.44	-24146.40
	14432,00		-733795.	-11083330.	12854.65	1466.44	~21717.26
	146400	-13977335. -11940214.	-993719.	-14946943,	17356.14	1250.49	-18356.56
	14832.40	-11947214. -7594395.	-1225316.	-13296717.	21273.43	-946.33	-13694.25
	15747.30	-7594415. -2468895.	-1375517.	-2,1452425.	23810.56	-614.57	-8665.33
	15231.00		-1462952.	-21642724.	25233.70	-218.64	-2715.50
	15440.00	2397394.	-1468485.	-21021845.	25244.32	161.76	2940.36
	15632.50	7526584. 11874795.	-1392309.	-20382337.	23831.14	567.73	8923.80
	15840.50	1:874795.	-1249259-	-18171625.	21270.45	916+62	14010.64
	15552.33		-1023361.	-14749184.	17279.64	.245.23	18737.06
	16247.50	18791545. 25847991.	-760084.	-14839193.	12666.62	.485.97	22131.59
	16432.07		-430916.	-5937276.	6939-16	663.63	24547.53
	2) 16814.80	21632325. 19941127.	-1)207J. 5545J2.	~1136737.	1267.66	.741.22	25484.32
	~/ *OUX-T*VU	7224777 ( •	5545J4•	8431529.	-9953.80	.634.29	23503.77

<sup>1)</sup> S-IVB/CSM Separation
2) End of Nominal IU Lifetime

TABLE 4C

TIME .	×ε	YE	ORBITAL FLIGHT LISTING ZE	DX€	DYE	CZE
(SEC)	(FT)	(FT)	(FT)	(FT/S)	(FT/S)	(F1/S)
			(1),	(7)	(175)	(+1/5)
624-63	-197242.	322736.	6324193.	-7163.94	1787.69	23082.59
830.00	-2175593.	717393.	10855830,	-12119.64	2038.41	20854.10
1030.00	-5.42952.	1140120.	14721349.	-16434.22	2167.47	17631.03
1231.00	-8694609.	1575 -17.	17850217.	-19933.17	2156.35	13539.05
1430.00	-12948881.	1992115.	20091890.	-22435.78	1987.42	8786.98
1630.00	-17594735.	2358989.	21337052.	-2383J.51	1653.57	3616.19
1830-00	-224 2647.	2643956.	21527827.	-24050.69	1161-24	-1712.53
2030.00	-27136292.	28145)5.	20659281.	-23092•09	.526.69	-6932.95
2230.00	-31564389.	2846188.	18779397.	-21008.98	-220.70	-11787.41
2430.00	-35472055.	2720785.	15984530.	-1791J.12	-1041.93	-16039.86
2639.00	-30671J54.	2427606.	12416930.	-13951.90	-1890.14	-19487.13
2830.00	-41018473.	1966394.	825423 1.	-9329.57	-2713.32	-21967.85
3,/30.0) 3230.00	-42373498.	1347554.	3701902.	-4267.17	-3457.41	-23368.84
	-42792122.	592248.	-1017114.	993.14	-4C69.43	-23626.58
3439,00 3630,00	-41979685.	-268152.	-5673 <del>0</del> 01.	6201.76	-4500-77	-22740.80
3830.00	-4.1241319.	-1193250.	-10039802.	11112.13	-4711.33	-20750.09
4030.00	-37570287.	-2135421.	-13905876.	15491.20	-4667,41	-17753.76
4230.00	-34094335.	-3042167.	-17083765.	19128.62	-4354.33	-13895.94
	-29980391.	-358929.	-19419103.	21347.75	-3768.49	-9362.34
4430.00 4650.00	-25425648.	-4532257.	-20793207.	23512.04	-2923.71	-4372,58
	~20651500.	-5013157.	-21153933.	24334.73	-1850.73	829.35
4830500 5030500	-1,890143.	-5263427.	-20469415.	23383.46	- ~596.52	5986.6C
5230+30	-11374799.	-5243744.	-18779392.	21583.77	777.50	10842.33
5430+30 5430+00	-7327764.	-4946271.	-16168843.	18718.98	2198.48	15152.19
5630.00	-3949118.	~4366561.	-12708932.	14927.09	3586.09	18702-62
5666.80	-14/6418.	-3519390.	-8757488.	.J394.Ŭ9	4657.30	21312.74
583 ,00	-104-357.	-3336321.	-7959141.	9496.58	5072+55	21679.57
6350.00	175177.	-2435988.	-4312175.	5353.60	5935 + 87	22870.89
6230.00	717336.	-1162856.	322886.	40.82	6741.01	23272.14
6387.80	19232).	238993.	4923679,	-5270.22	7214.45	22512.05
6430.00	-959190.	1389672.	8374423.	-9284.17	7325.69	21124.22
6630.00	-1372720.	1698715.	9256107.	-10312.91	7314.07	20641-15
0830,00	-3897968.	3138586.	13113420.	-14835.76	7016.97	17762.C5
7030.00	-7257328 <b>.</b>	4479426.	1630584%	-18018.70	6323.43	14034.07
7230.J0	-11283568.	5643328.	13683404.	-21478.37	5256.49	9650,41
7430.00 7430.00	-15778)76.	6560050.	2013740).	-23282.89	3860.83	4837.76
7630.00	-2,523978.	7170039.	20606090,	-23953.52	2201.32	-161.56
	-2,282240.	7428355.	20076782.	-23467.47	358.72	-51C1.58
78300	-29832931.	7307492.	18585291.	-21857.78	-1574-26	-9745.15

TABLE 4C

	ORBITAL FLIGHT LISTING							
	TIME	ΧF	Y∉ ·	Zε	DXE	DYE	DZE	
	(Sec)	(FT)	(FT)	(FT)	(FT/S)	(FT/S)	(FT/\$)	
	6330.00	-32955994•	6799279.	16213562.	-19209.45	-3497.86	-13874.95	
	8430400	-43168249.	4689161.	9349104.	-11365.81	-6912.92	-19875.74	
	8630-00 8830-00	-41954661.	3170705.	5196442.	-6532.15	-8214.52	-21483.12	
		-42707280.	1428732.	824831.	-1391.65	-9136.19	-22057.95	
	9030.00	-42516358.	~454273.	-3556398.	3822.92	-9615.22	-21579.78	
	9231. 0	-4124233	-2384932.	-773-522.	8869 - • 8	-9608.82	-20075.16	
	9430.00	-33995824.	-4263993.	-11522777.	13509.10	-9097.27	-17616.87	
	9630.00 9830.00	-3,880,579.	-5990496.	-14729405.	17523.06	-8086.45	-14321.89	
		-32043001.	-7467425.	-17296150.	20704.20	-6609.22	-10347.42	
	10 (3 (.00	-27655041.	-8607535.	-1883578).	22899.28	-4725.59	-5884.94	
1)	1 (23 (.00 10495.00	-22957150.	-9336214.	-19542)63.	23983.18	-2521.11	-1151.95	
1)		-16632425.	-9531637.	-19009360.	23625.88	707.65	5137.18	
	1 495.JQ 1 630.70	-166 2425.	-9581637.	-19709365.	23625.88	7,7.65	5137-18	
	1 2021 - 20	-13472917.	-93717)1.	-18103714.	22649.85	2472.51	81489.37	
	1283 .00	-9102820.	-6643-18.	-16046825.	2)272.11	4862.09	12333.41	
	11030.00	-5431313.	-7439118.	-13216885.	16887.56	7138,06	15845.05	
	1123	-2463857.	-58 19115.	-976252	12663.39	9099.30	18552.39	
	11430.00	~4 17395.	-3828223.	-5858255.	7811.21	10628.94	20326.57	
	11630.00	6:4997.	-1592717.	-1699490.	2575.36	11631.79	21088.51	
	11840,00	585105.	905229.	2715755.	-3 145 - 65	12 44.47	20771.78	
	12/32.00	-483337.	3255939.	6597115.	-8038.61	11814.13	19509.01	
	12240400	-2632869.	5579921.	10428308.	-13313.94	10979.81	17176.93	
	12432.00	′ー5573930。	7547982.	13456829.	-16985.22	9503.83	14253.31	
	12640,00	-9479998.	9318512.	16033365.	-20404.88	7433.35	10422.20	
	12832.40	-13624237.	10328 444.	17,659502.	-22601.09	· 5178.51	6452.73	
	13141.11	-15474592.	11299517.	18529326.	-23831.85	2268.39	1900.51	
	15232.51	-23070315.	11409116.	18487166.	-23862.37	-520.38	-2332.67	
	13440.00	-27934403.	11043779.	17538311.	-22705.04	-3568.12	-6737.58	
	13032.10	-321 -6474.	10096.58.	15881762.	-20594.51	-6264.78	-10450.92	
	13840.00	-30053703.	8511932.	13336477.	-17292.1)	-8907.56	-13911.84	
	14732.00	-39025763.	6597+93.	10419694.	-13449.14	-10963.50	-16451.55	
	14240.00	-4133119	4130756.	6772841.	-8521.73	-12652.66	-18375.6C	
	14432.10	-42524)57 <b>.</b>	1598416.	3141222.	-3755-46	-13626+26	-19316.C4	
	14541.00	-42738246.	-1285433.	-901347.	1708.92	-13976.48	-19385.54	
	14832.00	-41928339.	944906.	-4563387.	0099.41	-13616.30	-18588.03	
	15340.00	~39997925•	-6572625.	-8259531.	11784-82	-124°5.48	-16821.53	
	15232.03	-3732503.).	-8916:42.	-1127 371.	15956.57	-10792.64	-14435.53	
	15441.00	-336.12565.	-10916313.	-13945778+	19680.92	-8333.15	-11171.68	
	15632.00	-29565844.	-12259083.	-15759489,	22210.63	-5588.31	-7662-93	
	15840.00	-24757263.	-15.79188.	-16926513.	23827.20	-2247.61	-3515.33	
	15.32,00	-2,131541.	-13198543.	-17221445.	24183.45	1020.05	449.24	
	15240.00	-15168751.	-12615595.	-16685009.	23323.79	457C.79	4679.77	
	16432-00	-13857399.	-11435282.	-15430033.	21416.44	7683-65	8338.64	
	16814.69	<u>.</u> 3815727 <b>.</b>	-7457560.	-11029742.	14839.53	12791.24	14291.35	

S-IVB/CSM Separation
 End of Nominal IU Lifetime

TABLE 4C

#### ORBITAL FLIGHT LISTING

		U	witer through preside	<i>5</i>		
TIME	PHIP	PHIY	PHIR	LØNG	DECL	LATT
'(SEC)	(DEG)	(DEC)	(DEG)	(056)	(DEG)	(CEG)
				4		
524 <b>-63</b>	-175.86	3.36	-0.49	-61.24	31.41	1.58
830-00	-123,12	2.04	-1.98	-45.83	31.27	1.44
1030.00	-136+64	2.13	-2.55	-31.19	29.20	9.36
12300 10	-15.1.14	1.49	-2.9	-17.42	25.44	5,58
1430.00	-163.61	2.77	-3.2	-4.68	20.34	0.46
1630.00	~177#f16	÷ ". βω	-3.3	7.09	14.28	4.36
1830.00	169.52	-3.77	-3.2	13.14	7.59	7.64
2 130.07	156.11	-1.50	-2.9.	28.81	0.61	2+62
2230.004	142.73	-2.15	-2 • 5·	39.42	-6.38	6.42
2430.06	129+35	-2.68	-1.9	53.32	-13.19	3.18
253 1 00	115.99	-3.38	-1.3	61.34	-19.25	9.37
2330.J7	1 12.63	-3.31	-0 , ž.	74.22	-24.51	4.65
3030.00	89+29	-3.37	0.2	87.59	-28.53	8.68
3230.00	75.90	-3.24	6.9	1)1.84	-30.96	1.12
3430+00	62 • 63.	-2.94	1.7	116.57	-31.58	1.74
3630.00	49.31	-2.49	2.3	131.29	-30.31	3.47
3830.00	35.99	-1.90	2.8	145.18	-27.28	7-43-
4530.00	22.07	-1.20	3.2	158.22	-22.79	?•92
4230,00	9.33	-2.44	3.4	173.29	-17.17	1.27
443.4.00	-4.52	₹.36	3.4	-178.43	-10.77	1.84
630.00	-17,40	1.14	3.2!	-167.66	-3.91	1.94
1839.00	-3).81	1.86	2	-157.25	3.12	3.14
3030.50	-44.26	2.48	2.3!	-140.26	10.04	:13
3230.00	-57.74	2.98	1.74	-134.97	16.55	1.65
143100	-71.25	3.31	0.9°	-122.86	22.31	
630.00	-34.78	3.47	0.14	-139.73	26.98	!.44 !.13
1666.80	-87.27	3.48	0. 14	-107.20	27.68	1.84
183 1. 20	-98.33	3.43	-J. 6.	-95.59	30.16	
,030.00	-111,89	3.21	-1.47	-83.72	31.56	:.32
1235.00	-125.43	2650	-2.1:	-65.73		-72
,387,80	-136:15	2.37	-2.02	-54.22	31.01	-17
1430.00	-138.95	2.24	-2.7?	-51.23	29.23	1.29
1030.00	-152.43	1.56	-3.1 <i>i</i>		28.58	.73
830.00	-165.67	1.79	~2+10 43.40	~37.66 -3= 1.4	24.52	-66
1936.00	-179,28	-v•v2	-3.40 -3.55	-25.16	19-21	1.33
2300	167.36	-0.84	-3.4!	-13.58	13.72	1.10
430.00	154.52	-7.64 -1.61		-2.67	6.28	-32
630.00	140.72	-2.30	-3.1t	7.90	-0.69	.69"
7830.00	127.43		-2.71	18.48	-7.62	• 66 ~
@\$(**)£	121042	-2.86	-2.1)	29,39	-14,22	.30

TABLE 4C

	ORBITAL FLIGHT LISTING							
TIME	PHIP	HIY	PHIR	l øng	DECL	LATT		
(\$80)	(DEC)	(DEG)	(DEG)	(DEG)	(086)	(DEG)		
8633.00	114.16	-3.28	1.39	41.95	-20.20	-20.32		
18430.00 8630.00	87.07	-3.58	0.22	66 <b>.</b> 80	-29.0(	-29.16		
8830.00	74.43 61.20	-3.45	1.04	81.04	-31.16	-31.33		
9036,00	51 • 20 49 • 80	-3.14	1.80	95.68	-31.51	-31.68		
9230.10	47.00 34.73	-2.68	2.46	11).14	-30.01	-3C.17		
9430.00	21.48	-2-74	3. ,2	123.93	-26.80	-26.95		
9630.00	21.48 8.2':	-1.31	3.41	136.77	-22.18	-22.31		
9836.00	-17.94	-0-50	3.52	148.67	-16.48	-16.58		
15537.00	-38.43	1.56 2.28	3.31	159.84	-10.:3	-1C.1G		
10230,00	-51.84	2.88	2.86	173.56	-3.16	-3-18		
17495.40	-54.53	2.99	2.25	-178.85	3.87	2.89		
1,495.00	-34.53	2.99	2.11	Ió4.41	12.91	12.59		
10030.00	-54.53	2.99	2.11	~164.41	12.51	12.99		
10330.00	-54153	2.99	2.11	-136.66	17.20	17.30		
11/37.00	-54:53	2.99	2.11	-144.44	22.87	23.00		
11230.00	-34÷53	2.99	2.11	-131:19	27-39	27.55		
11430.00	-54,53	2.99	2.11 2.11	-116.94	30.40	30.56		
11030,00	-54,53	2.99		-102.04	31.59	31.76		
11340.00	-47: 13	2.38	2.11 -12.39	-87.97	30.83	31.CO		
12,32,00	1,57	1.68		-71.96	28.04	28.19		
12243.00	12,40	₹ 83	-106.39	-59 - 38	23.96	24.10		
12432.00	-3,46	.0,400	-176.33 -176.23	-46.21	18.29	18.40		
12540.00	-14,36	0.90	-176.34	-35.21	12.25	12.33		
12832.00	-27,15	1.69	-176.63	-23.92	√ 5 <b>•19</b>	5.23		
13040.00	-43 , 99	2.45	· ·	p13.79	-1.51	-1.52		
13232.00	-53.74	3. 12	-177.13	-2.76	-8.69	-8.74		
134,40.00	-67,54	3.49	-177.74 -178.53	7.78	-14.96	-15.65		
13632,30	-87.26	3.74	-179.34	19.92	-21 - 17	-21.19		
13843.05	-94.54	5.60		31.99	-25.76	-25.90		
14032.00	-1.06.75	3.67	179.75	45.08	-29.46	-29.62		
14244.00	-12),52	3.33	178.92	59.83	~31.33	-31.47		
14432.00	-133,22	2.34	178.93	75.07	-31.40	-31.56		
14640.00	-146,99	2.16	177:39	38.89	-29.71	-29.87		
14832.00	-159.72	1.41	176.79	103.10	-26.15	-26.30		
15)40,00	-173,53	U.52	176.39	115.30	-21.55	-21.68		
19232.30	173.70	0.34 0.34	176.16	127.56	-15.50	-15.59		
15440.00	159.81	1.26	176.14	138.20	-9.23	-5.29		
15632.00	146.95	2+04	176.33	149.31	-2.04	-2.C6		
15846.00	132,97	2.78	176.70	159.49	4.70	4.73		
16)32,50	12/+ 12	4.70 -3.33	177.37	170.80	11.81	11.89		
16240.00	1 15. 96	-5.73	177.99	-178.19	17.92	18.03		
16432.00	92.95	-3.70 -3.90	178.86	-165.34	23.67	23.81		
16314.60	66,99	-3.40	179.73	-152.48	27.84	27.59		
	20, , ,	72400	-178-53	-124.48	31.60	31.76		

Manual control begins at 8995 and ends at 9175

<sup>1)</sup> S-IVB/CSM Separation 2) End of Nominal IU Lifetime

TABLE 5C S-IB STAGE REENTRY DATA

TIME	ALT	RRR	VTH*	VVV*	AZ#
(SEC)	(FT)	(FT)	(DEG)	(FT/S)	(DEG)
144.49	203396.	21113067.	63.42	7630-72	75.70
160.00	252205.	21161604.	66.62	7325.25	75.90
183.00	304564.	21214073.	70.90	7094.01	76.10
200.00	345241.	21254660.	75.44	6912.59	76.30
220.00	374287.	21283618.	80.17	678J <sub>=</sub> 69	76.50
243.00	391734.	21300979.	85.05	6705.49	76.70
260.00	397600.	21306760.	90.30	6673.47	76.90
280.00	391892.	21300968.	94.96	6703.15	77.10
300.00	374604.	21283596.	99.84	6783.02	77.39
320.00	345717.	21254627.	104-57	6911-55	77.51
340.00	305202.	21214031.	109.10	7592.24	77.71
360.00	253025.	21161773.	113.39	7316.15	77.92
380.00	189314.	21397583.	117.38	7529.83	78.15
400.00	117056.	21025650.	120.60	6996.22	78768
420.00	66614.	20975164.	116,56	2961.45	82 - 85
460.00	37841.	20946379.	110.53	1548.96	88.74
480.00	27815.	20936351-	108.46	1461.63	89.33
500.00	19207.	20927743.	106.49	1413.67	89.68
520.00	11671.	20923206.	104.84	1384.76	89.88
540.00	4956.	20913491.	103.49	1365.28	90.32
556.17	0.	20908535.	102.59	1355.00	90.08

TABLE SC S-IB STAGE REENTRY DATA

	TIME,	*ASS	THRUST	VTHE .	AZI	RANGE	VVVE
	(SEC)	(L8)	(LB)	(DEG)	(DEG)	(FT)	(FT/S)
	144.49	101713.03	44478.	58.28	72.23	2,3242.	6493.74
	160.00	10Ç511.76	ე.	61.79	72.41	286730.	6150.22
	180.00	100511.76	0.	66.73	72.64	393443.	5874.12
	200.00	100511.76	0.	72.09	72.87	499545.	5654.08
	229.00	100511.76	0.	77.83	73.11	605205.	5492.24
	240.00	100511.76	ე.	83.84	73.35	710568.	5392.91
	260.00	10051176	0.	90.30	73.60	815777.	5359-13
	280.00	190511.76	0.	96.16	73.86	920976.	5391.96
	300.00	100511.76	0.	102,18	74.12	1026306.	5493.36
	323.00	100511.76	0.	107.92	74.39	1131910.	5651.30
	340.00	100511.76	0.	113.29	74.66	1237932.	587.13
	360.00	100511.76	0.	118.24	74.94	1344507.	6138.48
	380.00	100511.76	C.	122.75	75.22	1451451.	6404.46
	400.00	100511.76	0.	127.03	75.50	1554519.	5914.17
	420.00	1J0511.76	0.	134.67	75.76	L615684.	1883.67
	460.00	100511.76	ō.	166.89	75.40	1634705.	557.69
	480.00	100511.76	9.	172.46	74.77	1634495.	466.88.
	500.00	100511.76	0.	175.74	75.11	1635373.	402.37
, <u>س</u>	520.00	100511.76	0.	177.94	77.42	1635785.	354.85
129	540.00	100511.76	õ.	179.65	103.75	1635922.	318.52
	556.17	100511.76	٥.	179.31	236.99	1635897.	295.38

TABLE 5C S-IB STAGE REENTRY DATA

TIME	X	Y	Z	DX	, * DY	DΖ
(SEC)	(FT)	(FT	(FT)	(FT/S)	(FT/S)	(FT/S)
144.49	21109331.	116829	217018.	3∠89 <b>.</b> 19	389.20	407/ /2
160.00	21155889.	122652	485032.	2753.40	387.29	6874.42
189.00	21204598.	130569	620390.	2121.72		6778.24
200.00	21240784.	138225	755357.	1497.37	384.35	6758.37
220.00	21264507.	145815	889873.	875.27	381.19	6737.69
240.00	21275805.	153336.	1023865.	254.71	377-83	6713.34
260.00	21274702.	160784.	11.57255.	-364.99	374.25	6685.18
280.00	21261207.	16815.4.	1289966.		370.47	6653.17
300.00	21235320.	175441.	1421923.	-984.47	366.47	6617.29
320.00	21197026.	182642.	1553539.	~1604.38	362.26	6577.49
340.00	21146296.	189752.		-2225.37	357.82	6533.70
360.00	21083099.	196767.	1683240.	-2848.01	353.15	6485.68
380.00	21007594.	203681.	1812424.	-3471-35	348.25	643J.75
400.00	23923198.	210494.	1940208.	-4065.52	343.15	6328,68
420.00	23425140.		2062389.	-4137.79	338.34	5631.28
460.00	20828278.	217239.	2144073.	-1589.21	336.58	2476.16
480.00		230568.	2209180.	-692.61	329.55	1345.72
500.00	20815317.	237143.	2235437.	-607.61	328-36	1288.16
520,00	20803829.	243709.	2260914.	-544.71	328.32	1262.52
,	20793424.	250275.	2286025.	-498,12	328.28	1249.66
540.00	20783834.	256839.	2310935.	-462.64	328.04	1241.91
556.17	20776541.	262142.	2330986.	-440.36	327.63	1238.86

TABLE 5C S-IB STAGE REENTRY DATA

TIMĖ	ΧE	ΥE	ZE	DXE	DYE	DZE
(SEC)	(FT)	(FT)	(FT)	(FT/S)	(FT/S)	(FT/S)
144.49	202402•	-110.	205039,	3365.34	3.57	.5553.66
160.00	250221.	-24.	289960.	2837.38	7.57	5456.60
180.20	300818.	183.	398885.	2223.53	13.19	5437.00
200.00	339188.	507.	507445.	1613.96	19.28	5418.80
220.00	365390.	957.	615627.	1006.53	25.83	5399.16
240.00	379459.	1543.	723401.	403.51	32.82	5377.92
260.00	3£1416.	2272.	830733.	-204.77	40.24	5355.36
280,00	371269.	3155.	937592,	-809.98	48.13	5333.56
300.00	349013.	4199.	1043944.	-1415.78	56.37	5304.38
320.00	314630.	5413.	1149756.	-2022.83	65.05	5276.46
340.00	268089.	6804.	1254990.	-2631.74	74.13	5246.63
360.00	·209353.	8380.	1359593.	-3241.61	83.56	5212.19
380.00	138569.	10145.	1463223.	-3823.58	92.58	5132.01
430.00	59047.	12021.	1561639.	-3899.07	89.61	4445.97
420.00	4100.	13235.	1619111.	-1422.66	28.24	1234.29
460.00	~25911.	13575.	1633913.	-551.29	0.41	84.22
480.00	~36047.	13562.	1634927.	-466.19	-1.14	25.36
500.00	~44697.	13540.	1635138.	-402.37	-0.96	-1.22
520.00	~52243.	13526.	1634966.	-354.54	-0.51	-14.73
540.00	~58949.	13520.	1634584.	-317.69	-0.11	-22.94
556.17	_ 4 3 0 0 0	13520.	1634176.	-294.22	0.09	-26.22

TABLE 5C S-IB STAGE REENTRY DATA

	TIME.	CHIP	CHIY	CHIR	LØNG	CECL	LATT
	(SEC)	(DEG)	(CEG)	(DEG)	(DEG)	(DEG)	(DEG)
	144.49	-59.32	c.oc	0.00	-79.96	28.53	28.69
	160.00	-59.32	C-80	0.00	-79.71	28.60	28,76
	180.00	-59.32	€.30	0.00	-79.39	28.69	128.85
	200.00	-59.32	, 0.00	0.00	-79.08	28.78	28.94
	220.00	-59.32	C.CO	0.00	-78.76	28.86	29.52
	240.00	-59.32	C.00	0.00	-78.44	28.94	29.11
	260.00	-59-32	C:00	0.00	-78.13	29.03	29.19
	280.00	-59.32	¢.08	0,00	-77.81	29.11	29.27
	300.00	·59.32	C.0C	0.00	-77.49	29.19	29.35
	320.00	<i>-</i> 59.32	C.00	0.00	-77.18	29.27	29.43
	340.00	-59-32	C+00	0.00	-76.85	29.34	29.51
	360.00	+59.32	C+30	0.00	-76 <b>.5</b> 3	29.42	29.58
	380.0ა	•59.32	C.OC	0.00	-76.21	29.50	29.66
	400.00	.59.32	C.00	0.00	-75.89	29.57	29.73
	420.00	-59.32	C-90	0.00	-75.70	29.61	29.77
	463.00	-59.32	. 6.00	0.00	-75.65	29.62	29.79
H	483.00	-59.32	C.00	_,0∓00	-75.65	29.62	29.79
132	500.00	-59.32	C.00	0.00	-75.64	29.62	29.79
-	520.00	-59.32	¢ - ôo	0.00	-75.64	29.62	29.79
	540.00	.59.32	C - OC	0.00	-75.64	29.62	29.79
	556.17	.59.32	C.00	0.00	-75.44	29.62	29.79

TABLE 5C S-IB STAGE REENTRY DATA

BAIT	ACH	DRAG	ALP	QQQ	'VR	TACEL
(SEC)	(U)	(LB)	(DEG)	(L8/FT2)	(FT/S)	(FT/S2)
144.49	6.36	15469.	0.00	11.	6490.39	9.00
160.00	6.65	1731.	0.00	1.	6146.75	0.55
180.00	6.49	94.	ა. აა	0.	5870.50	0.03
200.00	5.58	16.	0.00	Ů.	5650.34	J.00
220.00	4.81	2.	<b>0.00</b>	0.	5488.39	J. U)
240,00	4.35	1.	0.00	o.	5389.33	J. 00
260.00	4.15	1.	0.00'	0.	5355.19	J.60
280.00	4.35	1.	0.00	o.	5388.05	3.03
300.00	4.80	2.	0.00	0.	5486.52	0.00
320.00	5.57	10.	0.00	5.	5647.55	J.00
340.00	6.48	90.	0.00	0.	5866.51	J.03
360.00	6.65	1659.	0.00	1.	6135.01	ა.53
380.00	.6.11	15632.	0.00	17.	6397.15	8.20
400.00	5.75	30275.	0.00	.62 <b>.</b>	5911.03	124.93
420.00	1.98	535575.	0.00	396.	1888.92	171.44
460.00	0.56	115309.	0.00	.03.	548.04	36.91
480.00	0.45	111464.	-0.00	.04.	463.72	35.68
500.00	0.38	108548.	-0.00	.03.	401.48	34.75
520.00	0.32	106644.	-0.00	.02.	354.70	34.14
540.00	0.28	105340.	-0.00	.01.	318.55	33.72
556.17	0.26	104318.	-0.00	.00.	295.36	35.39

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R-AERO-AD, Mr. Dunn	R-1
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R-AERO-DD, Mr. Ryan	R-1
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